

## CTF 4.0 User's Manual

 $R.\ Salko^1,\ M.\ Avramova^2,\ A.\ Wysocki^1,\ J.\ Hu^1,$ 

A. Toptan<sup>2</sup>, N. Porter<sup>2</sup>, T. Blyth<sup>3</sup>, C. Dances<sup>3</sup>,

A. Gomez<sup>3</sup>, C. Jernigan<sup>4</sup>, and J. Kelly<sup>5</sup>

<sup>1</sup>Oak Ridge National Laboratory

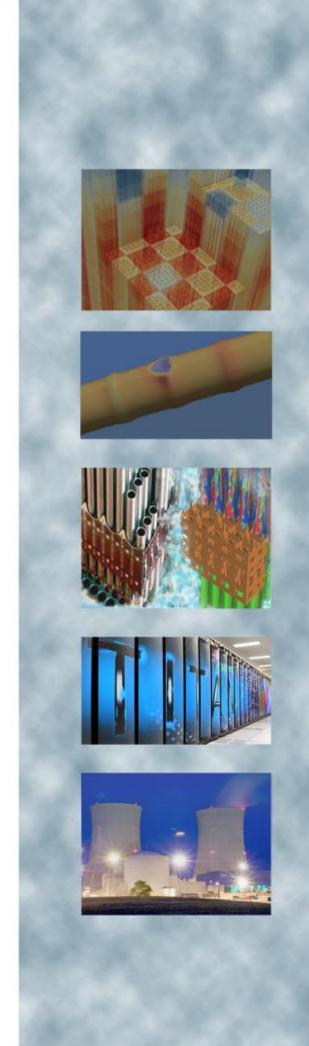
<sup>2</sup>North Carolina State University

<sup>3</sup>The Pennsylvania State University

<sup>4</sup>Holtec International

<sup>5</sup>Nuclear Regulatory Commission

3/24/2019





### THE NORTH CAROLINA STATE UNIVERSITY

DEPARTMENT OF NUCLEAR ENGINEERING

REACTOR DYNAMICS AND FUEL MODELING GROUP



# CTF 4.0 User's Manual

March 24, 2019

	•
	A CD CATA A
	ACRONYMS

 ${f PWR}$  Pressurized-Water Reactor

 $\mathbf{BWR}$ Boiling-Water Reactor

CONTENTS

1	Inp	t Manual	1
	1.1	General Remarks	1
		1.1.1 Revision History	1
		1.1.2 How to Read this Document	1
		1.1.3 Standard CTF Input File	2
2	Ma	Problem Control Data	4
	2.1	Units of Physical Quantities in Input and Output	4
	2.2	Restart Data	4
	2.3	Iteration Control	5
	2.4	Title Card	7
3	Car	Group 1	8
	3.1	Selection of Physical Models	8
	3.2	Global Boundary Conditions	12
	3.3	Initial Conditions	13
	3.4	Custom Friction Model Input	14
	3.5	Under-relaxation coefficients	14
	3.6	Notes on Card Group 1	15
4	Car	Group 2	16

5	Card Group 3	20
6	Card Group 4	<b>25</b>
7	Card Group 5	29
8	Card Group 6	31
9	Card Group 7	33
10	Card Group 8	42
	10.1 Rod Geometry Data	46
	10.2 Unheated Conductor Data	48
	10.3 Rod Temperature Initialization Tables	50
	10.4 Radiation Initialization Tables	51
	10.5 Radiation Channel Orientation Array	53
	10.6 Cavity to Cavity Radiation Weighting Factor	57
	10.7 Radiation Location Type Information	57
11	Card Group 9	62
11	Card Group 9  11.1 Nuclear Fuel Geometry Types	<b>62</b>
11	•	
	11.1 Nuclear Fuel Geometry Types	64
12	11.1 Nuclear Fuel Geometry Types	64 68
12	11.1 Nuclear Fuel Geometry Types	64 68 <b>71</b>
12	11.1 Nuclear Fuel Geometry Types	64 68 <b>71</b> <b>73</b>
12	11.1 Nuclear Fuel Geometry Types	64 68 <b>71</b> <b>73</b> 75
12 13	11.1 Nuclear Fuel Geometry Types	<ul><li>64</li><li>68</li><li>71</li><li>73</li><li>75</li><li>75</li></ul>
12 13	11.1 Nuclear Fuel Geometry Types	64 68 <b>71</b> <b>73</b> 75 75
12 13 14	11.1 Nuclear Fuel Geometry Types  11.2 Non-nuclear Geometry Types  Card Group 10  Card Group 11  13.1 Total Power Forcing Function  13.2 Gap Conductance Forcing Function  13.3 Radial Power Profile Forcing Function  Card Group 12	64 68 <b>71</b> <b>73</b> 75 75 75

www.mne.psu.edu/rdfmg

	16.2 New Card Group 14	90
17	Card Group 15	95
18	3 Card Group 16	97
19	Card Group 17	98
20	Card Group 18	102
21	Card Group 19	104
22	2 Users' Guide	109
	22.1 General	109
	22.2 Specification of the Geometry Data	112
	22.2.1 Instructions to CARD GROUP 2	112
	22.2.2 Instructions to CARD GROUP 3	117
	22.2.3 Instructions to CARD GROUP 4	122
	22.2.4 Instructions to CARD GROUP 5 and 6	129
	22.2.5 Instructions to CARD GROUP 7	132
	22.3 Specification of the Conductors' Data	136
	22.3.1 Instructions to CARD GROUP 8	136
	22.3.2 Instructions to CARD GROUP 9	141
	22.3.3 Instructions to CARD GROUP 10	148
	22.4 Specification of the Initial and Boundary Conditions	148
	22.4.1 Instructions to CARD GROUP 1	149
	22.4.2 Instructions to CARD GROUP 11	150
	22.4.3 Instructions to CARD GROUP 13	154
	22.4.4 Boron Tracking and Precipitation Modeling	159
	22.4.4.1 Application of the Boron Tracking/Precipitation Model	159
	$\begin{array}{cccccccccccccccccccccccccccccccccccc$	160
	22.5 Turbulent Mixing and Void Drift Modeling	161

Re	efere	nces	186
	A.3	CTF Input Deck for the GE 3x3 Test Point 2B2	178
	A.2	CTF Model of the GE 3x3 Test Section	177
	A.1	GE 3x3 Experimental Parameters	176
$\mathbf{A}$	Calo	culation Notes and the 3x3 GE Experiments Input Decks	176
		22.7.3 Instructions to Preparation of Input Files for Restart Calculations	170
		22.7.2 Instructions to Input of Time Domain Data	169
		22.7.1 Instructions to Input of Main Problem Control Data	168
	22.7	Main Problem Control and Time Domain Data	168
		22.6.1 Instructions to CARD GROUP 14 (Legacy)	167
	22.6	Results Reporting	167
		22.5.2 Instructions to CARD GROUP 12	166
		22.5.1 Instructions to CARD GROUP 12	165

## LIST OF FIGURES

4.1	of where the origin can be placed	18
4.2	Definition of the $(0,0)$ location and $\mathbf X$ and $\mathbf Y$ for core geometry	18
10.1	Radiation Geometry Type 1	53
10.2	Radiation Geometry Type 2	53
10.3	Radiation Geometry Type 3	54
10.4	Radiation Geometry Type 4	54
10.5	Radiation Geometry Type 5	54
10.6	Radiation Geometry Type 6	54
10.7	Radiation Geometry Type 10	55
10.8	Radiation Geometry Type 11	55
10.9	Radiation Geometry Type 12	55
10.10	Radiation Geometry Type 13	55
10.11	Radiation Geometry Type 14	56
10.12	Radiation Geometry Type 15	56
10.13	Radiation Geometry Type 16	56
10.14	Radiation Geometry Type 17	56
19.1	Core map of 3x3 assemblies	100

### LIST OF FIGURES

22.1 Basic mesh cell	110
22.2 Mesh cell for axial momentum	110
22.3 Mesh cell for transverse momentum	110
22.4 Basic Subchannel	111
22.5 Subchannel node numbering convention	113
22.6 Subchannel connections at section boundaries allowed by the subchannel splitting logic	114
22.7 Typical configuration for convection of transverse momentum between sections	116
22.8 Axial momentum mesh cell at section boundary	118
22.9 Global coordinate systems	120
22.10Axial momentum mesh cell at section boundary	121
22.11Convection of transverse momentum by an orthogonal transverse velocity	122
22.12Diagram of the variable axial node length	124
22.13Allowable vertical connections between subchannels at section boundaries	125
22.14Common subchannel splitting errors	127
22.15Common subchannel splitting errors	128
22.16Examples of axial variation in continuity and momentum area and wetted perimeter of a subchannel	130
22.17Example of subchannels with local form losses due to spacer grids	134
22.18Example of rod surface numbering in CARD 7.5	134
22.193x3 BWR Bundle	139
22.20Nuclear fuel rod geometry	142
22.21Heater rod geometry	147
22.22Nuclear fuel rod power profile	151
22.23Heat input over one fluid node	151
22.24Heater rod crossing section boundaries	152
22.25Control volume for pressure sink boundary conditions	157
A 1. Cross section of the CTE model of the CE 3v3 rod bundle	177

Chapter $1$	
1	
	INPUT MANUAL

### 1.1 General Remarks

#### 1.1.1 Revision History

The deck version is specified using the PPV flag on Card 1.1. In general, you should always use the latest version of the CTF input deck in order to make sure your model is treated correctly in CTF. This feature was added to retain backwards compatibility for older input decks. When an input deck is created with the CTF pre-processor, the latest version will always be used. Table 1 provides the revision history of the CTF input deck. The first column gives the version number of the deck and the second columns gives a description of what changed since the previous version of the input deck.

#### 1.1.2 How to Read this Document

This document describes how to make a CTF input deck. A CTF input deck is organized into Card Groups and Cards. A Card Group is a collection of Cards. A Card is defined as a line of input. Each Card may contain multiple data. A Card is terminated by making a new line.

This document has been organized so that each Card Group is discussed in its own dedicated chapter. Each card is discussed in its own dedicated section. Each data in the card is discussed in its own block. The block gives information about the data, including the number of the input, the title, a description of the meaning of the data, units, data type, and so on. An example block is shown below to discuss the meaning of each entry in the block.

{Card Group Number}.{Card Number}	{Parameter Name}	[{Physical units of the parameter in SI}]	[{Physical units of the parameter in US}]	
{Description of the parameter. Possible values.}				
{Data type}	{Required or Optional}			

Table 1: Revision history of the CTF input deck

#### Version Remarks

- 0 Original CTF input deck version
- 1 Added ability to have more ghost entities (rods, gaps, channels) for parallel models. All parallel decks should be at least Version 1 to work correctly
- 2 Added new term, SYMROD, on Card 8.2, which allows the user to specify if the rod is a partial rod (if a symmetry line runs through the rod). All symmetry models should be at least version 2.
- 3 Added new friction correlations that allow for modeling surface roughness effects as well as for user to input custom friction factor correlation. These new correlations require surface roughness to be specified or correlation coefficients to be specified, which required a change to the input deck format.
- 4 Removed the gap momentum cell coordinates from Card 3.3.5. Code will only read in gap number and gap norm.
- 5 Added  $l_2$ -norm stopping criteria to Card 19 of the input deck.
- 6 Added ability to specify HTC and TKE maps for fuel rods in Card 8 as well as specify how many sub-regions each rod surface should be divided into when doing coupled crud/TH simulations
- 7 Added ability to connect to more than just 6 channels above/below in Card 4.4

### 1.1.3 Standard CTF Input File

- The CTF executable can be run with the input file name as an argument on the command line. If no file name is provided, the code will look for a file called "deck.inp" by default. For all options, run the CTF executable with the "-help" argument.
- Input files are read in a **free-formatted** way only
- Because of the free-formatted input structure **each** of the parameters described below has to be specified in the input file (as dummy parameters, in case they are not used in the code)
- Lines must **not** be longer than **200** characters. Otherwise the remaining characters are truncated, which could cause reading errors

CTF is intended to be built and run on a Linux-based operating system. For a serial run, only one input file is required. This can be called the default name, "deck.inp", or a custom name "\name\name\name\name\".inp". If the default naming convention is used, the code may be run either as:

```
${CTF}
or
${CTF} deck.inp
or
${CTF} deck
If the custom naming convention is used, the code may be run either as:
${CTF} (name).inp
```

#### \${CTF} \(\lame\)

For a parallel run, a separate input file is required for each solution domain in the model. Each input deck should include the channels, rods, and gaps solved by that domain plus any additional ghost channels, rods, and gaps that are needed for sharing information with other solution domains. A single master input file is also required to summarize the total number of rods, channels, and gaps in the model, the global CTF mesh, as well as the total number of solution domains in the model. Creating parallel input files by hand can be confusing and error prone and, therefore, it is recommended that the CTF preprocessor, "xml2ctf", be used to automate this process. The default naming convention for the parallel input file is "pdeck.  $\langle nproc \rangle$ .  $\langle rank \rangle$ . inp" and "pmaster.inp" for the master input file. The term, " $\langle nproc \rangle$ " is the number of solution domains in the model and " $\langle rank \rangle$ " is the index of the solution domain described by the input file (using 1-based indexing). The custom naming convention for the parallel input file is " $\langle name \rangle$ .  $\langle nproc \rangle$ .  $\langle rank \rangle$ . inp" and " $\langle name \rangle$ . master.inp" for the master input file. The parallel simulation is run for default file names as:

```
mpirun -np (nproc) ${CTF}
or
mpirun -np (nproc) ${CTF} pdeck
or
mpirun -np (nproc) ${CTF} pdeck.inp
The parallel simulation is run for custom file names as:
mpirun -np (nproc) ${CTF} (name)
or
mpirun -np (nproc) ${CTF} (name.inp)
Additional options exist for running CTF. These can be found by executing CTF with the -help command:
${CTF} -help
```



This card group is read by subroutines INPUT and COBRAI.

# 2.1 Units of Physical Quantities in Input and Output

INPUT.1	ICOBRA	[—]	[—]		
Units used in input and or	Units used in input and output files:				
0 - US input / SI output	0 - US input / SI output				
1 - SI input / SI output					
2 - US input / US output					
3 - SI input / US output					
T .	D . 1				
Integer	Required				

## 2.2 Restart Data

Note that the CTF restart feature is currently not functional. All following inputs must be set so the restart feature is not being utilized.

INPUT.2 INITIAL [—] [—]

Vessel initialization option:

1 - Initial start
 Initial values are specified in Card 1.2

2 - Full restart (not currently working)
 Variable arrays are filled with data obtained from the restart file 'deck.crs'. Operating conditions, boundary conditions, and time domain data can be changed for the restart fun.

3 - Simple restart (not currently working)
 Variable arrays are filled with data obtained from the restart file 'deck.crs'. Only time domain data can be changed for the restart run.

Integer Required

If INITIAL is set equal to 2 (full restart), then the restart file:

• must contain Card Groups:

INPUT.1, INPUT.2, INPUT.3, COBRA.1, 15

• may contain Card Groups:

1, 11, 12, 13, 14

• must not contain Card Groups:

2, 3, 4, 5, 6, 7, 8, 9, 10

INPUT.2	DUMPF	[—]	[—]
Flag for restart file genera	tion:		
0 - No restart file is gene	rated		
1 - Restart file 'deck.cd the calculation	.m' is generated in time in	teverals of DMPINT (Card 1	15.1) and also at the end of
Integer	Required		

### 2.3 Iteration Control

The following three controls were implemented to allow the user to modify solution tolerances and iteration caps on the CTF numerical solution algorithm. However, not all of these controls are currently functional in the code.

EPSO will only have an effect if PETSc is used to solve the pressure matrix (ISOL=5 or 7). In this case, EPSO is defined as the relative tolerance at which the iterative pressure matrix solution can be stopped for a given outer iteration of CTF. For steady-state simulations, it is permissible to set this value to a relatively high value. This may speed up the pressure matrix solve substantially at the cost of slightly larger error in the intermediate calculated pressure distribution; however, the intermediate pressure solution is not of great importance in a steady-state solution. At the end of the simulation, the solution will arrive at the same pressure distribution as if a lower relative tolerance were used. For transient simulations, the

intermediate CTF outer iteration solutions are important, and a lower relative tolerance should be used. The PETSc default tolerance of  $1.0 \cdot 10^{-5}$  is recommended for transients. A default value for steady-state solutions cannot be recommended, as an in-depth analysis of this parameter effect on solution time has not been performed. Furthermore, the optimal value will change with problem size and number of solution domains (larger problems will spend more time in the pressure matrix solve due to increased communication costs), as well as system hardware (i.e., network speed).

OITMAX was historically put in the code to set the maximum allowable number of outer iterations in CTF; however, at some point in CTF's development history, the outer iteration loop has been disabled. CTF will currently take only one outer iteration per timestep. The assumption is that the timestep size will be sufficiently small to ensure that the solution obtained at the end of the single iteration is accurate. OITMAX currently has no effect whatsoever on the CTF simulation, and so it may be set to any value.

IITMAX will have different effects depending on the pressure matrix solver employed. If using the multisimulation-group Gauss-Siedel solver (ISOL=0 and NSIM>1), IITMAX will set the maximum allowable number of iterations of the Gauss-Siedel solution. If one of the PETSc solvers is used (ISOL=5 or 7), this term serves as the maximum allowable iterations in a single pressure matrix solve. Similar to the EPSO term, it is permissible to set this to a value that reduces the total number of pressure matrix iterations for steady-state solves, while transient solves should converge the pressure matrix more tightly at each timestep. The PETSc default of 1E4 is suggested for transients. Again, the optimal value for IITMAX will depend on problem size, parallel domain decomposition, model conditions, and machine performance, so a one-size-fits-all optimal value cannot be specified; however, limited testing has revealed that a value of 40 seems to work well for single phase models ranging in size from 1 to 25 17x17 assemblies (1 to 25 solution domains).

INPUT.3	EPS0	[—]	[—]		
Relative tolerance for PETSc pressure matrix solve. The default of $1 \cdot 10^{-5}$ should be used for transients. A larger value may be used for steady-state solutions.					
Float	Required				
INPUT.3	OITMAX	[—]	[—]		
Maximum number of oute	r iterations - no effect on so	olution, enter any value			
Integer	Required				
	1				
INPUT.3	IITMAX	[—]	[—]		
If using Gauss-Siedel solver, this is the maximum number of Gauss-Siedel iterations. If using the PETSc solver, this is the maximum number of PETSc pressure matrix iterations. The suggested value is 40 for the Gauss-Siedel solver. When using PETSc, the default of 1E4 should be used for transients. A smaller value may be used for steady-state solutions.					
Integer	Required				
INPUT.3	COURANT	[—]	[—]		
The Courant number to be used in setting the timestep size. If left blank, it will default to the traditional CTF value of 0.8. A suggested value for steady-state simulations is 3.0, but check the 'deck.run' file for numerical stability issues associated with the timestep becoming too large. The proper Courant value will be highly dependent on the particular case being modeled. Timestep size may also be limited using controls in Card Group 15.					
Float	Optional				

## 2.4 Title Card

COBRA.1	TEXT	[—]	[—]
Alphanumeric information to identify the simulation (maximum 30 characters)			
Character	Optional		

CHAPTER 3	
I	
	CARD GROUP 1

This card group is read by subroutine READ\_CARD\_1.

# 3.1 Selection of Physical Models

1.1	NGAS	[—]	[—]
Number of non-condensable gases. Can be between 1 and 8. Description of gasses that can be modeled are			
givin in Card 1.4.			
Integer	Required		

1.1 | IRFC | [--] | [--]

Rod friction factor correlation. See Section ?? of the theory manual for full description.

- 1.  $\lambda = \max(1.691Re^{-0.43}, 0.117Re^{-0.14})$
- 2.  $\lambda = 0.204 Re^{-0.2}$

$$3. \ \lambda = \left(-2.0 \log_{10} \left(\frac{\epsilon}{3.7 D} + \frac{2.51}{Re} \left[1.14 - 2 \log_{10} \left(\frac{\epsilon}{D} + \frac{21.25}{Re^{0.9}}\right)\right]\right)\right)^{-2}$$

4. 
$$f_f = 2 \left[ \left( \frac{8}{Re} \right)^{12} + \frac{1}{(a+b)^{3/2}} \right]^{1/12}$$

$$a = \left(2.475 \ln \left[ \frac{1}{\left(\frac{7}{Re}\right)^{0.9} + 0.27 \left(\frac{\epsilon}{D}\right)} \right] \right)^{16}$$

$$b = \left(\frac{3.753 \times 10^4}{10^4}\right)^{16}$$

5. User defined correlation:  $A + BRe^{C}$ 

Note that if Correlations 3, 4, or 5 are selected, you must use input deck version 3 (Set PPV>=3). If correlations 3 or 4 are selected, you must enter a surface roughness for each solid object in the model (see Card 9.2/9.6). If correlation 5 is selected, you must enter Card 1.5 to specify values for A, B, and C.

Integer Required

1.1 EDMOD [—] [—]

Entrainment and deposition model:

- 0 No entrainment or deposition
- 1 Original model, which models entrainment and deposition

Integer Required

1.1 | IMIX | [—] | [—]

Mixing and void drift model:

- 0 Neither mixing nor void drift
- 1 User-specified constant two-phase turbulent mixing coefficient
- 2 Single-phase mixing coefficient according to Rogers and Rosehart (1972)
- 3 User-specified constant single-phase turbulent mixing coefficient; two-phase multiplier according to Beus (1970)

Integer Required

1.1	ISOL	[—]	[—]		
Solver for the pressure equ	nation:				
0 - Direct Gaussian elimination $\underline{\text{or}}$ Iterative Gauss-Seidel solver (For pressure equation, selection by means of parameter NSIM in Card Group 4.1)					
3 - Iterative Krylov solve	r: BCGS (Bi-Conjugate Gr	adient Method Stabilized)			
a parallel run)		-	automatically choose if it is sues with option 1. Options		
2 and 4 were not tes		d as there were memory is.	sucs with option 1. Options		
Integer	Required				
	I	Fa / 1			
1.1	GINIT	[kg/s]	[lbm/s]		
	flux will be the Section 1 in		h. The cross-sectional area is done for GTOT. Enter 0.0		
Float	Required				
1.1	NOTRANS	[—]	[—]		
Flag to specify whether C criteria are used instead:	TF should use TEND from C	Card Group 15 to end the s	simulation or if convergence		
0 - Use Card Group 15 to	ransient information				
1 - Use internal code com	vergence criteria				
	ll be named according to the	_	interval of the transient. If $I''$ convention, where "N" is		
Integer	Required				
1 1	MEGIT	r 1	r 1		
1.1	MESH				
			meshing information) or if it de the meshing information		
0 - No meshing informati	on				
1 - Meshing information	provided on Cards 2.2 and	new Card 3.3.5			
Integer	Required				

1.1	MAPS		
		channel map informatio	n necessary for writing HDF5
0 - Do not look for Grou	p 17 or print HDF5 edits		
1 - Group 17 is present,	read it and produce HDF5	edit file	
Integer	Required		
1.1	TDDODG	[ ]	
Fluid property tables to u	IPROPS		
		aug gaumaga)	
_	property tables (mix of various)	ous sources)	
1 - IAPWS IF97 water p	roperty tables		
_	roperty tables, using a second to temperature and pre-		ce approximation for property
11 - FLiBe liquid salt pro	operty correlations		
Integer	Required		
1.1	MFLX	r 1	
mass flow rate. If using th	is option, GTOT and GINIT r IT based on the channel flo low BC as mass flux	nust be set to 0.0. CTF	using a mass flux rather than will use the mass flux supplied
Integer	Required		
		r 1	
1.1	IBTM		
and precipitation model, i $0$ - No boron tracking/pr	s read in. Additional initial		r applying the boron tracking and Group 1.
1 1 1150 order decurate t	upwind boron tracking mod	el and boron precipitation	on model (Kim's correlation)
2 - Second order accurate correlation)			on model ( $Kim$ 's correlation) on precipitation model ( $Kim$ 's
correlation) Boron information provide boundary conditions, and	e Modified Godunov boron ed on Card Group 13 (Car Cards 13.2 and 13.3 specif	tracking model and bord d 13.1 specifies the total y the boron forcing func	on model (Kim's correlation) on precipitation model (Kim's number of vertical mesh cell etions; and a new Card: Card ameters are provided on Card
correlation)  Boron information provide boundary conditions, and 13.11, to introduce the board.	e Modified Godunov boron ed on Card Group 13 (Car Cards 13.2 and 13.3 specif	tracking model and bord d 13.1 specifies the total y the boron forcing func	on precipitation model (Kim's number of vertical mesh cell etions; and a new Card: Card

1.1	PPV	[—]	[—]	
The input deck version number. See Table 1 for a meaning of the version numbers. Always try to use the				
most recent version of the	input deck.			
Integer	Required			
1.1	BWRMODEL	[—]	[—]	
It is not recommended this option be used when generating a CTF input deck by hand. It is intended that the preprocessor should be used to activate this feature. This feature tells CTF that this is a BWR mode where the assemblies are completely separate from one another (not connected at top or bottom). This feature will enable an outer iteration loop in CTF that will adjust the inlet mass flow rates to make the pressure drop in all fuel assemblies equal. Model must be one axial section and have inlet mass flow rate specified in the first level of every channel. Must also be a parallel model. Note that, if this model is enabled the pressure matrix solver will be set to ISOL=3 by default. Options:  0—Disable outer iteration loop				
1—Enable outer iteration loop (use for multi-assembly BWR models created by preprocessor)				
Integer	Required			

# 3.2 Global Boundary Conditions

1.2	GTOT	[kg/s]	$[\mathrm{lbm/s}]$		
Total inlet mass flow rate. If GTOT $\neq 0$ and the inlet boundary condition type (see Card 13.4) is inlet mass					
flow rate and inlet enthalpy (BC type 2), the code will calculate subchannel mass flow rates according to the					
	subchannels flow areas as specified in CARD 2.1 and will ignore the subchannel mass flow rates in CARD				
		_	and inlet enthalpy (BC type		
2), the user must specify s	subchannel mass flow rates	in CARD 13.4. Enter 0.0 is	$f  ext{ setting MFLX} = 1.$		
Float	Required				
1.2	AFLUX	$[\mathrm{kW/m}]$	$[\mathrm{kW/ft}]$		
Average linear heat rate p	per rod. To calculate AFLU	X the total bundle power	is divided by the total rod		
length multiplied by the to	otal number of rods.				
Float	Required				
1.2	DHFRAC	[—]	[—]		
Fraction of local heat rate	generated by the heater r	ods which is released direc	tly into the coolant. (As a		
coarse approach, the direc	t heat is added to the <i>liquid</i>	d only—not to the vapor).			
Float	Required				
<u> </u>					
1.2	MFLUX	$[{ m kg/s\text{-}m^2}]$	$[\mathrm{lbm/s ext{-}ft^2}]$		
The inlet and initialization	n mass flux. Only read if M	FLX = 1			
Float	Conditional				

## 3.3 Initial Conditions

1.3	PREF	[bar]	[psi]	
Initial pressure in the fluid	d domain			
Float	Required			
			\	
1.3	HIN	$[\mathrm{kJ/kg}]$	[BTU/lbm]	
Initial enthalpy in the flui				
	TIN must be supplied**	4000 1 7 /1		
	red as a positive value, such	m as~1300~kJ/kg.		
Float	Required - Option 1 of 2			
	I	[0.01	[own]	
1.3	TIN	[°C]	[°F]	
Initial temperature in the				
	TIN must be supplied**	For example 310 °C week	ld be entered as -310.0 for	
TIN.	itered by setting it negative	. For example, 510 °C wou	id be entered as -310.0 for	
Float	Required - Option 2 of 2			
1.3	HGIN	$[\mathrm{kJ/kg}]$	[BTU/lbm]	
Enthalpy of non-condensa	ble gas mixture			
Float	Required			
	<del>-</del>			
1.3	VFRAC(1)	[—]	[-]	
Initial liquid volume fract	ion in the liquid-vapor-gas	mixture		
Float	Required			
	<del>-</del>			
1.3	VFRAC(2)	[—]	[—]	
Initial vapor volume fract	ion in the vapor-gas mixtur	e		
Float	Required			
1.3	BRIN	[ppm]	[ppm]	
Initial boron concentration	n (uniform distribution)			
Float	Conditional			
1.3	RDIF	[—]	[—]	
Boron physical diffusion c	oefficient:			
0.0 - Suggested value (fir	st order unwind scheme TR	<b>РТМ</b> — 1)		
0.0 - Suggested value (first order upwind scheme, IBTM $= 1$ )				
1.0 - Suggested value (modified Godunov scheme, IBTM $= 2$ )				
Til .	G 193 1			
Float	Conditional			

1.4	GTYPE(I)	[—]	[—]	
Name of non-condensible	gas; acceptable names:			
air, argo, heli, hydr, kryp,	nitr, oxyg, xeno			
Character	Required			
1.4	VFRAC(I+2)	[—]	[—]	
Initial volume fraction of non-condensable gas $I$ in the vapor- gas mixture				
Float	Required			

## 3.4 Custom Friction Model Input

1.5	A	[-]	[-]		
	Coefficient in the user-defined friction correlation:				
$A + BRe^C$					
Only enter if IRFC=5 and	PPV >= 3.				
Float	Required if IRFC=5				
1.5	B	[-]	[-]		
Coefficient in the user-defi	ned friction correlation:				
$A + BRe^C$					
Only enter if IRFC=5 and	PPV >= 3.				
Float	Required if IRFC=5				
1.5	C	[-]	[-]		
Coefficient in the user-defined friction correlation:					
$A + BRe^{C}$					
Only enter if IRFC= $5$ and PPV>= $3$ .					
Float	Required if IRFC=5	·			

### 3.5 Under-relaxation coefficients

Several important time under-relaxation coefficients used in CTF have been exposed to the user as optional key/value pairs. If you leave the following lines out, all under-relaxation coefficients will be set to their defaults. It is not recommended these coefficients be used unless there is an issue with converging the code and other options for stabilizing the solution have failed (e.g., turning on the conduction solve and adjusting RTWFP). It is not recommended these coefficients be used at all for transients, as they will impact the transient solution.

The under-relaxation coefficients are used to perform logarithmic interpolation.

$$x^{n} = (x^{n})^{\beta} * (x^{n-1})^{1-\beta}$$
(3.1)

The under-relaxation coefficient must be entered as a key/value pair. The key is the name of the coefficient surrounded by curly braces. The value is the coefficient, which shall be between 0.0 and 1.0. The key and

value shall be seperated by whitespace.

1.6	beta_entr	[-]	[-]	
Under-relaxation coefficient for the droplet entrainment rate. Default 0.3.				
Float	Optional			
1.6	beta_deentr	[-]	[-]	
Under-relaxation coefficient	nt for the droplet de-entrain	ment rate. Default 0.1.		
Float	Optional			
1.6	beta_inthtc	[-]	[-]	
Under-relaxation coefficient	nt for the interfacial heat tr	ansfer coefficient. Default	0.1.	
Float	Optional			
1.6	beta_int_vl	[-]	[-]	
	nt for the interfacial vapor/	liquid drag. Default 0.3.		
Float	Optional			
	1			
1.6	beta_int_ve	[-]	[-]	
	nt for the interfacial vapor/	droplet drag. Default 0.1.		
Float	Optional			
1.0		r 1		
1.6	beta_int_vl_gap	[-]	[-]	
	nt for the interfacial vapor/	liquid drag in gaps. Defaul	lt 0.1.	
Float	Optional			
1.6	hata int no non	[]	[]	
	beta_int_ve_gap	[-]	[-]	
	nt for the interfacial vapor/	droplet drag in gaps. Defa	ult 0.1.	
Float	Optional			
1.6	beta_htc	[-]	[-]	
	nt for the solid/fluid heat tr			
		ansier coemcient. Default	U.2.	
Float	Optional			

## 3.6 Notes on Card Group 1

The initial mass flow rate is set to 0.0 by default. Therefore, a time-dependent  $\mathring{\text{aA}}\text{IJramp}\mathring{\text{aA}}\mathring{\text{I}}$  for the inlet mass flow rate should be applied (see description in CARD GROUP 13). Variables PREF and HIN are used to determine the initial properties of the fluid in computational domain. Because the mass conservation equation for non-condensable gases is not actively solved in this version of CTF, the volume fraction of vapor in the vapor-gas mixture should not be less than 0.9999! Variables BRIN and RDIF are only read if the boron tracking/precipitation model is activated (IBTM  $\neq$  0). RDIF is only used when IBTM = 2.

CHAPTER 4			
			_CARD GROUP 2
	y subroutine READ_CARD_2.	9	
The first line indicates the	e group number: NGROUP =	2	
2.1	NCHANL	[—]	[-]
Total number of subchan			
Integer	Required		
2.1	NDUM2:NDUM14	[—]	[—]
Not used, but entry is ob 0 - Suggested value	ligatory:		
Integer	Required		
Cards 2.2 and 2.3 are read CARD 2.2: I, AN(I), P	d in pairs NCHANL times.	NAMGAP(I), X(I), Y(I),	XSIZ(I), YSIZ(I)
2.2	I	[—]	[—]
Index of subchannel			
Integer	Required		
2.2	AN(I)	$[m^2]$	$[in^2]$
Nominal channel area	AN(1)	[111 ]	[111 ]
Float	Required		
2 2000	Tooquirou		
2.2	PW(I)	[m]	[in]
Wetted perimeter	1	•	
Float	Required		

2.2	ABOT(I)	$[\mathrm{m}^2]$	[in <sup>2</sup> ]
	om of the channel for use in	the momentum equation. If A	BOT(I) is entered as zero, it is
set to AN(I).			
Float	Required		
2.2	ATOP(I)	$[\mathrm{m}^2]$	$[in^2]$
Area of the top of AN(I).	of the channel for use in the m	nomentum equation. If ATOP(I	is entered as zero, it is set to
Float	Required		
2.2	NAMGAP(I)		
0.1		of channel I convects transvers	se momentum between sections.
Integer	Required		
2.2	X(I)	D GROUP 2.2 if MESH $= 1$ (se $[m]$	[in]
The X location o	f the geometric center of char		
Float	Conditional - if MES	$\mathtt{SH} = 1$	
2.2	Y(I)	[m]	[in]
The Y location o	f the geometric center of char	nnel I	
Float	Conditional - if MES	SH = 1	
2.2	XSIZ(I)	[m]	[in]
The size of chann	nel I in the X direction		
Float	Conditional - if MES	$\mathtt{H}=1$	
2.2	YSIZ(I)	[m]	[in]
	nel I in the Y direction	[m]	[in]
		ırr 1	
Float	Conditional - if MES	H = T	

Note 1: Do not enter PW(I) equal to zero, because in that case the calculated hydraulic diameter of the channel becomes infinite.

Note 2: In case of more than one axial sections, NAMGAP(I) has to be specified differently from zero and CARD 2.3 has to be specified, too, in order to get a correct momentum transfer in the last axial cell of each section. In case of only one axial section, the input values of CARD 2.3 are ignored. They are created automatically by the code. That means, NAMGAP(I) can be set to zero and CARD 2.3 can be omitted. (The procedure was not yet tested for reverse flow conditions!)

Note 3: For the meshing information, the X and Y positions should specify the geometric center of the channel (the channel is assumed rectangular in shape). The XSIZ and YSIZ terms should create a box that completely covers the bounds of the channel. The X and Y positions should NOT be the flow area centroid. The selected origin of the model is up to the user. The X and Y terms can be positive or negative. These terms are used for visualizing data in the VTK file and will not impact the code solution. This information may be omitted by setting MESH=0 on Card 1.1. See Figure 4.1 for an example of these terms in a graphical way. A group of 9 subchannels is shown, setup as a single assembly. The zero location is in the lower-left. The four parameters are defined for channel 5. Similarly, an example is given for a quarter core model (see

Figure 4.2).

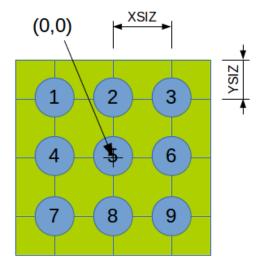


Figure 4.1: Definition of the X, Y, XSIZ, and YSIZ terms for Card 2.2 for a single assembly, plus a suggestion of where the origin can be placed

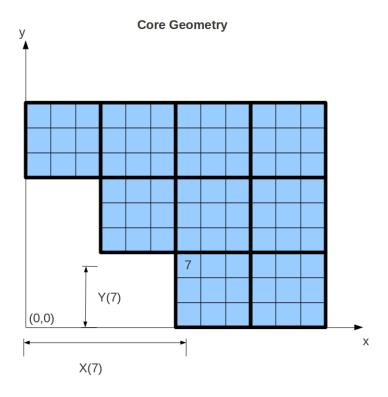


Figure 4.2: Definition of the (0,0) location and X and Y for core geometry

Card 2.3: INODE(I,N), KGAPB(I,N), KGAPA(I,N); N = 1:NAMGAP(I)

Card 2.3 is only read if NAMGAP(I) >0 for channel I.

2.3	INODE(I,N)	[—]	[—]	
The index number of the n	ode where the vertical velo	city of channel I convects to	ransverse momentum across	
a section boundary.				
			or the top of the channel,	
`	, ·	•	e section containing channel	
I. INODE is defined in the	section where the vertical r	nomentum equation is solve	ed.	
Integer	Conditional - if NAMGAP(I)	>0 for channel I		
2.3	KGAPB(I,N)	[—]	[-]	
The index number of the	gap below the section boun-	dary:		
0 I(1) : 1.1	.1 .: 1 1			
0 - If there is no gap belo	ow the section boundary			
Note: If KGAPB $\neq 0$ , the p	positive velocity of channel	I at INODE(I,N) convects	transverse momentum from	
KGAPB into KGAPA. The ne	egative velocity of channel	I at INODE(I,N) convects	transverse momentum from	
KGAPA into KGAPB.				
If $KGAPB = 0$ , this momen	tum is dissipated.			
Integer	Conditional - if NAMGAP(I	>0 for channel I		
	1			
2.3	KGAPA(I,N)	[—]	[—]	
The index number of the	gap above the section boun	dary:		
O If there is no gan abo	we the coetion houndamy			
0 - If there is no gap abo	ve the section boundary			
Note: If $KGAPB \neq 0$ , the positive velocity of channel I at INODE(I, N) convects transverse momentum from				
${\it KGAPB}~(if~{\it KGAPB} \neq 0)~into~{\it KGAPA}.~The~negative~velocity~of~channel~I~at~INODE(I,N)~convects~transverse$				
momentum from KGAPA in	momentum from KGAPA into KGAPB.			
If $KGAPA = 0$ , this momen	tum is dissipated.			
Integer	Conditional - if NAMGAP(I)	>0 for channel I		
Card 2.4: I, CHAN_GLOBAL, CHAN_OWNER				
Card 2.4 is only read if thi	s is a parallel model with n	nultiple CTF input decks.		
2.4	Т	[_]		
The channel local index (c	_		L J	
	· · · · · · · · · · · · · · · · · · ·	11 1		
Integer	Conditional - if this is a pa	arallel run		
0.4	GILAN GLODAT	[ ]	[ ]	
2.4	CHAN_GLOBAL			
The channel global index.				
Integer	Conditional - if this is a pa	arallel run		

2.4

Integer

Conditional - if this is a parallel run

CHAN\_OWNER

The index of the domain that owns this channel.

CHAPTER 5_			
ı			
			_CARD GROUP 3
<u></u>			_CAND GROUT 3
T1.	11 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1		
This card group is re	ead by subroutine READ_CARD_3.		
This card is omitted	if there are no transverse connection	ctions between channels.	
The first line indicat	tes the group number: NGROUP =	3	
	-		
CARD 3.1: NK, NDU NDUM13, NDUM14	M2, NDUM3, NDUM4, NDUM5, ND	UM6, NDUM7, NDUM8, NDUM9	, NDUM10, NDUM11, NDUM12,
3.1	NK	[]	[—]
Total number of gap	os (transverse connections betwee	en subchannels)	
Integer	Required		
3.1	NDUM2: NDUM14		[—]
Not used, but entry	is obligatory:	L J	
0 - Suggested value	2		
0 10 1000 10 10 10 10 10 10 10 10 10 10			
Integer	Required		
r. GADD a a	1 CARR 0 0		
Lines <u>CARD 3.2</u> and	d CARD 3.3 are read in pairs NK	times.	
	(K), JK(K), GAPN(K), LENGTH(	(K), WKR(K), FWALL(K), IC	GAPB(K), IGAPA(K), FACTOR(K
(IGAP(K,N), JGAP(	K,N), N = 1:3)		
3.2	K		[—]
Index of gap	1	-	
Note: Gap indices r	nust be entered sequentially from	1 to NK. Skipping numbers i	s not permitted.
Integer	Required		

3.2	IK(K)	[—]	[—]
Identification number	er of the <i>lower</i> -numbered	d subchannel of the pair connected	ed to gap K
Integer	Required		
3.2	JK(K)	[—]	[—]
Identification number	er of the higher-numbere	ed subchannel of the pair connect	sed to gap K
Integer	Required		
3.2	GAPN(K)	[m]	[in]
Nominal gap width			
Float	Required		
3.2	LENGTH(K)	[m]	[in]
		(and the center of subchannel J	. ,
Float	Required	and the center of subchanner 3	
rioat	Required		
3.2	WKR(K)	[m]	[in]
	loss coefficient (velocity	. ,	
		whole gap only. It cannot be s	pecified node-wise in the axial
direction.	1 · · · · · · · · · · · · · · · · · · ·		
Float	Required		
3.2	FWALL(K)	[-]	[—]
Wall friction factor	for the gap:		
0.0 - No walls			
0.5 - One walls			
1.0 - Two walls			
77 / 4 11 1 .	. 11 76	11 . 1	of the control of
	ng a gap is no waii. If on e perpindicular to the ga	the wall is determined, a wall with the	the surface area of LENGIH(K)
Float	Required	<i>p</i> .	
rioat	rtequired		
3.2	IGAPB(K)	[—]	[-]
	e gap in the section below	w gap K	[ ]
	-	<b>.</b>	
0 - If there is no ga	ap below gap K		
The velocity of IGAR	PB(K) convects vertical 1	momentum at node 1 into (or out	t of) channel IK(K) and out of
(or into) JK(K).			
Integer	Required		

3.2	IGAPA(K)	[—]	[—]
Index number of the gap is	n the section above gap K		
0 - If there is no gap about	ve gap K		
The velocity of IGAPA(K) of IK(K) and out of (or into)		at the top node of the sect	tion into (or out of) channel
Integer	Required		

 $\underline{Note}$ : The input for FACTOR, IGAP and JGAP is required only if the three-dimensional form of the transverse momentum equation is desired.

3.2	FACTOR(K)	[—]	[—]	
Gap orientation flag:				
the global coordinate $(default)$	`		direction as positive flow for ate system	
Float	Required			
3.2	IGAP(K,N)	[—]	[—]	
Gap numbers facing the IK(K) side of gap K:				
-1 - If the gap faces a wall				
- $(50+IK(K))$ - For gaps facing a pressure source boundary condition, enter the negative sum of 50 plus the channel that has the pressure source boundary condition.				
-(40+IK(K)) - For gaps facing a mass source boundary condition, enter the negative of the sum of 40 and the channel that has the mass source boundary condition (note that since the value can only be between 40 and 50, this means mass flow boundary conditions can only be applied to channels 1 through 10).				
**Note: Three sets of IGAP, KGAP are required in Card 3.2. If there are fewer gaps to specify, then enter 0 until the end of the line is reached**				
Integer	Required			
	1			
3.2	JGAP(K,N)	[—]	[—]	

Gap numbers facing the JK(K) side of gap K:

- -1 If the gap faces a wall
- -(50+JK(K)) For gaps facing a pressure source boundary condition, enter the negative sum of 50 plus the channel that has the pressure source boundary condition.
- -(40+JK(K)) For gaps facing a mass source boundary condition, enter the negative of the sum of 40 and the channel that has the mass source boundary condition (note that since the value can only be between 40 and 50, this means mass flow boundary conditions can only be applied to channels 1 through 10).

\*\*Note: Three sets of IGAP, KGAP are required in Card 3.2. If there are fewer gaps to specify, then enter 0 until the end of the line is reached\*\*

	Integer	Required
--	---------	----------

3.3	GMULT(K)	[—]	[—]		
Number of actual gaps mo	Number of actual gaps modeled by gap K				
There are no internal gap	s in a lumped subchannel.	Only gaps lying on the	interface to the neighboring		
lumped subchannel are reg	garded.				
Integer	Required				
3.3	ETANR(K)	[—]	[—]		
Crossflow de-entrainment	(deposition) fraction				
Float	Required				

CARD 3.3.5 is read only if meshing information is being provided (MESH = 1 from CARD 1.1). It is read NK times (once for each gap in the model).

 $\underline{\text{CARD } 3.3.5}$ : K, X(K), Y(K), NORM; K = 1:NK

3.3.5	K	[—]	[—]	
Gap Number				
Integer	Conditional - if $MESH = 1$			
3.3.5	X(K)	[m]	[in]	
X location of the center of the momentum cell that makes up the gap (it will be on the face between subchannels that are connected by the gap). See Figure 4.1 for how the global coordinate system is setup.				
Float	Conditional - if $MESH = 1$			
3.3.5	Y(K)	[m]	[in]	
Y location of the center of the momentum cell that makes up gap K				
Float	Conditional - if $MESH = 1$			
3.3.5	NORM	[—]	[—]	
The direction that flow through the gap travels (it will be either 'x' or 'y' and the orientation is with regards				

#### CARD 3.4: NLMGAP

 ${\bf Character}$ 

3.4	NLMGAP	[—]	[—]
This is required only for the	ect orthogonal transverse me three-dimensional formula form of the transverse me	lation of the transverse mo	-
Integer	Required		

Card 3.5 is read NLMGAP times.

This means,  $\underline{\text{only}}$  in the case that the  $\underline{\text{three-dimensional form of the transverse momentum equation}}$  is applied.

 $\underline{\mathrm{CARD}\ 3.5}$ : KGAP1(N), KGAP2(N), KGAP3(N); N = 1:NLMGAP

to the X and Y directions that are shown in Figure 4.1)

Conditional - if  $\mathtt{MESH} = 1$ 

One set of KGAP1, KGAP2, KGAP3 entries are specified per line. Enter a new line for each set.

3.5	KGAP1(N)	[—]	[—]
Index number of the gap	whose velocity transports to	cansverse momentum from	one gap to another
Integer	Conditional - if NLMGAP $\neq$	0	
3.5	KGAP2(N)	[—]	[—]
Index number of the gap which receives the transverse momentum convected by the positive velocity of gap $KGAP1$ A number $\neq 0$ must be entered.			
Integer	Conditional - if NLMGAP $\neq$	0	
3.5	KGAP3(N)	[—]	[—]
Index number of the gap that the positive velocity of KGAP1 transports transverse momentum out of. (The positive velocity of KGAP1 transports momentum from KGAP3 to KGAP2. Note: The negative velocity of KGAP1 will transport transverse momentum in the opposite direction; i.e., from KGAP2 into KGAP3.)  A number $\neq 0$ must be entered.			
Integer	Conditional - if NLMGAP $\neq$	0	

Card 3.6 is read only if this is a parallel model with one CTF input file per domain.

#### CARD 3.6: K, GAP\_GLOBAL, GAP\_OWNER

One set is entered per line.

3.6	K	[—]	[—]
Gap index number (consis	tent with 3.2 above).		
Integer	Conditional if this is a parallel run		
3.6	GAP_GLOBAL	[—]	[—]
Global index of the gap in the model.			
Integer	Conditional if this is a par	allel run	
3.6	GAP_OWNER	[—]	[—]
Index of the domain that owns this gap.			
Integer	Conditional if this is a parallel run		

CHAPTER 6			
			_CARD GROUP 4
m: 1 . 11	l ' DEAD GADD A		
	subroutine READ_CARD_4.		
The first line indicates the	group number: NGROUP =	4	
	M, IREBAL, NDUM4, NDUM5	, NDUM6, NDUM7, NDUM8,	NDUM9, NDUM10, NDUM11,
NDUM12, NDUM13, NDUM14			
4.1	NSECTS	[—]	[—]
Number of sections			
Integer	Required		
4.1	NSIM	[—]	[—]
Number of simultaneous s	solution groups:		
1 - Direct Gaussian elim	ination (for pressure equation	on)	
<1 - Iterative Gauss-Seio	del solver (for pressure equa	ation)	
	( · · · · · · · · · · · · · · · · · · ·		
Integer	Required		
4.1	IREBAL	[_]	[_]
	eration control (acceleration	of convergence):	
0 - No rebalancing	(		
1 - Rebalancing			
$oxed{Note:  ext{If NSIM} = 1,  ext{IREBA}}$	L is set to 0.		
Integer	Required		

4.1	NDUM4:NDUM14	[—]	[—]
Not used, but entry is obligatory:			
0 - Suggested value			
Integer	Required		

<u>Note</u>: Within a simultaneous solution group, different solvers can be selected by means of parameter ISOL in  $CARD\ 1.1$ . But in the case of the Krylov solver (ISOL = 1) it is not recommended to divide the computational domain into more than one simultaneous solution group.

Cards  $\underline{\text{CARD 4.2}}$ ,  $\underline{\text{CARD 4.3}}$ , and  $\underline{\text{CARD 4.4}}$  are read in a group NSECTS times.

CARD 4.2: ISEC, NCHH, NONODE, DXS(ISEC,1), IVARDX

4.2	ISEC	[—]	[—]	
Section number				
Begin with section numbe	r 1 on the bottom of the ve	essel and proceed towards	the top, incrementing ISEC	
by 1.				
Integer	Required			
4.2	NCHN	[—]	[—]	
Number of channels in sec	tion ISEC			
Integer	Required			
4.2	NONODE	[—]	[—]	
Number of vertical levels	(continuity mesh cells without auxiliary lower or upper boundary cells) in section			
ISEC				
Integer	Required			
4.2	DXS(ISEC,1)	[m]	[in]	
Vertical mesh cell length i	n section ISEC			
Float	Required			
4.2	IVARDX	[—]	[—]	
Flag for variable node length in section ISEC:				
0 - Constant mesh cell le	ngth (default)			
$>0$ - Read IVARDX pairs in variable $\Delta X$ table (see Card 4.3)				
Integer	Required			

Card 4.3 is only read if IVARDX >0.

Up to 5 sets of (JLEV, VARDX) may be entered. If IVARDX is greater than 5, repeat CARD 4.3 until IVARDX pairs has been entered. If IVARDX is less than 5 enter (0; 0.0) until 5 sets of (JLEV, VARDX) have been entered.

 $\underline{\mathrm{CARD}}\ 4.3$ : JLEV(I), VARDX(I); I = 1:IVARDX

4.3	JLEV(I)	[—]	[—]
Last axial level in section ISEC to have a node length of VARDX(I)  JLEV(IVARDX) must be greater than or equal to NONODE + 1			
Integer	Conditional - if IVARDX >0		
4.3	VARDX(I)	[m]	[in]
Axial mesh cell length			
Float	Conditional - if IVARDX >0		

Card 4.4 is read NCHN times for section ISEC. If PPV>=7 on Card 1.1, then you must specify how many channels are connected above and how many are connected below. If PPV<7, a hard-coded number of 6 above and below connections must be specified. The PPV>=7 option allows the user to connect a channel to any arbitrary number of channels above or below. Keep the line length below the maximum allowed length by entering large numbers of channels on new lines.

 $\underline{\mathrm{CARD}}\ 4.4$ : PPV<7: I, KCHANA(I,J); J = 2:7, KCHANB(I,J); J = 2:7 PPV>=7: I, NCHA, KCHANA(I,J); J = 2:NCHA, NCHB, KCHANB(I,J); J = 2:NCHB

4.4	I	[—]	[—]	
Identification number of a	channel in section ISEC			
Integer	Required			
4.4	NCHA	[—]	[—]	
Only enter if PPV>=7. S	Only enter if PPV>=7. Specifies the number of channels connected to this one in the section above.			
Integer	Enter if PPV>=7			
4.4	KCHANA(I,J)	[—]	[—]	
Indices of channels in the section above ISEC that connect to channel I. If channel I does not connect to any channels above, enter I in KCHANA(I,2). A maximum of 6 channels can be connected to channel I when PPV<7. If using PPV<7 and the number of channels connected to channel I is less than 6, enter 0 until 6 KCHANA values have been entered. When using PPV>=7, enter NCHA channels				
Integer	Required			
4.4	NCHB	[—]	[—]	
Only enter if PPV>=7. Number of channels connected to this one in the section below.				
Integer	Enter if PPV>=7			
4.4	KCHANB(I,J)	[—]	[—]	
Indices of channels in the section below ISEC that connect to channel I. If channel I does not connect to				
any channels below, enter I in KCHANB(I,2). A maximum of 6 channels can be connected to channel I if				
using PPV $<$ 7. If using PPV $<$ 7 and the number of channels connected to channel I is less than 6, enter 0 until 6 KCHANB values have been entered. Enter NCHB channels if using PPV>=7.				

CARD 4.5: IWIDE

Required

Integer

4.5	IWIDE	[—]	[—]	
Maximum difference between	Maximum difference between the index numbers of adjacent cells (vertical or transversal, respectively) in a			
simultaneous solution group (For the simplest case of only one simultaneous solution group, one section, and				
no sub-divided subchannels âĂŞ i.e. only one mesh cell per subchannel radial direction âĂŞ IWIDE is equa				
to the number of subchannels).				
Integer	Required			

 $\underline{\mathrm{CARD}}$  4.6:  $\underline{\mathrm{MSIM}}$ (I); I = 1: $\underline{\mathrm{NSIM}}$ 

Twelve values are entered per card. If  $\mathtt{NSIM}$  is greater than 12, repeat CARD 4.6 until  $\mathtt{NSIM}$  values have been entered.

4.5	MSIM(I)	[—]	[—]
Last cell number in each simultaneous solution group.			
This is the total number of continuity mesh cells within the simultaneous solution group I (number of			
subchannels times the number of vertical layers NONODE).			
Integer	Required		

CHAPTER 7	
I	
	CARD GROUP 5

This card group is read by subroutine SETIN.

Note: The input for this group allows the user to specify vertical variations in the *continuity area*, momentum area, or wetted perimeter for channels, and in the transverse width for gaps. It can be omitted, if such variations are not needed.

The first line indicates the group number: NGROUP = 5

CARD 5.1: NAFACT, NAXL, NDUM3, NDUM4, NDUM5, NDUM6, NDUM7, NDUM8, NDUM9, NDUM10, NDUM11, NDUM12, NDUM13, NDUM14

5.1	NAFACT	[—]	[—]	
Total number of geometry	variation tables			
Integer	Required			
5.1	NAXL	[—]	[—]	
Maximum number of pairs of entries in the vertical variation tables. See CARD 5.2				
Integer	Required			
5.1	NDUM3:NDUM14	[—]	[—]	
Not used, but entry is obligatory:  0 - Suggested value				
Integer	Required			

Card 5.2 is read NAFACT times. The tables are numbered sequentially in the code.

Eight pairs of (JAXL, AFACT) are entered per card. The tables are numbered sequentially in the code. All entered tables must have a full set of NAXL pairs of (JAXL, AFACT) data.

 $\underline{CARD\ 5.2}$ : JAXL(I,N), AFACT(I,N); N = 1:NAXL(I)

5.2	JAXL(I,N)	[—]	[—]		
Axial node number at whi	Axial node number at which to apply the geometry variation factor for table I, point N				
Integer	Required				
5.2	AFACT(I,N)	[—]	[—]		
Geometry variation factor for table I, point N					
Whether AFACT(I, N) is applied to channel areas, wetted perimeter, or gap width, is specified in CARD					
6.2.					
Float	Required				

CHAPTER {	3		
			_CARD GROUP 6
	is read by subroutine SETIN.	C	
	licates the group number: NGROUP = NDUM2, NDUM3, NDUM4, NDUM5, N		9, NDUM10, NDUM11, NDUM12
6.1	N1	[-]	[-]
Total number of	f channel and gap variation table ca	ards to be read in CARD 6.2	
Integer	Required		
6.1	NDUM2:NDUM14	[—]	[—]
	ntry is obligatory:	L J	
0 - Suggested v	value		
Integer	Required		
Card 6.2 is read	N1 times.  T, IAMT, IPWT, ICRG(M); M = 1:	12	
		r 1	r 1
6.2 Table number for	r geometry variation:		
	annel continuity area variation		
$<0$ - For $gap\ u$	width variation		
Integer	Required		

6.2	IAMT	[—]	[—]
Table number for subchan	nel momentum area variation	on:	
0 - If IACT $<$ 0			
Integer	Required		
6.2	IPWT	[—]	[—]
Table number for subchan	nel wetted perimeter variati	ion:	
0 - If IACT $<0$			
Integer	Required		
6.2	ICRG(M)	[—]	[—]
Index numbers of the subc	hannels (if IACT >0) or gap	os (if IACT < 0) that the tab	oles specified in IACT, IAMT,
and IPWT are to be applied	d to.		
Up to $12$ channels or gaps	may be specified per card.	If the number of specified	channels or gaps is greater
than 12, repeat CARD 6.2	until all values have been	entered. If the number of s	specified channels or gaps is
less than 12, enter 0 until	12 values have been entered	l.	
Integer	Required		

CHAPTER 9			
			_CARD GROUP 7
The first line indicates the	mitted if there are no space $\frac{1}{2}$ group number: $\frac{1}{2}$ NGROUP = $\frac{1}{2}$	7	D, IXFLOW, NDUM11, NDUM12,
7.1	NCD	[—]	[—]
	7.2 specifying local spacer passure losses are specified in		ese include vertical pressure
Integer	Required	CARD GROOT 3.	
	NGT  a special geometrical specification of the pressure loss coefficient		[—]
>0 - If spacers are defin	_	eled blockages in CARD 7.	3 through CARD 7.9 <b>or</b> if
Integer	Required		
7.1	IFGQF	[—]	[-]
Flag for grid quench front $\geq 1$ - On $< 1$ - Off			
Integer	Required		

7.1	IFSDRP	[—]	[—]	
Flag for small drop model	:			
≥ 1 - On				
< 1 - Off				
\ 1 - OII				
Integer	Required			
7.1	IFESPV	[—]	[—]	
Flag for grid convective en	nhancement:			
0 - Off				
1 - On for single phase va	apor only			
2 - On for both single-ph	ase vapor and liquid (multi	plied by the HTC calculat	ed by Dittus-Boelter only)	
			v v	
	option with grid enhancementer CARD 7.3 through CAR		e grid enhancement locations	
on CARD 7.1.6	iter eriteb 7.9 tillough erit	the r.s. instead, speeny the	grid emancement locations	
Integer	Required			
	T			
7.1	IFTPE	[—]		
Flag for two-phase enhance	ceent of dispersed flow heat	transfer:		
≥ 1 - On				
< 1 - Off				
Integer	Required			
	Tooquirou			
7.1	IGTEMP	[—]	[—]	
Flag for grid quench calcu	lations (see CARD 7.8):			
≥ 1 - On	,			
< 1 - Off				
T., 4	Di i			
Integer	Required			
7.1	NFBS	[—]	[—]	
Number of flow blockages	1120	l J	L J	
Integer	Required			
inveger	recquired			
7.1	IFCD	[—]		
Flag for internal calculation of grid pressure loss coefficients:				
≥ 1 - On				
< 1 - Off				
Integer	Required			
11100gc1	LIWGUIICU			

7.1	IXFLOW	[—]	[—]	
Flag for modeling lateral-e	exhange cross-flow effects ca	aused by the spacer grids (	see CARD 7.9):	
0 - No special modeling	0 - No special modeling			
1 - Directed cross-flow m	odel			
2 - Enhanced turbulence	mixing model			
3 - Both models included	3 - Both models included			
Integer	Required			
7.1	NDUM11:NDUM14	[—]	[—]	
Not used, but entry is obligatory:				
0 - Suggested value				
Integer	Required			

Card 7.1.5 is read **only** if IFESPV = 1, 2, or 3. The card is read NGT times. The terms C1, C2, C3, C4, A, and PHI in Card 7.1.5 refer respectively to the coefficients a, b, c, d, A, and  $\Phi$  in

$$\frac{Nu}{Nu_0} = \left[1 + a\epsilon^2 \exp\left[b\frac{x}{D}\right]\right] \left[1 + A^2 \tan^2(\Phi) \exp\left[c\frac{x}{D}\right]\right]^d. \tag{9.1}$$

This equation represents the mixing-vane form of the Yao-Hochreiter-Leech formula, with the original reference values for all coefficients shown in the Card 7.1.5 input description. However, it has been observed that the mixing-vane component of the equation does not produce accurate results. Therefore, it is suggested that the C3, C4, A, and PHI terms be set to 0.0 to disable that portion of the correlation. Additional details on the Yao-Hochreiter-Leech formula and its implementation in CTF are given in the CTF Theory Manual [4].

SIMPLE GRID MODELING: If not requiring the advanced models offered by Cards 7.3 through 7.5, the simple grid modeling option can be utilized by setting IFESPV to 3 on Card 7.1, setting grid type parameters on Card 7.1.5, setting grid enhancement locations on Card 7.1.6, and entering grid loss coefficients on Card 7.2.

Card 7.1.5 is read NGT times (if IFESPV > 0). Enter parameters for each grid on a separate line. Line 1 will be Grid ID 1, line 2 will be Grid ID 2, and so on.

 $\underline{\text{CARD 7.1.5}}$ : C1(N), C2(N), C3(N), C4(N), A(N), PHI(N), N = 1:NGT

7.1.5	C1(N), C2(N), C3(N), C4(N)	[—]	[—]	
Coefficients in the Yao-Hochreiter-Leech formula				
C1 - Reference value is 5.55				
C2 - Reference value is -0	).13			
C3 - Reference value is -(	0.034			
C4 - Reference value is 0	.4			
Float[4]	Conditional - if IFESPV >0	0		
7.1.5	A(N)	[—]	[—]	
The blockage ratio normal	to the flow caused by the	mixing vanes		
Float	Conditional - if IFESPV >0		-	
7.1.5	PHI(N)	[degrees]	[degrees]	
Angle of the vanes with re	espect to the flow direction			
Float	Conditional - if IFESPV >0	0	_	
$\it Note :  ext{If IFESPV} = 3,  ext{ the f}$	ollowing parameters must b	oe included in Card 7.1.5 a	s well:	
7.1.5	ABLOC(N)	[—]	[—]	
Blockage ratio of the grid	straps			
Float	Conditional - if IFESPV =	3		
7.1.5	SPBLOC(N)	[—]	[—]	
Blockage ratio of the grid	springs			
Float	Conditional - if IFESPV =	3		
Card 7.1.6 is read only if IFESPV = 3. It specifies the location of each enhancement in the model. CTF will continue reading lines of Card 7.1.6 until it encounters a negative number on a new line.  CARD 7.1.6: GRIDTYPE, J, CHANNEL(M); M = 1:12				
7.1.6	GRIDTYPE	[—]	[—]	
The grid type ID of this g	rid. Data for each grid type	e were entered on Card 7.1	.5.	
Integer	$oxed{ ext{Conditional - if IFESPV}} =$	3		
7.1.6	J	[—]		
The axial momentum cell in which this grid type is located				
Integer	Conditional - if IFESPV =	3		
710	GHANNEL (M)	r 1	Г	
7.1.6	CHANNEL (M)	10.1.1.2		
	s grid type is located. Ente exial level, enter 0 until 12 e		If the grid resides in fewer	
Integer		3		

\*\*Note: A negative number **must** be entered on a new line in order to terminate Card 7.1.6.

CARD 7.2 is read NCD times. Do not enter this card if IFESPV = 1 or 2. Enter it if IFESPV = 0 or 3. Note that if IFESPV is used, a loss coefficient for each location in the model that contains a grid type that will cause a mixing enhancement in the flow  $\mathbf{MUST}$  be entered. For example, if Grid ID 1 at axial level 5 is placed in channel 3, a loss coefficient in axial level 5 and channel 3 that corresponds to that grid type must be entered. While fewer loss coefficients than there are enhancements to the flow cannot be entered, it is permissible to enter more loss coefficients than there are enhancements.

 $\underline{\mathrm{CARD}}$  7.2: CDL, J,  $\underline{\mathrm{ICDUM}}(\underline{\mathrm{I}})$ ;  $\underline{\mathrm{I}}$  = 1:12

7.2	CDL	[—]	[—]	
Pressure loss coefficient (vertical) of spacer				
Float	Conditional - if IFESPV $= 0$ or $3$			
7.2	J	[—]	[—]	
	Node number at which the pressure loss coefficient is applied.  Note: The vertical node number is relative to the beginning of the section containing the channel(s) listed in ICDUM(I).			
Integer	Conditional - if IFESPV = $0 \text{ or } 3$			
7.2	ICDUM(I)	[—]	[—]	
Index number of channel to which the pressure loss coefficient will be applied at axial node J. Up to 12 channels may use the specified pressure loss coefficient CDL at vertical node J. If the number of channels is fewer than 12, enter 0 until 12 values have been entered.				
Integer	${\bf Conditional - if \ IFESPV} =$	0 or 3		

Note: If pressure loss coefficients CDL are specified in CARD 7.2, then CARD 7.3 through CARD 7.9 must be omitted. (Pressure losses are specified either by  $\zeta$  values or by geometrically modeled flow blockages.)

CARD 7.3, CARD 7.4, and CARD 7.5 are read in groups NG times, where NG = 1:NGT. Only enter this group if IFESPV = 1 or 2.

CARD 7.3: ING, NGAL(NG), NGCL(NG), IGMAT(NG), GLOSS(NG), GABLOC(NG), GLONG(NG), GPERIM(NG), SPBLOC(NG), TPROBE(NG)

7.3	ING	[—]				
Grid type number (must be sequential starting with 1)						
Integer	Conditional - if IFESPV	Conditional - if $IFESPV = 1$ or 2				
7.3	NGAL(NG)	[—]	[—]			
Number of axial locations for grid type ING Maximum = 16						
Integer	Conditional - if IFESPV	Conditional - if IFESPV $= 1$ or $2$				
7.3	NGCL(NG)	[—]	[—]			
Number of channels containing grid ING at levels NGAL						
Integer	Conditional - if IFESPV	Conditional - if IFESPV $= 1$ or $2$				

7.3	IGMAT(NG)	[—]	[—]			
	corresponding to material t	ypes in CARD GROUP 10	L J			
Integer Conditional - if IFESPV = $1 \text{ or } 2$						
7.3	GLOSS(NG)	[—]	[—]			
Pressure loss coefficient m	ultiplier:					
1.0 - Suggested value for	round-egde grids					
1.4 - Suggested value for	square-eage grids					
Float	Conditional - if IFESPV =	1 or 2				
11000		1 01 <b>2</b>				
7.3	GABLOC(NG)	[-]	[—]			
Fraction of channel area b	locked by grid					
Float	$\overline{\text{Conditional - if IFESPV}} =$	1 or 2				
7.3	GLONG(NG)	[m]	[in]			
Grid length						
Float	Conditional - if IFESPV =	1 or 2				
		r 1				
7.3	GPERIM(NG)	[m]	[in]			
Grid perimeter						
Float	Conditional - if IFESPV =	1 or 2				
7.3	SPBLOC(NG)	[_]	[_]			
		channel flow area	L J			
Float	Ratio of the area blocked by the grid springs to the channel flow area  Float   Conditional - if IFESPV = 1 or 2					
7.3	TPROBE(NG)	[s]	[s]			
Time scale for grid quench	l					
Float	Conditional - if IFESPV =	1 or 2				
7.3	GTHICK(NG)	[m]	[in]			
Thickness of the straps						
Float	Conditional - if IFESPV =	1 or 2				
CARD 7.4: NNGL(NG, NN)	$NN = 1 \cdot NGAT$					
OMED 1.4. MAGE(NG, NN), NN - 1.NGAL						
7.4	NNGL(NG, NN)	[—]	[—]			
Axial node number of mor	mentum cells containing gri	d type ING				
Integer	Conditional - if IFESPV =	1 or 2				

Card 7.5 is read NGCL(NG) times. Up to 6 sets of NGROD, NGSURF may be entered. If the number of rods with surfaces surrounding a grid is less than 6, enter (0; 0) until 6 sets have been entered.

 $\underline{\mathrm{CARD}\ 7.5}$ : NCNGL(NG, M), GMULT(NG, M), NGROD(NG, M, L), NGSURF(NG, M, L); L = 1:6; M = 1:NGCL(NG)

7.5	NCNGL(NG, M) [—] [—]
Channel ID number with a	grid type ING at axial levels NNGL(NG, NN) as specified
Integer	Conditional - if $IFESPV = 1$ or 2
7.5	GMULT (NG, M) [—] [—]
Number of grids contained	d in channel NCNGL(NG, NN)
Integer	Conditional - if IFESPV = 1 or 2
7.5	NGROD(NG, M, L) [—] [—]
Whole rod number with su	urface surrounding grid
Maximum of 6	
Integer	Conditional - if IFESPV = 1 or 2
7.5	NGSURF(NG, M, L) [—] [—]
Rod surface index of whole	e rod NGROD(NG, M, L) surrounding grid ING
Note: The average temper	rature of all surfaces surrounding grid is used to transport heat between grid and
heater rods.	
Integer	Conditional - if IFESPV $= 1$ or $2$

Card 7.6 and 7.7 are read NFBS times.

 $\underline{\mathrm{CARD}\ 7.6}$ : IBS, IFB(IBS), JSFB(IBS), NRFB(IBS), SPOINT(IBS), DSEP(IBS), THROAT(IBS), AFLBLK(IBS), CDFB(IBS), ABLOCK(IBS); IBS = 1:NFBS

7.6	IBS	[—]	[—]				
_	Flow blockage index number						
(Must be sequential from	1 to NFBS)						
Integer	Conditional - if NFBS $>0$						
7.6	IFB(IBS)	[—]	[—]				
Channel index number							
Integer	Conditional - if NFBS $>0$						
7.6	JSFB(IBS)	[—]	[—]				
Axial node index number							
Integer	Conditional - if NFBS $>0$						
7.6	NRFB(IBS)	[—]	[—]				
Number of rods that bloc	k this channel						
Integer	Conditional - if NFBS $>0$						
	· · · · · · · · · · · · · · · · · · ·						
7.6	SPOINT(IBS)	[m]	[in]				
Axial position of the flow	separation point						
Float	Conditional - if NFBS >0						
	•						

7.6	DSEP(IBS)	[m]	[in]
Channel diameter at sepa	ration point		
	D =	$\sqrt{\frac{4A}{\pi}}$	(9.2)
		$V \pi$	(- /
Float	Conditional - if NFBS $>0$		
7.6	THROAT(IBS)	[m]	[in]
Diffuser diamter at exit	1111/0111 (126)	[111]	[111]
Float	Conditional - if NFBS >0		
7.6	AFLBLK(IBS)	[—]	[—]
Area for DBM, (fraction of	, ,		
Integer	Conditional - if NFBS $>0$		
7.6	CDFB(IBS)	[_]	[_]
Presure loss coefficient (R		l J	l J
Integer	Conditional - if NFBS >0		
*****			
7.6	ABLOCK(IBS)	[—]	[—]
Blockage area ratio			
Integer	Conditional - if NFBS $>0$		
CARD 7.7: NRODFB(IBS,	N), KRODFB(IBS, N), ANG	GIHT(IBS, N), ARAIHT(IB	S, N); N = 1:NRFB
7.7	NRODFB(IBS, N)	[—]	[—]
Index number of rod	<u> </u>	L J	L J
Integer	Conditional - if NFBS $>0$		
7.7	KRODFB(IBS, N)	[—]	[—]
Surface number	Tail and a second		
Integer	Conditional - if NFBS >0		
7.7	ANGIHT(IBS, N)	[degrees]	[degrees]
Angle for impact heat tra		[degrees]	[dogrees]
Float	Conditional - if NFBS >0		
7.7	ARAIHT(IBS, N)	$[m^2]$	$[in^2]$
Area for impact heat tran	, ,		
Float	Conditional - if NFBS $>$ 0		

Card 7.8 is read only if IGTEMP  $\neq 0$ . This option bypasses the original TGRID calculation.

 $\underline{\mathrm{CARD}\ 7.8}$ : TGRID(I), QFGRID(I); I = 1:IGTEMP

7.8	TGRID(I)	[°C]	[°F]			
Grid temperature						
Float	Float Conditional - if $IGTEMP \neq 0$					
7.8	QFGRID(I)	[m]	[in]			
Length of the quenched portion of the grid						
Float	Conditional - if IGTEMP $\neq 0$					

Card 7.9 is read only if IXFLOW  $\neq$  0. This option is for the spacer grid-induced cross-flow effects.

### $\underline{\mathrm{CARD}\ 7.9}$ VALUE\_ANGLE, NSETS\_XFLOW, NSETS\_SG\_MULT

7.9	VALUE_ANGLE	[degrees]	[degrees]		
Vane angle					
Float	Conditional - if IXFLOW $\neq$	0			
7.9	NSETS_XFLOW	[—]	[—]		
Number of data sets in in	put file xflow_data				
If $IXFLOW = 1$ or 3, then I	$VSETS\_XFLOW should be > 0.$				
Integer	Conditional - if IXFLOW $\neq 0$				
7.9	NSETS_SG_MULT	[—]	[—]		
Number of data sets in input file sg_mult_data					
If $IXFLOW = 2$ or 3, then $NSETS\_SG\_MULT$ should be $>0$ .					
Integer	Conditional - if IXFLOW $\neq$	0			

CHAPTER $10_{-}$					
			$\mathbf{C}$	ARD GROU	DΩ
L			OF	and Gnou.	ГО
This card group is read	by subroutine SETIN.				
	odeling any conductors, t		er (8) and Care	d 8.1, specifying	zero
	the group number: NGROU				
	ROD, NC, NRTAB, NRAD,		, NCAN, RADFL	G, W3CHF, IHTC	, DNBCH
8.1	NRROD	[—]		[—]	
Number of rods		,	,		
Integer	Required				
8.1	NSROD	[—]		[—]	
Number of unheated co	onductors		I		
Integer	Required				
8.1	NC	[—]		<u> </u>	— <sub> </sub>
Conduction model flag		L J		L J	
0 - No conduction					
1 - Radial conduction	and.				
	·				
2 - Radial and axial co					
3 - Radial, axial, and	azimuthal conduction				
	nductors, radial conductio	on is regarded at most			
Integer	Required				

0.1	NDMAD	Г 1		Г 1		
8.1	NRTAB		CLDD a a			
Number of temperature initialization tables to be read in CARD 8.6 through CARD 8.9:						
$\geq 1$ - Mandatory						
Integer	Required					
8.1	NRAD	[—]		[—]		
Number of radiation change	nels:					
0 - If no radiation is mod	${ m leled} \; ({ m if} \; { m RADFLG} = 0)$					
Integer	Required					
8.1	NLTYP	[—]		[—]		
Number of geometry types	s for the radiation calculati	ons:	<u>'</u>			
0 - If no radiation is mod	leled (if RADELC — 0)					
0 - II no radiation is mod	icica (ii itabi La = 0)					
Integer	Required					
	required					
8.1	NSTATE	[—]		[—]		
Flag for steady-state calculation of rod temperatures:						
0 Full transient calculat	ion (default)					
0 - Full transient calculation (default)						
> 0 - Steady-state solution	on of rod temperatures at t	he beginning of a calcula	ation			
	I					
Integer	Required					
0.1	NVD	F 1		Г 1		
8.1	NXF					
Number of time steps bety	ween radiation calculations:					
1 - Default						
Integer	Required					
8.1	NCAN	[—]		[—]		
Flag for the Yamanouchi o	canister quench model (spra	ay cooling after LOCA):				
$\leq 0$ - Off						
> 0 - On						
T., 4	Di 1					
Integer	Required					

8.1	RADFLG	[-]	[-]		
Flag for the radiation heat transfer calculations only:					
0 - Off					
≠ 0 - On					
2 - On, excluding fluid					
2 - On, excluding fluid					
Integer	Required				
0.1	HOGHE	r 1			
8.1	W3CHF				
Critical heat flux (CHF) c	alculation option:				
<0 - No correlation					
0 - Standard correlation (	(Biasi)				
1 - W3 correlation					
2 - Bowring correlation					
3 - Groeneveld look-up ta	ables				
When W3CHF>=0, a cal	culation is performed for	each conductor cell at each	ch timestep to determine if		
			erformed using CHF values		
interpolated from the mon	nentum mesh onto the rod	mesh, as described in the	CTF Theory Manual [4].		
Integer	Required				
8.1	IHTC	[—]	[—]		
	iling model. Whichever is	chosen will be used in bot	h sub-cooled and saturated		
boiling regions:					
0 - Chen					
1 - Thom					
Integer	Required				

8.1 DNBCHK [—] [—]

DNB check option:

-1 - Do not check DNB data

0 - Standard correlation (Biasi)

1 - W3 correlation

2 - Bowring correlation

3 - Groeneveld look-up tables

Has no effect when W3CHF>=0. This option is used if the W3CHF model is disabled. It calculates DNB data in the model prior to writing edits. When W3CHF is -1, it is not possible for the code to go into post-CHF heat transfer regimes; however, it is still useful to do a calculation of DNB data in the model in order to inform the user of the DNB margin. Because this check is only done prior to writing edits, rather than every iteration, the simulation cost is greatly reduced. This option is disabled when W3CHF>=0 because the DNB data is already being calculated using the correlation selected with W3CHF for each iteration. If either W3CHF>=0 or DNBCHK>=0, a special post-processing CHF calculation will be performed using CHF values interpolated from the scalar mesh onto the rod mesh, as described in the CTF Theory Manual [4].

Integer Required

8.1 [H] [H] [H]

This option is used to set the pellet thermal conductivity model. Negative numbers will apply one of the built-in burnup-dependent models. The selected model will be applied in all fuel rods in the model and initial burnup and gadolinia content in the rod will be zero. These options are intended to be used while CTF is coupled to an external code that will set burnup and gadolina. Option zero uses the default MATPRO-11 model, which has no burnup depedence. Option 1 means that Card 8.14 will be read in later to specify the local burnup and gadolina distribution in each rod in the model. This also allows different models to be used for different rods and axial locations. The specific model to be used will be set on Card 8.14.2 if using option 1.

- -5 Amaya correlation (MOX fuel)
- -4 Halden correlation (MOX fuel)
- -3 Duriez/Modified NFI correlation (MOX fuel)
- -2 Halden correlation (UO<sub>2</sub> fuel)
- -1 Modified NFI correlation (UO<sub>2</sub> fuel)
- 0 Do not read Card 8.14. The default fuel thermal conductivity model is MATPRO-11.

$$k_{uo2} = C \left\{ \max \left( \frac{2335}{464+T}, 1.1038 \right) + 7.027 \cdot 10^{-3} \exp \left( 1.867 \cdot 10^{-3} T \right) \right\}$$
  
where  $C = \frac{1-\beta(1-TD)}{1-0.05\beta}$  and  $\beta = 2.58 - \left( 5.8 \cdot 10^{-4} \right) T$ 

1 - Read Card 8.14.

Integer Required

## 10.1 Rod Geometry Data

Card 8.2, Card 8.3, and Card 8.4 are read NRROD times.

 $\underline{\mathrm{CARD}\ 8.2}$ : N, IFTYP(N), IAXP(N), NRENODE(N), DAXMIN(N), RMULT(N), HGAP(N), ISECR(N), HTAMB(N), TAMB(N), SYMROD(N), IHTCMAP, ITKEMAP, NSUBSECT, NSUBAX

8.2	N	[—]	[—]				
Index of rod Note: Rod indices must be entered sequentially, from 1 to NRROD. Skipping numbers is not permitted Note 2: A negative value of the rod index is used to flag a rod with inside connections to a fluid channel and cause Card 8.4 to be read.							
Integer	Required						
0.0	(a)	r 1	r 1				
8.2	IFTYP(N)						
Geometry type identification (Refers to CARD GROUP)	on number 9 for geometry type input	data)					
Integer	Required						
8.2	IAXP(N)	[—]	[—]				
(Refers to CARD 11.3 an	Axial power profile table identification number (Refers to CARD 11.3 and 11.4 for axial power profile input data). If left blank, a uniform axial power distribution will be assumed.						
Integer	Required						
8.2	NRENODE(N)	[—]	[—]				
Re-noding flag for heat transfer solution for rod N:  0 - No fine-mesh re-noding  > 0 - Re-noding every NRENODE(N) time steps  < 0 - Re-noding every NRENODE(N) time steps, based on inside surface temperatures							
Integer	Required						
8.2	DAXMIN(N)	[m]	[in]				
Minimum axial node size	DAVLITH(II)	[m]	[in]				
This is only used if NRENO	$\mathtt{DE} \neq \mathtt{0}.$						
Float	Required						
8.2	RMULT(N)	[—]	[—]				
-	(number of rods modeled b ling a partial rod, use the S	,					
Float	Required						

8.2	HGAP(N)	$\left[\frac{W}{m^2K}\right]$	$\left[\frac{BTU}{hrft^2 \circ F}\right]$
Constant gap con	iductance:		1
0 - If rod N does	not model a nuclear fuel rod		
This parameter is specified by their		rods that do not have the d	lynamic gap conductance mode
Float	Required		
o <b>o</b>	TOPOP (N)	r 1	r 1
8.2	ISECR(N)		[]
	sections containing rod N		
Integer	Required		
8.2	HTAMB(N)	$\left[\frac{W}{m^2K}\right]$	$\left[\frac{BTU}{hrft^2\circ F}\right]$
Heat transfer coe	fficient for heat loss to the an		y y
Note: This option	n is normally not related to the	he real fuel rod simulation. I	t is used in some simulations of
-	· ·	or the heat loss from a non w	vater-cooled rod to the ambience
or to the medium			
Float	Required		
8.2	TAMB(N)	[°C]	[°F]
	e for heat loss to ambience	[ ]	[1]
Float	Required		
11000	rtequired		
8.2	SYMROD(N)	[—]	[—]
symmetry line ru	ns through the rod. For examing will have a quarter rod (SY	ple, a quarter symmetry mod	s a partial rod (meaning that a el with a symmetry line running (SYMROD=0.5). Can only be
Float	Required if PPV=2		
model; it does no represents the sec	t need to start in Section 1. tion index; it is used only to	In the following documents show that an entry should be	start in any axial section in the ation for Cards 8.3 and 8.4, IS e made for each section. There y specifies to which channel the
8.2	IHTCMAP		[—]
	t high-fidelity heat transfer e	_	r map that will be used by this crud solutions. Set to 0 for no
Integer	Required if PPV=6		
0.0	TTUEMAD	r 1	[ ]
8.2	ITKEMAP		
	~		ap that will be used by this rod solutions. Set to 0 for no TKE
Integer	Required if PPV=6		

8.2	NUM_SUBSECT	[—]	[—]	
Only required if PPV>=6. Number of subsectors into which CTF rod sectors (coupling between rod surface				
		ement is used when an HTC	C or TKE map is applied to	
the rod only and only for	crud coupling.			
Integer	Required if PPV=6			
8.2	NUM_SUBAX	[—]	[—]	
Only required if PPV>=6. Number of sublevels into which CTF rod levels will be divided. Rod surface				
mesh refinement is used when an HTC or TKE map is applied to the rod only and only for crud coupling.				
Integer	Required if PPV=6			

Cards 8.3 and 8.4 are read in pairs ISECR(N) times. Note that a rod can start in any axial section in the model; it does not need to start in Section 1. In the following documentation for Cards 8.3 and 8.4, IS represents the section index; it is used only to show that an entry should be made for each section. There is no need to state the actual sections in which the rod exists. The user only specifies to which channel the rod connects.

CARD 8.3: NSCHC(IS, K), PIE(N, K); K = 1:8

Up to 8 sets of (NSCHC, PIE) may be entered. If the rod N is thermally connected to fewer than 8 channels, enter (0; 0.0) until 8 sets of (NSCHC, PIE) have been entered.

8.3	NSCHC(IS, K)	[—]	[—]			
Channel number with the	Channel number with thermal connections to rod N					
Integer	Required					
8.3	PIE(N, K)	[—]	[—]			
Azimuthal fraction of rod N thermally connected to channel NSCHC(K)						
(Integer, Float)	Required					

Up to 8 sets of (NSCHC,PIE) may be entered. If the rod N is thermally connected to fewer than 8 channels, enter (0; 0.0) until 8 sets of (NSCHC,PIE) have been entered.

Note: If a rod has internal fluid connections, as indicated by a negative value of the rod index N, then only zeros should be entered for channel numbers on the outside of the rod. This version does not support fluid connections on both sides of a heater rod.

Card 8.4 is read only if an inside surface for rod N exists, i.e. if NSCHC(IS,1) < 0.

 $\underline{\mathrm{CARD}\ 8.4}$ : NISCHC(N, IS, K); K = 1:8

8.4	NISCHC(N, IS, K)	[—]	[—]
Negative of channel number connected to the inside of (N,IS,K) azimuthal section K of rod N			
Integer	Not Operational - Require	d	

### 10.2 Unheated Conductor Data

Card 8.5 is read NSROD times for all unheated conductors (also called âĂIJheat slabsâĂİ).

CARD 8.5: N, ISTYPE(N), HPERIM(N), HPERIMI(N), RMULS(N), NOSLCH, NSLCHC, HTAMBS(N), TAMBS(N)

8.5	N	[—]	[—]
Index of unheated conduc			
	r indices must be entered s	equentially, from 1 to NSRO	DD. Skipping numbers is $\underline{\text{not}}$
permitted.			
Integer	Required		
8.5	ISTYP(N)	[—]	[—]
Geometry type identification (Refer to CARD GROUP)	ion number 9 for geometry type input	data.)	
Integer	Required		
	1		
8.5	HPERIM(N)	[—]	[—]
(Perimeter which is applie conductor N)	ide surface of unheated cond to calculate heat transfer reconnected to outside surface.	from the adjacent subchan	anel NOSLCH to the unheated
Float	Required		
	. 1		
8.5	HPERIMI(N)	[—]	[—]
(Perimeter which is applie conductor N)	de surface of unheated cond d to calculate heat transfer re connected to <i>inside</i> surfa	from the adjacent subchan	anel NSLCHC to the unheated
Float	Required		
8.5	RMULS(N)	[—]	[—]
conductor N) This number can contain	plication factor (number of	f unheated conductors mod	eled by the single unheated
Float	Required		
8.5	NOSLCH	[_]	[_]
	to outside surface of unhea	ytad aandusta: N	
, and the second	connected to <i>outside</i> surfa		
Integer	Required		
8.5	NSLCHC	[—]	[—]
Channel number adjacent	to inside surface of unheat	sed conductor N	
v	connected to inside surface		
Integer	Required		
J	*		

8.5	HTAMBS(N)	$\left[\frac{W}{m^2K}\right]$	$\left[ rac{BTU}{hrft^2 \circ F}  ight]$		
Heat transfer coefficient for heat loss to the ambience					
	This is used in some simulations of experiments when the user wants to account for the heat loss from the				
	,		channel (not water-cooled)		
to the ambience (outside of	of the bundle, inside a water	r channel, inside a guide tu	ıbe).		
Float	Required				
8.5	TAMBS(N)	$[^{\circ}\mathrm{C}]$	[°F]		
Sink temperature for heat loss to the ambience					
Float	Required				

# 10.3 Rod Temperature Initialization Tables

Cards 8.6 through 8.9 are read to specify which temperature tables apply to which rods and unheated conductors. The sequence is repeated NRTAB times, and all rods and conductors must be accounted for.

CARD 8.6: I, NRT1, NST1, NRAX1

8.6	I	[—]	[—]		
Identification number of t	emperature table				
Integer	Required				
8.6	NRT1	[—]	[—]		
Number of rods using tab	le I				
Integer	Required				
8.6	NST1	[—]	[—]		
Number of unheated cond	luctors using table I				
Integer	Required				
	·				
8.6	NRAX1	[—]	[—]		
Number of pairs of elements in table I					
Integer	Required				

Card 8.7 is read only if NRT1 > 0

 $\underline{\text{CARD 8.7}}$ : IRTAB(I, L); L = 1:NRT1

8.7	IRTAB(I, L)	[—]	[—]	
Identification numbers of rods using table I for temperature initialization				
Enter the negative of the re	od identification number if t	the temperature boundary i	s to be applied to the inside	
surface of the rod.				
Note: The steady-state conduction equation is solved for these rods using the temperatures from table I as a				
boundary condition on the rod surface.				
Integer	Conditional - if $NRT1 > 0$			

Card 8.8 is read only if NST1 > 0

 $\underline{\text{CARD 8.8}}$ : ISTAB(I, L); L = 1:NST1

8.8	ISTAB(I, L)	[—]	[—]	
Identification numbers of unheated conductors using table I for temperature initialization				
Note: A flat radial temperature profile is assumed initially in unheated conductors.				
Integer	Conditional - if $NST1 > 0$			

 $\underline{\text{CARD 8.9}}$ : AXIALT(I, L), TRINIT(I, L); L = 1:NRAX1

8.9	AXIALT(I, L)	[m]	[in]			
Vertical location	Vertical location					
Float	Required	Required				
8.9	TRINIT(I, L)	[°C]	[°F]			
Temperature to be applied at AXIALT(I, L)						
Note: The vertic	Note: The vertical locations of the bottom and top of each rod or unheated conductor using table I must be					

Note: The vertical locations of the bottom and top of each rod or unheated conductor using table I must be contained within the range AXIALT(I, 1) to AXIALT(I, NRAX1).

Float Required

## 10.4 Radiation Initialization Tables

Card 8.10 through Card 8.11.5 are read in to specify orientation and which location type tables apply to which fluid channels, rods, and unheated conductors.

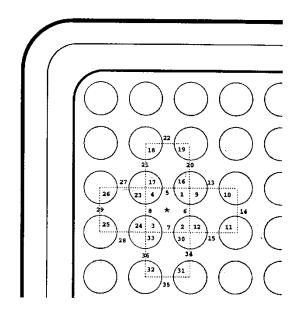
Channel Orientation and Location Type Card: 8.10

Card 8.10 is read NRAD times, if NRAD > 0

CARD 8.10: IRAD, NSIDR(IRAD), LOCATE(IRAD), NRRAD(IRAD), NSYMF(IRAD), MLTF(1, IRAD), MLTF(2, IRAD), MLTF(3, IRAD), MLTF(4, IRAD), VDMLT(IRAD)

8.10	IRAD	[—]	[—]		
Index of radiation subchar	Index of radiation subchannel				
Note: Radiation subchann permitted.	Note: Radiation subchannel indices must be entered sequentially, from 1 to NRAD. Skipping numbers is not permitted.				
Integer	Conditional - if $NRAD > 0$				
8.10	NSIDR(IRAD)	[—]	[—]		
Index of fluid subchannel	which contains radiation su	bchannel IRAD			
Integer	Conditional - if $NRAD > 0$				
8.10	LOCATE(IRAD)	[—]	[—]		
Location type for radiation	Location type for radiation channel IRAD:				
< 0 - Contains no unheated conductors					
> 0 - Has both rods and unheated conductors					
Integer	Conditional - if NRAD $> 0$				

8.10	NRRAD(IRAD)	[—]	[—]		
Number of contributing ra	diation surfaces for:				
20 - Location types 1 and	l 10				
19 - Location type 2					
18 - Location type 3					
14 - Location type 12					
13 - Location type 4					
12 - Location types 5, 13	, 14, 15, 16, and 17				
8 - Location type 6					
Integer	Conditional - if $NRAD > 0$				
8.10	NSYMF	[—]	[—]		
Flag for fluid channel or re	od lumping:				
0 No lumping					
0 - No lumping					
1 - Lumped fluid channel	$\mathbf{S}$				
Integer	Conditional - if NRAD $> 0$				
8.10	MLTF(1, IRAD)	[—]	[—]		
Surface lumping factor for	surface position 1. Ratio o	of total calculated to actual	ly modelled surface areas of		
	location type IRAD times the	ne ratio of total surface area	as in all channels of this rod		
type to this surface area:					
1.0 - Default					
Float	Conditional - if NRAD $> 0$				
8.10	MLTF(2, IRAD)	[—]	[—]		
			ly modelled surface areas of		
v =	location type IRAD times the	ne ratio of total surface area	as in all channels of this rod		
type to this surface area:	type to this surface area:				
1.0 - Default					
Float	Conditional - if $NRAD > 0$				



25 | 22 | 19 | 10 | 25 | 26 | 17 | 16 | 13 | 10 | 27 | 31 | 28 | 15 | 11 | 11 | 28 | 32 | 30 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 330 | 3

Figure 10.1: Radiation Geometry Type 1

Figure 10.2: Radiation Geometry Type 2

8.10	MLTF(3, IRAD)	[—]	[—]		
Surface lumping factor for surface position 3. Ratio of total calculated to actually modelled surface areas of this rod type contained in location type IRAD times the ratio of total surface areas in all channels of this rod type to this surface area:  1.0 - Default					
Float	Conditional - if NRAD $> 0$				
8.10	MLTF(4, IRAD)	[]	[]		
Float	Conditional - if NRAD $> 0$				
8.10	VDMLT(IRAD)	[—]	[—]		
Vapor/droplet multiplication factor. Total number of radiation channels being modeled by this location type: $1.0 - \mathrm{Default}$					
Float	Conditional - if NRAD $> 0$				

# 10.5 Radiation Channel Orientation Array

Repeat Card 8.10.1 and Card 8.10.2 until all radiation channels have been entered.

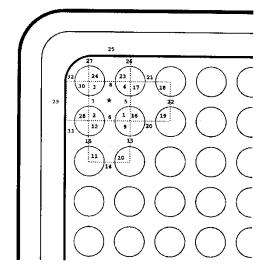


Figure 10.3: Radiation Geometry Type 3

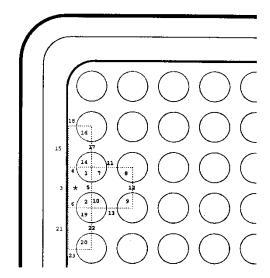


Figure 10.4: Radiation Geometry Type 4

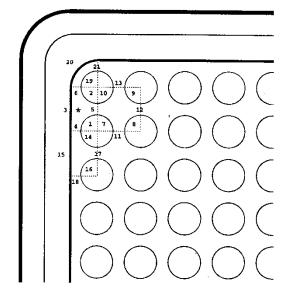


Figure 10.5: Radiation Geometry Type 5

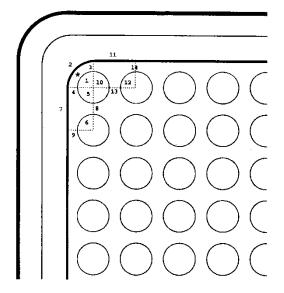


Figure 10.6: Radiation Geometry Type 6

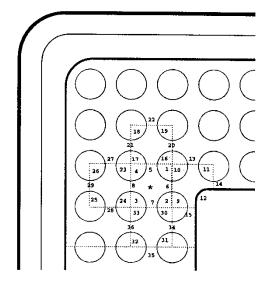


Figure 10.7: Radiation Geometry Type 10

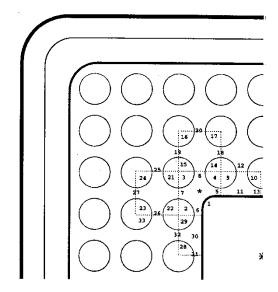


Figure 10.8: Radiation Geometry Type 11

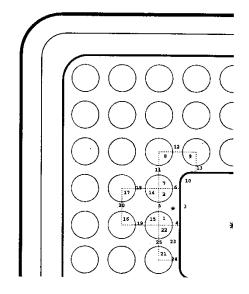


Figure 10.9: Radiation Geometry Type 12

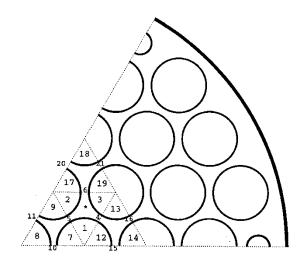


Figure 10.10: Radiation Geometry Type 13

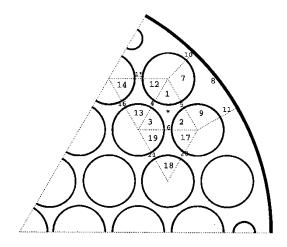


Figure 10.11: Radiation Geometry Type 14

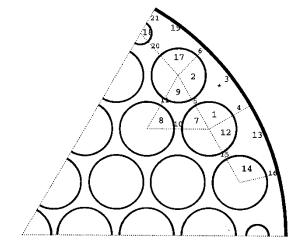


Figure 10.12: Radiation Geometry Type 15

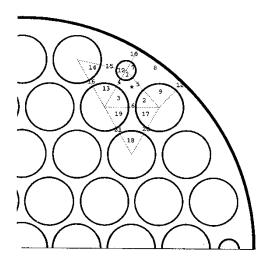


Figure 10.13: Radiation Geometry Type 16

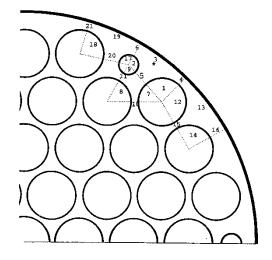


Figure 10.14: Radiation Geometry Type 17

 $\underline{\text{CARD } 8.10.1}$ : LRAD(IRAD, J); J = 1:NRRAD(IRAD)

8.10.1	LRAD(IRAD, J)	[—]	[—]		
-	in position J for appropriate radiation subchannel IRAD corresponding to location type				
LOCATE(IRAD). Negative for an inside surface.					
Integer	Conditional - if $NRAD > 0$				

# 10.6 Cavity to Cavity Radiation Weighting Factor

 $\underline{\text{CARD } 8.10.2}$ : CAVSM(IRAD, J); J = 1:4

Four records per card are required.

8.10.2	CAVSM(IRAD, J)	[—]	[—]		
Weighting factor to be applied to cavity-to-cavity radiation:					
1.0 - Suggested value					
Integer	Conditional - if NRAD $> 0$				

# 10.7 Radiation Location Type Information

 $\underline{\text{CARD 8.11}}$ : IDTYP(I); I = 1:NLTYP

8.11	IDTYP(I)	[—]	[—]			
Location type to be input	Location type to be input:					
> 0 - Manual input to fo	> 0 - Manual input to follow					
< 0 - Auto view factor routine to be used						
Integer	Conditional - if $NLTYP \neq 0$					

#### Manual Location Type Input

If  $\mathtt{IDTYP}(\mathtt{I}) < 0$ , skip CARD 8.11.1 through CARD 8.11.4

Area Input

 $\underline{\mathrm{CARD}\ 8.11.1}$ : ARAD(J); J = 1:JTOT

JTOT is the total number of surfaces for location type IDTYP(I).

8.11.1	ARAD(J)	$[{ m cm}^2]$	$[in^2]$			
Surface area of position J for location type IDTYPE(I)						
Float Conditional - if IDTYP > 0						

Emissivity Input

 $\underline{\mathrm{CARD}\ 8.11.2}$ :  $\underline{\mathrm{ERAD}}(\mathtt{J})$ ;  $\mathtt{J}$  = 1:JTOT

JTOT is the total number of surfaces for location type IDTYP(I).

	8.11.2	ERAD(J)	[—]	[—]		
	Emissivity of position J for location type IDTYPE(I)					
Float Conditional - if $IDTYP > 0$						

View Factor Input

 $\underline{\text{CARD } 8.11.3}$ : FRAD(J, K); J = 1:JL, K = J:JL

JL is the total number of radiant surfaces in location type <code>IDTYP(I)</code>.

8.11.3	FRAD(J, K)	[—]	[—]			
Radiation view factor between surface $J$ and surface $K$ where $J < K$						
Continue until all J surfaces have been entered, starting each J surface group with a new card set. Eight						
records are required per card set.						
Float	Conditional - if IDTYP > 0					

Beam Length Input

 $\underline{\text{CARD } 8.11.4}$ : DRAD(J, K); J = 1:JL, K = J:JL

JL is the total number of radiant surfaces in location type IDTYP(I).

8.11.4	DRAD(J, K)	[—]	[—]	
Beam length between surface $J$ and surface $K$ where $J < K$				
Continue until all J surfaces have been entered, starting each J surface group with a new card set. Eight				
records are required per card set.				
Float	Conditional - if IDTYP > 0	)		

Repeat CARD 8.11 through CARD 8.11.4 until all IDTYP(I) values are entered for IDTYP(I) > 0

Auto View Factor Input

Omit if IDTYP(I) > 0.

 $\underline{\mathrm{CARD}\ 8.11.5}$ : APAR(III); III = 1:10

8.11.5	APAR(III)	[—]		[—]
$\Pi^{th}$ parameter for auto	view factor input according	g to location type.	See Table	2. Continue for all ten
parameters.				
Float	Conditional - if $IDTYP < 0$	)		

Table 2: Summary of View Factor Inputs

Radiation Channel Type	Apar1	Apar2	Apar3	Apar4	Apar5	Apar6	Apar7	Apar8	Apar9	Apar10
Type 1	RD	RE	LRD	RP	-	-	-	-	-	-
Type 2	RD	RE	RWG	RP	0	0	WE	LRD	-	-
Type 3	RD	RE	RWG	RP	0	0	WE	LRD	-	-
Type 4	RD	RE	RWG	RP	0	0	WE	-	-	-
Type 5	RD	RE	RWG	RP	0	0	WE	-	-	-
Type 6	RD	RE	RWG	RP	0	0	WE	-	-	-
Type 10	RD	RE	WCR	RP	HWW	DWB	WCE	-	-	-
Type 11	RD	RE	WCR	RP	HWW	DWB	WCE	-	-	-
Type 12	RD	RE	WCR	RP	HWW	DWB	WCE	-	-	-
Type 13	R1D	R1E	R2D	R2E	R3D	R3E	P1:2	P1:3	P2:3	-
Type 14	R1D	R1E	R2D	R2E	R3D	R3E	P1:3	P1:3	P2:3	-
Type 15	R1D	R1E	R2D	R2E	R3D	R3E	D1T	D2T	G1W	G2W

#### Key for Table 2:

RD - Rod Diameter

RE - Rod Emissivity

LRD - Large Rod Diameter

RP - Rod Pitch

RWG - Rod-Wall Gap

WE - Wall Emissivity

WCR - Water Channel Corner Radius

HWW - Half Width Water Channel

DWB - Distance from Water Channel Centerline to Radius of the Channel Boundary

WCE - Water Channel Emissivity

R1D - Rod 1 Diameter

R2D - Rod 2 Diameter

R3D - Rod 3 Diameter

R1E - Rod 1 Emissivity

R2E - Rod 2 Emissivity

R3E - Rod 3 Emissivity

P1:2 - Pitch 1-2

P1:3 - Pitch 1-3

P2:3 - Pitch 2-3

D1T - Distance from Tube Centerline to Rod 1 Centerline

D2T - Distance from Tube Centerline to Rod 2 Centerline

G1W - Gap Length from Rod 1 to Tube Wall

 $\mathrm{G2W}$  -  $\mathrm{Gap}$  Length from Rod 2 to Tube Wall

Card 8.12 is read only if NCAN > 0.

### CARD 8.12: QNINIT, TLIDEN, EWET

8.12	QNINIT	[s]	[s]				
Time at which canister qu	ench is initiated						
Float	Conditional - if $NCAN > 0$						
8.12	TLIDEN	TLIDEN [°C] [°F]					
Temperature at Leidenfro	st point, at which: $\frac{dq}{d(T_{wall}-1)}$	$\frac{1}{T_{sat}} = 0$					
Float	Conditional - if $NCAN > 0$						
8.12	EWET [—] [—]						
Emissivity of the canister							
Float	Conditional - if $NCAN > 0$	Conditional - if $NCAN > 0$					

Card 8.13 is read only for parallel runs - if run\_parallel == True

### CARD 8.13: N, ROD\_GLOBAL(N), OWNER, DOMAINS\_W\_ROD\_GHOSTS(N,L); L=1:3

8.13	N	[—]				
Rod ID	Rod ID					
Integer	Conditional - if run_para	llel == 1				
8.13	ROD_GLOBAL(N)	[—]	[—]			
The global index of the ro	od in the model.					
Integer	Conditional - if run_para	llel == 1				
8.13	OWNER					
The solution domain that owns this rod.						
Integer	Conditional - if run_para	llel == 1				

Card 8.14 is read only if IKFUEL  $\geq 0$ . Cards 8.14.1 and 8.14.2 are read in pairs for each rod N.

### <u>CARD 8.14.1</u>: N

8.14.1	N	[—]	[—]		
Index of rod					
Integer	Conditional - if $IKFUEL \ge 0$				

### CARD 8.14.2: EXPOSURE(N,J), GADCON(N,J), IMOX(N,J); J=1:NONODE

8.14.2	EXPOSURE(N, J)	$[\mathrm{GWd/t}]$	$[\mathrm{GWd/t}]$	
Burnup of fuel rod N at axial node J				
Float	Conditional - if $IKFUEL \ge 0$			

8.14.2	GADCON(N,J)	[—]	[—]
Fractional amount of Gd per UO <sub>2</sub> or Pu content of MOX fuel rod N at axial node J			
Float	Conditional - if $IKFUEL \ge 0$		
8.14.2	IMOX(N)	[—]	[—]
The burnup-dependent fuel thermal conductivity option for rod N at axial node J:			
1 Modified NFI correlation ( $UO_2$ fuel)			
2 Halden correlation (UO $_2$ fuel)			
3 Duriez/Modified NFI correlation (MOX fuel)			
4 Halden correlation (MOX fuel)			
5 Amaya correlation (MOX fuel)			
Integer	Conditional - if IKFUEL $\geq$	0	

CHAPTER 11	
I	
	CARD GROUP 9

This card group is read by subroutine READ\_CARD\_9.

The geometry types are read in this group. The geometry types are numbered sequentially in the order in which they are read. Nuclear rod geometry types are read using cards Card 9.2 through Card 9.5. All other geometry types are read using cards Card 9.6 and Card 9.7.

The first line indicates the group number:  $\mathtt{NGROUP} = 9$ 

 $\underline{\mathrm{CARD}\ 9.1}$ : NFUELT, IRELF, ICONF, IMWR, IFSWELL, IFDENS, IFRELOC, IFCLADCREEP, NDUM10, NDUM11, NDUM12, NDUM13, NDUM14

9.1	NFUELT	[—]	[—]	
Number of geometry types to be read				
Note: A geometry type m	Note: A geometry type may be used by both rods and unheated conductors, but for the unheated conductor			
any heat generation specificed for that type will be ignored.				
Integer	Required			
	1		·	
9.1	IRELF	[—]	[—]	
Flag for Fuel relocation induced thermal conductivity degradation:				
1 - On				
0 - Off				
This flag is used for the thermal conductivity degradation due to the fuel relocation.				
Integer	Required			

9.1	ICONF	[—]	[—]	
Fuel degradation flag:				
1 - On				
0 - Off				
Note: If IRELF = 1, then	ICONF = 1			
Integer	Required			
9.1	IMWR			
Flag for metal-water react	ion (zirconium dioxide only	7):		
0 - Off				
1 - Cathcart (for best-est	imate analysis)			
2 - Baker-Just (for evalua	ation model analysis)			
T., 4	D			
Integer	Required			
9.1	IFSWELL	[—]	[—]	
Flag for fuel swelling:		L J	l J	
0 - Off				
1 - On, Matpro model for solid and Mod frapcon model for gaseous fission product swelling				
2 - On, Matpro model for both solid and gaseous fission product swelling				
This flag is used to choose the models to account for fuel swelling induced by both solid and gaseous fission products, and the total amount of fuel swelling is used to calculate the corresponding gap conductance.				
Integer	Required	<del>-</del>		
-				
9.1	IFDENS	[—]	[—]	
Fuel densification flag:				
0 - Off				
1 - On, Matpro model is used				
2 - On, Frapcon model is used				
3 - On, Escore model is used				
This flag is used to choose the models to account for fuel densification, and which is used to calculate the corresponding gap conductance.				
Integer	Required			
	i			

9.1	IFRELOC	[—]	[—]		
Flag for Fuel relocation induced gap conductance change:					
0 - Off					
1 - On, Frapcon model is	used				
2 - On, Escore model is u	ısed				
3 - On, Modified Escore	model is used				
This flag is used to choose sponding gap conductance		fuel relocation, which is u	used to calculate the corre-		
Integer	Required				
9.1	IFCLADCREEP	[—]	[—]		
Clad creep model:					
0 - Off					
1 - On, the Bison model	is used				
2 - On, the Escore model	is used				
This flag is used to choose gap conductance.	the model to account for c	lad creep, which is used to	calculate the corresponding		
Integer	Required				
9.1	NDUM9: NDUM14		[—]		
Not used, but entry is obligatory:					
0 - Suggested value					
Integer	Required				

# 11.1 Nuclear Fuel Geometry Types

Cards <u>CARD 9.2</u> through <u>CARD 9.5</u> are read only for nuclear fuel geometry types. If FTYPE(I)  $\neq$  NUCL, the geometry data are interpreted by cards <u>CARD 9.6</u> and <u>CARD 9.7</u>.

 $\underline{\mathrm{CARD}\ 9.2}$ : I, FTYPE(I), DROD, DFUEL(I), NFUEL, IMATF, IMATC, IMATOX(I), DCORE, TCLAD, FTDENS(I), IGPC, IGFORC, IRADP, EPSO

9.2	I	[—]	[—]	
Geometry type identification number				
Note: Geometry type index	x numbers must be entered a	sequentially from 1 to NFUE	LT. Skipping numbers is not	
permitted.				
Integer	Conditional - if FTYPE(I)	= nucl		

9.2	FTYPE(I)	[—]	[—]			
Alphanumeric geometry type flag:						
nucl - nuclear fuel rod						
Character	Conditional - if FTYPE(I)	= nucl				
9.2	DROD	[m]	[in]			
Outside rod diameter	,					
Float	Conditional - if FTYPE(I)	= nucl				
9.2	DFUEL(I)	[m]	[in]			
Fuel pellet diameter						
Float	Conditional - if FTYPE(I)	= nucl				
	T	I				
9.2	NFUEL					
Number of radial nodes in	_					
Integer	Conditional - if FTYPE(I)	= nucl				
0.0		r 1	r 1			
9.2	IMATF	[-]				
Fuel material properties fl	ag:					
$0$ - built-in $UO_2$ properti	es					
>0 - number of the mate	rial properties table (see C	ard 10)				
Integer	Conditional - if FTYPE(I)	= nucl				
0						
9.2	IMATC	[—]	[-]			
Clad material properties f	lag:					
0 - built-in zirconium dio	vide properties					
>0 - number of the mate	rial properties table (see C	ard 10)				
Intomon	Conditional if ETVDE(I)	l				
Integer	Conditional - if FTYPE(I)	= nuci				
9.2	IMATOX	[—]	[—]			
Clad oxide properties flag:						
0 - built-in zirconium dio	vido proportica					
>0 - number of the mate	rial properties table (see C	ard 10)				
Integer	Conditional - if FTYPE(I)	= nucl				

9.2	DCORE	[m]	[in]
Diameter of central void	(for cored fuel)		
0 - if uncored fuel (solid	pellet)		
Float	Conditional - if FTYPE(I)	= nucl	
9.2	TCLAD	[m]	[in]
Clad thickness			
Float	Conditional - if FTYPE(I)	= nucl	
9.2	FTDENS(I)	[—]	[—]
Fuel theoretical density a Used only if built-in $UO_2$ Note: Do <u>not</u> enter zero	s a fraction (like 0.95) properties have been flagge	d—i.e. if IMATF=0.	
Float	Conditional - if FTYPE(I)	= nucl	
9.2	IGPC	[—]	[—]
Enter a positive integer for IGPC elements) Enter a negative integer	ductance (as specified by HG or user-specified non-uniform for the dynamic gap conduction table read on CARD 9.4	n gap conductance (entered	
Integer	Conditional - if FTYPE(I)		
9.2	IGFORC	[—]	[—]
Note: valid only if IGPC > 0 - constant gap conduct		with ${\tt IGFORC}$ table entries	
Integer	Conditional II I I I I I I I I	— Huci	
9.2	IRADP	[—]	[—]
	al power profile table for the	e fuel pellet:	l J
0 - uniform pellet radial		•	
Integer	Conditional - if FTYPE(I)	= nucl	
9.2	EPS0	[m]	[in]
The roughness of this soli	d object. Enter if PPV>=3	. Only affects results if IRI	FC was set to 3 or 4.
Float	Required if PPV>=3		·

Card 9.3 is read only if FTYPE(I)=nucl and IGPC <0

 $\underline{\mathrm{CARD}\ 9.3}$ : PGAS(I), VPLEN(I), ROUFF(I), ROUFC(I), (GSFRAC(L); L=1,6), OXIDET

9.3	PGAS(I)	[bar]	[psia]
	re for nuclear fuel rod geome		
Float	Conditional - if FTYPE	E(I) = nucl && IGPC < 0	
9.3	VPLEN(I)	$[\mathrm{m}^3]$	[in <sup>3</sup> ]
Gas plenum volume	ALTEN(I)	[111 ]	[III ]
Float	Conditional - if FTYPE	$E(I) = \text{nucl } \&\& \ \text{IGPC} < 0$	
9.3	ROUFF(I)	[m]	[in]
Fuel pellet surface re	oughness		
Float	Conditional - if FTYPE	E(I) = nucl && IGPC < 0	
9.3	ROUFC(I)	[m]	[in]
correlation is empiricated surface ROUFF=	ical:	correspond to those suggested	
clad surface ROUFC=			
clad surface ROUFC-Float	= 0.000045 in	E(I) = nucl && IGPC < 0	
	= 0.000045 in	E(I) = nucl && IGPC < 0	
Float	Conditional - if FTYPE  GSFRAC(L); L=1:6	E(I) = nucl && IGPC < 0	[—]
Float 9.3	Conditional - if FTYPE  GSFRAC(L); L=1:6	E(I) = nucl && IGPC < 0	[—]
Float  9.3  Molar fraction of fill	Conditional - if FTYPE  GSFRAC(L); L=1:6	E(I) = nucl && IGPC < 0  [—]	[—]
Float  9.3  Molar fraction of fill  L=1: Helium	Conditional - if FTYPE  GSFRAC(L); L=1:6	E(I) = nucl && IGPC < 0	[—]
Float  9.3  Molar fraction of fill  L=1: Helium  L=2: Xenon	Conditional - if FTYPE  GSFRAC(L); L=1:6	E(I) = nucl && IGPC < 0	[—]
Float  9.3  Molar fraction of fill  L=1: Helium  L=2: Xenon  L=3: Argon	Conditional - if FTYPE  GSFRAC(L); L=1:6	E(I) = nucl && IGPC < 0	
Float  9.3  Molar fraction of fill  L=1: Helium  L=2: Xenon  L=3: Argon  L=4: Krypton	Conditional - if FTYPE  GSFRAC(L); L=1:6	E(I) = nucl && IGPC < 0	
Float  9.3  Molar fraction of fill  L=1: Helium  L=2: Xenon  L=3: Argon  L=4: Krypton  L=5: Hydrogen	Conditional - if FTYPE  GSFRAC(L); L=1:6  gas present, where:	E(I) = nucl && IGPC < 0  [—]  FRAC(L) = 1.000	
9.3  Molar fraction of fill L=1: Helium L=2: Xenon L=3: Argon L=4: Krypton L=5: Hydrogen L=6: Nitrogen	= 0.000045 in   Conditional - if FTYPE  GSFRAC(L); L=1:6  gas present, where: $\sum_{L=1}^{6} GS$		
Float  9.3  Molar fraction of fill  L=1: Helium  L=2: Xenon  L=3: Argon  L=4: Krypton  L=5: Hydrogen  L=6: Nitrogen  Note:		FRAC(L) = 1.000 E(I) = nucl && IGPC < 0	[m]
9.3  Molar fraction of fill L=1: Helium L=2: Xenon L=3: Argon L=4: Krypton L=5: Hydrogen L=6: Nitrogen Note:  Float		[—] FRAC(L) = 1.000	[in] d only if IMWR > 0)

Card 9.4 is read only if FTYPE(I) = nucl and |IGPC| > 0

 $\underline{\mathrm{CARD}\ 9.4}$ : AXJ(I,L), AGFACT(I,L); L=1:|IGPC|

9.4	AXJ(I,L)	[m]	[in]
Top-most vertical position	, measured from the botto	m of the rod, at which the	cold gap width or the gap
conductance AGFACT(I,L)	is applied. (All vertical le-	vels below AXJ(I,L) also g	get that value for gap width
or gap conductance).			
Float	Conditional - if FTYPE(I)	= nucl  &&   IGPC  > 0	
	I		<del></del> •
9.4	AGFACT(I,L)	$[m \ \underline{or} \ W/(m^2-K)]$	$[in \ \underline{or} \ btu/(hr-ft^2-{}^{\circ}F]$
if $IGPC < 0$ : Cold gap wi	idth		
if IGPC > 0: Gap conduc	tance		
Float	Conditional - if FTYPE(I)	$= \text{nucl } \&\& \   \text{IGPC}   > 0$	

Card 9.5 is read only if FTYPE(I) = nucl and IRADP > 0

CARD 9.5: RADP(L), POWR(L); L=1:IRADP

9.5	RADP(L)	[—]	[—]		
The relative radial location $(r/r_0)$ where the corresponding power factor POWR(L) is applied:					
$rac{r}{r_0} = rac{radius -  exttt{DCORE}/2}{( exttt{DFUEL(I)} -  exttt{DCORE})/2}$					
Float	Conditional - if FTYPE(I) = nucl && IRADP > 0				
9.5	POWR(L)	[—]	[—]		
Relative power factor (i.e., the ratio of local power at location RADP(L) to the total rod power)					
Note: RADP(L), POWR(L) pairs are entered with 4 pairs per line					
Float	Conditional - if FTYPE(I) = nucl && IRADP > 0				

# 11.2 Non-nuclear Geometry Types

Cards CARD 9.6 and CARD 9.7 are read in pairs for all geometry types that do  $\underline{not}$  describe nuclear fuel (i.e. FTYPE(I)= HROD, TUBE, WALL).

Card 9.6 is read only if FTYPE(I)  $\neq$  NUCL

 $\underline{\mathrm{CARD}\ 9.6}$ : I, FTYPE(I), DROD, DIN, NFUEL, IMATOX(I), IMATIX(I), NDUM8, NDUM9, NDUM10, NDUM11, NDUM12, NDUM13, NDUM14, EPSO

9.6	I	[—]	[—]	
Geometry type identification number				
Integer	Conditional - if FTYPE(I)	≠ nucl		

9.6	FTYPE(I)	[—]	[—]			
Alphanumeric geometry ty	type flag:					
hrod - solid cylinder						
tube - hollow tube						
wall - flat plate						
warr ride prace						
Character	Conditional - if FTYPE(I)	$\neq$ nucl				
	I	r 1	Fr. 1			
9.6	DROD	[m]	[in]			
Outside diameter for HRC Wetted perimeter for WA	DD or TUBE geometries, <u>or</u> LL geometries					
Float	Conditional - if FTYPE(I)	≠ nucl				
0.0	DIN	f1	[:1			
9.6	DIN	[m]	[in]			
Inside diameter for TUBE Thickness for WALL geom	· —					
Enter "0.0" for HROD geo						
Float	Conditional - if FTYPE(I)	≠ nucl				
	,	7				
9.6	NFUEL	[—]	[—]			
Number of radial regions v material.	within the conductor. Each	region has a uniform powe	r profile and consists of one			
Integer	Conditional - if FTYPE(I)	≠ nucl				
9.6	IMATOX(I)	[—]	[—]			
Material property table id	entification number for oxic	de on the <i>outside</i> surface:				
0 - built-in zirconium dio	xide properties (default)					
> 0 index of the metaric	al proporty table for metari	al in region NEUEL if there is	a no ovido progent (geo Card			
Group 10)	ar property table for materia	ai in region NFUEL ii there is	s no oxide present (see Card			
1						
Integer	Conditional - if FTYPE(I)	≠ nucl				
	I					
9.6	IMATIX(I)		<u>[-]</u>			
	dentification number for or	xide on the <i>inside</i> surface	(applies only to TUBE or			
WALL geometries):						
0 - built-in zirconium dio	xide properties (default)					
> 0 - index of the mater Group 10)	rial property table for mate	erial in region 1 if there is	no oxide present (see Card			
Integer	Conditional - if FTYPE(I)	≠ nucl				
<u> </u>	1	r				

9.6	NDUM8:NDUM14	[—]	[—]			
Not used, but entry is obligatory:						
0 - Suggested value	0 - Suggested value					
Note: these dummy values	are required as Card 9.6 u	tilizes the same read staten	nent as Card 9.2			
Integer	Required					
9.6	EPS0	[m]	[in]			
The roughness of this solid object. Enter if PPV>=3. Only affects results if IRFC was set to 3 or 4.						
Float	Required if PPV>=3					

Card 9.7 is read only if FTYPE(I) ≠ NUCL. Data sets for the NFUEL regions of geometry type I are entered starting at the centerline for HROD types and at the inside surface for TUBE and WALL types. Data sets are entered in sequence moving radially toward the outside surface.

CARD 9.7: NODER(L), MA	TR(L), TREG(L), QREG(L)	; L=1:NFUEL		
9.7	NODER(L)	[—]		
Number of radial heat tra	nsfer nodes in region L			
			ons hold: $2 \leq NODER(L) \leq$	
NFUEL, where NFUEL is the	number of radial regions s	pecified in Card 9.6		
Integer	Conditional - if FTYPE(I)	$\neq$ nucl		
9.7	MATR(L)	[—]	[—]	
Index of material properti				
In case of $FTYPE = TUBE$ .			$e \text{ MATR(L)} \ge 1 \text{ if it is TUBE}$	
Integer	Conditional - if FTYPE(I)	≠ nucl		
9.7	TREG(L)	[—]	[—]	
Thickness of region L				
For HROD:	NFUEL			
	TREG(L)	$O=0.5({ m DROD})$		
	L=1	, ,		
For TUBE:				
	NFUEL TREC(I) -	- 0.5(DROD_DIN)		
$\sum_{L=1} \mathtt{TREG}(\mathtt{L}) = 0.5(\mathtt{DROD-DIN})$				
For WALL:				
101 ((11111)	NFUEL	) o F (= ===)		
		0.5(DIN)		
	L=1			
Float	Conditional - if FTYPE(I)	≠ nucl		
		<u>:</u>	'	
9.7	QREG(L)	[—]	[-]	
Radial power generation f	actor for region L			
This radial power profile i	s automatically normalized	to unity		
Float	oat Conditional - if $FTYPE(I) \neq nucl$			
			·	

chapter 1	2		
CHALLER I			
			.CARD GROUP 10
This card group i	s read by subroutine READ_CARD_	10.	
9 (i.e., with non-z material properti	is required only if user-supplied mater values for IMATF, IMATC, IMATC es are used, (i.e., zircaloy and U es table has to be specified necess	OX(I),  or  MATR(L)  for  any  geomody,  this  group  is  omitted.   I	metry type). If only default
Γhe first line indi	cates the group number: NGROUP	=10	
	r, NDUM2, NDUM3, NDUM4, NDUM		DUM9, NDUM10, NDUM11, NDUM1
10.1	NMAT	[—]	[-]
Total number of	material properties tables		
Integer	Required		
10.1	NDUM2:NDUM14	[—]	[—]
	try is obligatory:	L J	
0 - Suggested va	· · · · · · · · · · · · · · · · · · ·		
Integer	Required		
	2 and CARD 10.3 are read in pai	rs NMAT times.	
		r 1	[ ]
10.2	ies table identification number		
materiai propert	Required		

	T		
10.2	NNTDP		[—]
	in materials property table N		
Must <b>not</b> exceed	100		
Integer	Required		
10.2	RCOLD(N)	$ m [kg/m^3]$	$[\mathrm{lbm/ft^3}]$
Cold density for n	naterial N		
Note: This value	is used to define the mass in the	heat transfer nodes composed	d of material type N
Float	Required		
10.2	IMATAN(N)	[—]	[—]
Alphanumeric lab	el for material—such as Stainles	s steel, Inconel 600. Enclose	all labels in quotes.
Character	Required		
CARD 10.3: TPROI	P(I,N), CPF1(I,N), THCF(I,N	); I=1,NNTDP	
10.3	TPROP(I,N)	[°C]	[°F]
Temperature for e	entry I in material properties ta	ble N	
Float	Required		
10.3	CPF1(I,N)	[kJ/(kg-°C)]	[btu/(lbm-°F)]
Specific heat capa	city at temperature TPROP(I,N)		
Float	Required		
	1		
10.3	THCF(I,N)	[W/(m-K)]	[btu/(hr-ft-°F)]
Thermal conducti	vity at temperature TPROP(I,N)	· · · · · · · · · · · · · · · · · · ·	
Float	Required		

CHAPTER	13		1
			CARD GROUP 11
This card group	o is read by subroutine READ_	CARD_11.	
The first line in	ndicates the group number: NO	#ROUP = 11	
	<u> </u>		
<u>CARD 11.1</u> : NQ NDUM14	A, NAXP, MNXN, NQ, NGPFF	, NQR, NDUM7, NDUM8, NDU	JM9, NDUM10, NDUM11, NDUM12, NI
	1204	r 1	
11.1	NQA nsient changes of the axial po		
		•	
1 - if the axia	l power profile is not transien	t	
Note: When sp	pecifying different power factor	tables (be they axial, radial,	or total power), ensure to cover the
			$ec{A} \ddot{Y} deck.out \hat{a} ec{A} \dot{Z} \ that \ {\it `tabular lookup}$
			transient power factor tables should
	ver I second. The interpolatio ing the current-time power fa	•	t 'end cap' with which to interpolate
Integer	Required	ciors.	
	1000		
11.1	NAXP	[—]	[—]
	al power profile tables to be r	ead for each transient time p	point $(\geq 1)$
Integer	Required		
11.1	MNXN	[]	[—]
	aber of pairs of elements in an	v axial power profile table	
Integer	Required	y carrier power prome tonic	
	1		
11.1	NQ	[—]	[—]
Number of pair	rs of elements in the total pov	ver forcing function table:	
0 - if total por	wer is constant		
			ı
Integer	Required		

11.1	NGPFF	[—]	[—]
Number of pairs of elemen	nts in the gap conductance	forcing function table:	
0 - if there is no forcing f	function on gap conductance	e	
Integer	Required		
11.1	NQR	[—]	[—]
Number of radial power p	rofile tables (for different tr	cansient points of time):	
1 - if radial power profile	is <u>not</u> transient		
Integer	Required		
	 	r 1	r 1
11.1	NDUM7 : NDUM14	[—]	
Not used, but entry is obl	igatory:		
0 - Suggested value			
Integer	Required		
<u>CARD 11.2</u> : YQA			
11.2	YQA	[s]	[s]
Transient time for each ax	tial power profile change	[s]	[s]
		[s]	[s]
Transient time for each ax Float	tial power profile change		
Transient time for each ax Float  Cards CARD 11.3 and CA	Required		
Transient time for each ax Float	Required		
Transient time for each ax Float  Cards CARD 11.3 and CA  CARD 11.3: I, NAXN(I)  11.3	Required  RD 11.4 are read in pairs I	JAXP times for each transien	
Transient time for each ax Float  Cards CARD 11.3 and CA  CARD 11.3: I, NAXN(I)  11.3  Axial power profile table	Required  RD 11.4 are read in pairs I	JAXP times for each transien	nt time YQA (M=1:NQA).
Transient time for each ax Float  Cards CARD 11.3 and CA  CARD 11.3: I, NAXN(I)  11.3  Axial power profile table profile table).  Integer	Required  RD 11.4 are read in pairs I  I  identification number (reference)	JAXP times for each transien	nt time YQA (M=1:NQA).
Transient time for each ax Float  Cards CARD 11.3 and CA  CARD 11.3: I, NAXN(I)  11.3  Axial power profile table profile table).  Integer  11.3	Required  RD 11.4 are read in pairs I  I  identification number (reference)  Required	IAXP times for each transier  [—]  to Card 8.2—each rod ca	nt time YQA (M=1:NQA).
Transient time for each ax Float  Cards CARD 11.3 and CA  CARD 11.3: I, NAXN(I)  11.3  Axial power profile table profile table).  Integer  11.3  Number of pairs of element	Required  RD 11.4 are read in pairs I identification number (reference)  Required  NAXN(I)  ats in axial power profile ta	IAXP times for each transier  [—]  to Card 8.2—each rod ca	nt time YQA (M=1:NQA).
Transient time for each ax Float  Cards CARD 11.3 and CA  CARD 11.3: I, NAXN(I)  11.3  Axial power profile table profile table).  Integer  11.3	Required  RD 11.4 are read in pairs I  I  identification number (reference)  Required	IAXP times for each transier  [—]  to Card 8.2—each rod ca	nt time YQA (M=1:NQA).
Transient time for each ax Float  Cards CARD 11.3 and CA  CARD 11.3: I, NAXN(I)  11.3  Axial power profile table profile table).  Integer  11.3  Number of pairs of element Integer	Required  RD 11.4 are read in pairs I identification number (reference)  Required  NAXN(I)  ats in axial power profile ta	IAXP times for each transient [—] To to Card 8.2—each rod card [—] Tole I	nt time YQA (M=1:NQA).
Transient time for each ax Float  Cards CARD 11.3 and CA  CARD 11.3: I, NAXN(I)  11.3  Axial power profile table profile table).  Integer  11.3  Number of pairs of element Integer	Required  RD 11.4 are read in pairs I  identification number (reference)  Required  NAXN(I)  ats in axial power profile tal  Required	IAXP times for each transient [—] To to Card 8.2—each rod card [—] Tole I	nt time YQA (M=1:NQA).
Transient time for each ax Float  Cards CARD 11.3 and CA  CARD 11.3: I, NAXN(I)  11.3  Axial power profile table profile table). Integer  11.3  Number of pairs of element Integer  CARD 11.4: Y(I,N), AXI  11.4	Required  RD 11.4 are read in pairs Notes in axial power profile talendary.  REQUIRED  REQUIRED  REQUIRED  NAXN(I)  Its in axial power profile talendary.  Required  ALZ(M,I,N); N=1:NAXN(I)	[—] to Card 8.2—each rod ca  [—] ble I	nt time YQA (M=1:NQA).  [—] In have its own axial power  [—]

11.4	AXIALZ(M,I,N)	[—]	[—]		
Relative axial power factor (ratio of local power to average power) at vertical location Y(I,N)					
Float Required					

# 13.1 Total Power Forcing Function

Card 11.5 is read only if NQ > 0.

 $\underline{\mathrm{CARD}\ 11.5}$ : YQ(N), FQ(N); N=1:NQ

11.5	YQ(N)	[s]	[s]		
Transient time					
Float	Float Conditional - if $NQ > 0$				
11.5	FQ(N)	[—]	[—]		
Power factor:	Power factor:				
$FQ(N) = \frac{power \text{ at time } YQ(N)}{initial power}$					
Float	Conditional - if $NQ > 0$				

# 13.2 Gap Conductance Forcing Function

Card 11.6 is read only if NGPFF > 0.

CARD 11.6: YGPFF(N), FGPFF(N), N=1:NGPFF

11.6	YGPFF(N)	[s]	[s]			
Transient time	Transient time					
Float	Conditional - if $NGPFF > 0$					
11.6	FGPFF(N)	[—]	[—]			
Conductance factor:  FGPFF(N) = conductance at time YGPFF(N) initial conductance						
Float	Conditional - if $NGPFF > 0$	)				

# 13.3 Radial Power Profile Forcing Function

Cards CARD 11.7 and CARD 11.8 are read in pairs NQR times.

<u>CARD 11.7</u>: YQR

11.7	YQR	[s]	[s]		
Transient time for each ra	Transient time for each radial power profile change				
Float	Required				

### $\underline{\mathrm{CARD}\ 11.8}$ : FQR(N), N=1:NRROD

11.8	FQR(N)	[—]	[—]		
Radial power factors for all rods (normalized to average power) starting from rod number 1. Skipping rod					
numbers is not permitted.					
Note: Eight values are en	Note: Eight values are entered per card. If NRROD is greater than 8, repeat CARD 11.8 until NRROD values				
for FQR have been entered.					
Float	Required				

CHAPTER 1	4		
			CARD GROUP 12
This card group is	read by subroutine SETIM	1.	
The first line indic	cates the group number: N	$\mathtt{GROUP} = 12$	
CARD 12.1 is read	d if $IMIX = 1$ (See Card 1.	.1)	
<u>CARD 12.1</u> : AAAK	, BETA		
12.1	AAAK	[—]	[—]
	ibution weighting factor $K$ nactive (turbulent mixing value		
Float	Conditional - if I	IMIX = 1	
12.1	BETA ase) turbulent mixing coef	[—]	[-]
Float	Conditional - if		
	m d if IMIX = 2 (See Card 1.		
12.2	AAAK	[—]	[—]
Equilibrium distri	bution weighting factor $K$	$T_m$ in void drift model:	
0.0 - Void drift $i1.4$ - Suggested v	nactive (turbulent mixing value	only)	

Conditional - if IMIX = 2

Float

12.2	DFROD	[m]	[in]		
		L J	L J		
Float	Outside rod diameter—must be consistent with DROD in CARD 9.2 or CARD 9.6!  Cloat   Conditional - if IMIX = 2				
rioat	Conditional - If $IMIX = 2$				
12.2	THETM	[ ]	T 1		
		· · · · · · · · · · · · · · · · · · ·			
	num two-phase turbulent m ngle-phase mixing coefficient				
Float	Conditional - if $\mathtt{IMIX} = 2$				
CARD 12.3: AAAK, BETA		f 1			
12.3	AAAK	[_]	[_]		
	weighting factor $K_m$ in void	drift model:	L J		
	(turbulent mixing only)				
Float	Conditional - if IMIX = 3				
12.3	ВЕТА	[—]	[—]		
Constant single-phase tu from the single-phase mi	rbulent mixing coefficient. B	eus will be used to set the	two-phase mixing coefficient		
Float	Conditional - if $IMIX = 3$				
12.3	THETM	[—]	[—]		
	num two-phase turbulent m ngle-phase mixing coefficient	•	~		

Conditional - if  $\mathtt{IMIX} = 3$ 

Float

CHAPTER 1	5		
ı			
			CARD GROUP 13
This card group is	s read by subroutine READ_C	ARD_13.	
The first line indicate	cates the group number: NGF	$\mathtt{ROUP} = 13$	
CARD 13.1: NIBN	D. NKBND. NFNUCT. NGBND.	, NIBNDB, BCVEL, NDUM7, NDU	JM8. NDUM9. NDUM10. NDUM11
NDUM12, NDUM13,		, NIBNES, ECVEE, NECKT, NEC	ne, neone, neonio, neonii
13.1	NIBND	[—]	[—]
Total number of	vertical mesh cell boundary	conditions. Includes inlet, outle	t, and internal mesh cells.
Integer	Required		
	1	r 1	
13.1	NKBND	ch cross-flow will be set to zero	
Integer	Required	cii cross-now will be set to zero	<u>'</u>
moeger	required		
13.1	NFUNCT	[—]	[—]
	forcing functions for the bou	*	·
Note: It is poss HMGA, GVALUE	ible to vary each boundary	value: BCVALUE1, BCVALUE2,	, BCVALUE3, DROPS, FDROPS,
Integer	Required		
11110801	Troquired		
13.1	NGBND	[—]	[-]
Number of group	s of adjacent transverse mon	nentum cells for which cross-flow	w will be set to zero
Integer	Required		
13.1	NIBNDB	[ ]	[]
		conditions to which the boron t	racking procipitation model is
	he number of inlet boron bo		racking/precipitation model is
Integer	Required		

13.1	BCVEL	[—]	[—]
Set to 1 if you we instead of mass fl	9	ng of BCVALUE1 (specified on	Card 13.4) to mean velocity
Integer	Required		
13.1	NDUM7:NDUM14	[—]	[-]
Not used, but ent		[ ]	1 1
0 - Suggested va	lue		
Integer	Required		
CARD 13.2: NPTS	d only if NFUNCT > 0  (K); K=1:NFUNCT		T 1
13.2	NPTS(K)		
-	of values in forcing function ta ered per card line. Repeat CA	ble K RD 13.2 until NFUNCT values ha	ave been entered.
Integer	Conditional - if NFUN	CT > 0	
13.3	d NFUNCT times IS(K,I), ORDINT(K,I); I=1 ABSCIS(K,I)	:NPTS(K); K=1:NFUNCT	[s]
Transient time			
Float	Required		
13.3	ORDINT(K,I)	[—]	[—]
Five pairs of (AB		BSCIS(K,I) er card. Repeat CARD 13.3 un ntering data until NFUNCT table	
Float	Required		

CARD 13.4, CARD 13.5, CARD 13.6, and CARD 13.7 are read NIBND times.

 $\underline{\mathrm{CARD}\ 13.4}$ : IBOUND(L,N), ISPEC(N), N1FN(N), N2FN(N), N3FN(N), BCVALUE1(N), BCVALUE2(N), BCVALUE3(N), INITGAS; L=1:2, N=1:NIBND

13.4	IBOUND(1,N)	[—]	[—]	
Index number of the channel at which boundary condition N is applied				
Integer	Required			

13.4	IBOUND(2,N)		[—]	[—]		
Vertical node number at v	which boundary condition I	V is applied.		t J		
	Note: The node number counts relative to the first node of the section in which the channel identified in					
	auxiliary (boundary) cells			,		
For example: 1 section with	th $NONODE = 80$ continuity	mesh cells:				
Inlet: $IBOUND(2,N) = 1$						
Outlet: $IBOUND(2,N) = 8$	2					
Integer	Required					
10.4	Tabba (V)	<u> </u>	r 1			
13.4	ISPEC(N)		[-]			
Boundary condition type:						
1 - pressure and enthalpy	(or temperature)					
2 - mass flow rate and en	thalpy (or temperature)					
3 - mass flow rate only						
4 - mass source (mass flo	w rate and enthalpy)					
5 - pressure sink (pressur	e and enthalpy)					
6 - mass flow rate, void,	and slip ratio (only works	for inlet)				
T ,	D . 1					
Integer	Required					
13.4	N1FN(N)		[—]	[—]		
Index number of the forcing will be varied	g function table by which th	ne first paran	neter of the bou	ndary condition (BCVALUE1)		
0 - if the boundary condi	tion is constant					
Note: The forcing function	n tables are numbered seque	entially in th	ne order they ar	re read in CARD 13.3		
	_			ill be adjusted according to		
_	able entered in CARD 13.3	•		v		
Integer	Required					
		1				
13.4	N2FN(N)					
Index number of the ford (BCVALUE2) will be varied	cing function table by wh	ich the seco	ond parameter	of the boundary condition		
0 - if the boundary condi	tion is constant					
For example: If ISPEC(N)	= 1  and N2FN(N) = 6,  th	ne specified	enthalpy will be	e adjusted according to the		
6th forcing function table		1	10			
Integer	Required					
13.4	N3FN(N)		[—]	[—]		
Index number of the for	cing function table by wl	hich the thi	rd parameter	of the boundary condition		
(BCVALUE3) will be varied						
0 - if the boundary condi	tion is constant					
Integer	Required					

13.4	BCVALUE1(N)	$[kg/s \ \underline{or} \ m/s]$	$[lbm/s \ \underline{or} \ ft/s]$	
Mass flow rate boundary condition if BCVEL=0 (Card 13.1) or velocity boundary condition if BCVEL=1.				
Only valid for certain ISPEC values. Enter if $ISPEC(N)=2$ , 3, 4, or 6. Has no effect if $ISPEC(N)=1$ or 5.				
Float	Required			

Second boundary condition value:

Enthalpy - if ISPEC(N) = 1, 2, 4, or 5  $\underline{\mathbf{or}}$ 

Temperature - enter as a negative of the number Example: for 300 °C, enter -300.0

Required

BCVALUE3(N)

0.0 - if ISPEC(N) = 3

Void - if ISPEC(N) = 6

Float

13.4

[bar or no units]

Third boundary condition value:

Pressure - if ISPEC(N) = 1, 4, or 5

0.0 - if ISPEC(N) = 2 or 3

Slip ratio - if ISPEC(N) = 6

Note: The enthalpy specified at the exit is not used by COBRA-TF if flow is in the positive direction (i.e. out of the model). In the case of positive flow, the user may enter any number for exit enthalpy

Float Required

13.4 | INITGAS | [—] [—]

Flag for the inlet boundary conditions of non-condensable gases (only relevant in the case that ISPEC(N)  $\neq$  3):

- 0 Inlet conditions will be entered in CARD 13.6 and CARD 13.7
- 1 Inlet conditions will be set equal to the initial conditions entered in CARD 1.3 and CARD 1.4 with the following assumptions:

HMGA = HGIN GVALUE = VFRAC NHMFN = O NGFN = O

Integer Required

CARD 13.5 is only read if ISPEC(N) = 4 (mass injection boundary condition)

CARD 13.5: DROPS(N), NDFN(N), FDROPS(N), NDFFN(N)

[psia or no units]

13.5	DROPS(N)	[m]	[in]	
Droplet diameter	•	,	<u>'</u>	
Float	Conditional - if ISPE	$\mathtt{CC(N)} = 4$		
13.5	NDFN(N)	[—]	[—]	
Index number of	the forcing function table by <b>v</b>	which the DROPS(N) parameter	is varied	
Integer	Conditional - if ISPE	Conditional - if ISPEC(N) = 4		
13.5	FDROPS(N)	$[\mathrm{kg/s}]$	$[\mathrm{lbm/s}]$	
Droplet mass flow	w rate at injection boundary	·		
Float	Conditional - if ISPE	Conditional - if ISPEC(N) = 4		
13.5	NDFFN(N)	[—]	[—]	
Index number of	the forcing function table by <b>v</b>	which the FDROPS(N) parameter	r is varied	
Integer	Conditional - if ISPE	$\mathtt{CC(N)} = 4$		
	1			

CARD 13.6 and CARD 13.7 are only read if ISPEC(N)  $\neq$  3 and INITGAS = 0. The structure of variables in CARD 13.6 is similar to that in CARD 1.3 and CARD 1.4, where enthalpy and void fractions are specified as initial conditions. (These initial values can be overriden by setting INITGAS = 1, see CARD 13.4)

CARD 13.6: HMGA(N), GVALUE,(NGA,N); NGA=1:NGAS+2

13.6	HMGA(N)	$[\mathrm{kJ/kg}]$	$[\mathrm{BTU/lbm}]$
Enthalpy of non-condensa	ble gas mixture		
Float	Conditional - if ISPEC(N)	eq 3  and INITGAS = 0	
13.6	GVALUE(NGA,N)	[—]	[—]
Volume fractions (inlet bo	undary condition) of:		
NGA = 1 - liquid in the li	quid-vapor-gas mixture		
NGA = 2 - vapor in the vapor-gas mixture			
$NGA \geq 3$ - gas $(NGA - 2)$ in the vapor-gas mixture			
Float	Conditional - if ISPEC(N) $\neq 3$ and INITGAS = 0		

#### CARD 13.7: NHMFN(N), NGFN(NGA,N); NGA=1:NGAS+2

13.7	NHMFN(N)	[—]	[—]		
Index number of forcing function applied to the enthalpy of non-condensable gas mixture HMGA (see CARD					
13.6)					
Integer	Conditional - if ISPEC(N)	eq 3  and INITGAS = 0			
13.7	NGFN(NGA,N)	[—]	[—]		
Index number of the forcing function applied to the volume fractions GVALUE (see CARD 13.6)					
Integer	Conditional - if ISPEC(N) $\neq 3$ and INITGAS = 0				

CARD 13.8 and CARD 13.9 are read in pairs NIBND times  $\,$ 

CARD 13.8 is only read if ISPEC(N) = 4 (mass injection boundary condition)

#### CARD 13.8: AINJT(K)

K is an integer corresponding to the total number of vertical boundary conditions cells at which boundary condition type 4 is specified. K = 1, 2, ..., total number of vertical boundary cells.

13.8	AINJT(K)	$[m^2]$	$[in^2]$		
Flow area of the mass injection					
Float	Conditional - if ISPEC(N) = 4				

CARD 13.9 is only read if ISPEC(N) = 5 (pressure sink boundary condition)

#### CARD 13.9: ASINK(K), SINKK(K), DXSINK(K)

K is an integer corresponding to the total number of vertical boundary conditions cells at which boundary condition type 5 is specified. K = 1, 2, ..., total number of vertical boundary cells.

13.9	ASINK(K)	$[m^2]$	$[in^2]$			
Flow area of the pressure	sink					
Float	Conditional - if ISPEC(N)	Conditional - if ISPEC(N) = 5				
13.9	SINKK(K)	[—]	[—]			
Pressure loss coefficient (v	relocity head) of the pressur	re sink				
Float	Conditional - if ISPEC(N) = 5					
13.9	DXSINK(K)	[m]	[in]			
Length of the momentum control volume for the sink						
Float	Conditional - if ISPEC(N) = 5					

CARD 13.10 is read NGBND times (only if NGBND > 0). This means that CARD 13.10 may be repeated as many times as necessary for a given gap K in order to identify all axial levels that have zero cross flow. The total number of transverse momentum cells with zero cross flow boundary conditions specified by CARD 13.10 must sum to NKBND.

#### CARD 13.10: K, JSTART, JEND

13.10	K	[—]	[-]		
Gap number to which a ze	ero cross flow is to be applied	ed			
Integer	Conditional - if NGBND > 0	Conditional - if $NGBND > 0$			
13.10	JSTART	[—]	[—]		
Continuity cell number at	Continuity cell number at which to start applying the zero cross flow				
$Note:\ The\ cross\ flow\ will\ be\ set\ to\ zero\ for\ gap\ {\tt K}\ between\ nodes\ {\tt JSTART}\ and\ {\tt JEND}.\ The\ node\ numbers\ are$					
given relative to the beginning of the section containing gap K.					
Integer	Conditional - if NGBND > 0	)			

13.10	JEND	[—]	[—]	
Continuity cell number at	Continuity cell number at which to stop applying the zero cross flow			
Note: The cross flow will	be set to zero for gap K be	tween nodes JSTART and J	END. The node numbers are	
given relative to the beginning of the section containing gap K.				
Integer	Conditional - if NGBND > 0	)		

CARD 13.11 is read NIBNDB times (boron inlet boundary conditions) if IBTM of CARD  $1.1 \neq 0$ 

Note: Variable NIBNDB, as the number of boron inlet B.C., does not need to match with the number of inlet B.C. entered in CARD 13.4

Note: There must be a match between where the inlet B.C. and the boron inlet B.C. are applied; (IBOUND(L,N), L=1,2) and (IBOUNDB(L,N); L=1:2)

Note: The boron tracking/precipitation model is applied when ISPEC(N) = 1, 2 or 3. The variables related with boron inlet B.C., N4FNB(N) and BCVALUE4B(N), are reassigned as extra inlet B.C. as N4FN(N) and BCVALUE4(N); N=1:NIBND

13.11: IBOUNDB(L,N), N4FNB(N), BCVALUE4B(N); L=1:2, N=1:NIBNDB

13.11	IBOUNDB(1,N)	[—]	[—]	
Index number of the channel at which boundary condition N is applied				
Integer	Conditional - if IBTM > 0			
19 11	TDUIMDD(O M)	[ ]	[ ]	

Vertical node number at which boundary condition N is applied

Note: The node number counts relatively to the first node of the section in which the channel identified in IBOUND(1,N) resides. The auxiliary (boundary) cells are counted, too

Example: 1 section with NONODE = 80 continuity mesh cells:

Inlet: IBOUND(2,N) = 1 Outlet: IBOUND(2,N) = 82

Integer Conditional - if IBTM > 0

13.11 N4FNB(N) [—] [—]

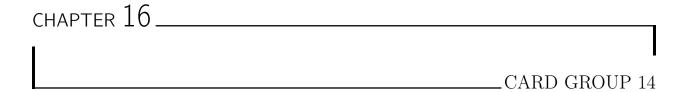
Index number of the forcing function table by which the fourth parameter of the boundary condition  $({\tt BCVALUE4})$  is varied:

0 - if the boundary condition is constant

Note: The forcing function tables are numbered sequentially in the order they are read in CARD 13.3 For example: If N4FN(N) = 5, the specified boron concentration will be adjusted according to the 5th forcing function table entered in CARD 13.3.

Integer | Conditional - if IBTM > 0

13.11	BCVALUE4B(N)	[ppm]	[ppm]	
Fourth boundary condition value - boron concentration				
Float Conditional - if IBTM > 0				



There are two options for specifying the code output options. The first option is the legacy Card Group 14. The second option is a new, more intuitive Card Group 14. The legacy Card Group 14 affords the user some options that the new one does not (such as specifying specific channels/rods to print out), but it is also more confusing to use and will not be supported going forward (e.g. one cannot turn the HDF5 file on/off from the legacy Card Group 14). These two options are described in this chapter. Only enter one of these two Card Groups 14s in the input deck.

# 16.1 Legacy Card Group 14

This card group is read by subroutine READ\_CARD\_14.

The first line indicates the group number: NGROUP = 14

 $\underline{\mathrm{CARD}}$  14.1: N1, NOUT1, NOUT2, NOUT3, NOUT4, IPROPP, IOPT, NDUM8, NDUM9, NDUM10, NDUM11, NDUM12, NDUM13, NDUM14

14.1	N1	[—]	[—]			
General output options:						
1 - print channels only						
2 - print channels and ga	ps only					
3 - print rods and unheat	ted conductors only					
4 - print rods, unheated	4 - print rods, unheated conductors, and channels only					
5 - print channels, gaps, rods, and unheated conductors						
6 - print nothing. The deck.out file will still be generated, but no channels, rods, or gaps information will be printed to the file. The results_channels.out and results_gaps.out files will not be generated. The DNB.out file will not be generated either. If IMESH = 1 is specified, the VTK file will still be printed						
7 - The same as N1 = 1, else no VTK files wi		be produced as well. Note	that IMESH must equal 1, or			
Integer	Conditional - if using Lega	acy Card Group 14				
14.1	NOUT1	[]				
Number of <i>channels</i> to be	printed (only if $N1 \neq 3$ ):					
0 - all channels will be pr	rinted					
_	channel numbers must be e	ntered on CARD 14.2				
Integer	Conditional - if using Lega	acy Card Group 14				
14.1	NOUT2	[—]	[—]			
Number of rods to be prin		l I	L J			
0 - all rods will be printe	d					
> 0 - an array of NOUT2	rod numbers must be enter-	ed on CARD 14.4				
Integer	Conditional - if using Lega	acy Card Group 14				
		r 1	[ ]			
14.1	NOUT3	[-]				
Number of gaps to be prin	,					
0 - all gaps will be printe	ed					
> 0 - an array of NOUT3	gap numbers must be enter	ed on CARD 14.3				
Integer	Conditional - if using Lega	acy Card Group 14				

14.1	NOUT4	[—]	[—]			
Number of unheated condu	uctors to be printed (only is	f N1 $>$ 2):				
0 - all unheated conductors will be printed						
> 0 - an array of NOUT4	unheated conductor number	rs must be entered on CAF	RD 14.5			
		10 111400 50 01100104 011 0111	VD 1110			
Integer	Conditional - if using Lega	acy Card Group 14				
14.1	IPROPP	[—]				
Property table print optio	n:					
0 - do not print the prop	erty table					
1 - print the property tak	ole					
Note: In the current code	version this ontion is avai	lable only for US units (10	COBRA = 2 or ICOBRA = 3);			
	printed in the output file dec	,	/,			
Integer	Conditional - if using Lega	acy Card Group 14				
			r 1			
14.1	IOPT					
Debug print option:						
0 - normal printout only						
2 - debug printout (print and krysolv.out)	extra data for subchannels	s in file results_channels.ou	nt, print extra files time.out			
3 - optional printout (prin	it averaged coolant tempera	tures in file mixture temp.	out) - may not be functional			
4 - optional printout (pri	nt subchannel void fraction	in file void.out)				
Integer	Conditional - if using Lega	acv Card Group 14				
	0 0	· ·				
14.1	ITTY	[—]	[—]			
Flag to print the deck.run	file:					
0 - print the deck.run file						
1 - do not print the deck.	run file					
Integer	Conditional - if using Lega	acy Card Group 14				
14.1	DNB	[—]	[—]			
Flag to print the dnb.out	file:					
0 - print the dnb.out file						
1 - do not print the dnb.	out file					
Integer	Conditional - if using Lega	acy Card Group 14				

14.1	IMASS	[—]	[—]
Flag to print the mass_ba	alance.out file:		t J
0 - print the mass_balan	ice.out file		
1 - do not print the mass	s_balance.out file		
Integer	Conditional - if using Lega	acy Card Group 14	
14.1	IHEAT	[—]	[—]
Flag to print the heat_ba	lance.out file:	L J	
0 - print the heat_balane	ce.out file		
1 - do not print the heat	_balance.out file		
Integer	Conditional - if using Lega	acy Card Group 14	
14.1	ICHAN	[_]	[—]
Flag to print the results_		L J	L J
0 - print the results_cha	nnels.out file		
1 - do not print the resul	ts_channels.out file		
Integer	Conditional - if using Lega	acy Card Group 14	
14.1	IGAP		
Flag to print the results		l J	l J
0 - print the results gap			
1 - do not print the resul			
Integer	Conditional - if using Lega	acy Card Group 14	
Integer			
Integer 14.1	KRY	acy Card Group 14	[—]
Integer  14.1 Flag to print the krysolv.c	KRY out file:		[—]
Integer 14.1	KRY out file: file		[—]
Integer  14.1  Flag to print the krysolv.out  0 - print the krysolv.out  1 - do not print the krysolv.out	KRY  out file:  file  olv.out file	[—]	[—]
Integer  14.1 Flag to print the krysolv.out 0 - print the krysolv.out 1 - do not print the krysol Integer	KRY  Out file:  file  olv.out file  Conditional - if using Lega	[—]	[—]
Integer  14.1 Flag to print the krysolv.out 0 - print the krysolv.out 1 - do not print the krysol Integer	KRY  Out file:  file  olv.out file  Conditional - if using Lega	[—]	[—]
Integer  14.1  Flag to print the krysolv.out  1 - do not print the krysol  Integer  CARD 14.2 is read only if	KRY  out file:  file  olv.out file  Conditional - if using Lega $\mathbb{N}1 \neq 3$ and $\mathbb{N}0\mathbb{U}$ $\mathbb{T}1 > 0$	[—]	
Integer  14.1  Flag to print the krysolv.out  0 - print the krysolv.out  1 - do not print the krysol	KRY  out file:  file  olv.out file  Conditional - if using Lega $\mathbb{N}1 \neq 3$ and $\mathbb{N}0\mathbb{U}$ $\mathbb{T}1 > 0$	[—]	
Integer  14.1 Flag to print the krysolv.or 0 - print the krysolv.out 1 - do not print the krysol Integer  CARD 14.2 is read only if CARD 14.2: PRINTC(I);	KRY  out file:  file  olv.out file  Conditional - if using Lega  N1 \neq 3 and NOUT1 > 0  I=1:NOUT1  PRINTC(I)	[—] acy Card Group 14	[—] . Repeat until NOUT1 values

CARD 14.3 is read only if N1 = 2 or 5 and NOUT3 > 0

CARD 14.3: PRINTG(I); I=1:NOUT3

14.3	PRINTG(I)	[—]	[—]
Index numbers of gaps to	be printed. Sixteen values	s are entered per card line.	Repeat until NOUT3 values
have been entered.			
Integer	Conditional - if using Lega	acy Card Group 14	

CARD 14.4 is read only if N1 > 2 and NOUT2 > 0

CARD 14.2: PRINTR(I); I=1:NOUT2

14.4	PRINTR(I)	[—]	[—]
	be printed. Sixteen values	are entered per card line.	Repeat until NOUT2 values
have been entered.			
Integer	Conditional - if using Lega	acy Card Group 14	

CARD 14.5 is read only if N1 > 2 and NOUT4 > 0

CARD 14.5: PRINTHS(I); I=1:NOUT4

14.5	PRINTHS(I)	[—]	[—]	
Index numbers of <i>unheated conductors</i> to be printed. Sixteen values are entered per card line. Repeat u NOUT2 values have been entered.				
Integer	Conditional - if using Lega	acy Card Group 14		

# 16.2 New Card Group 14

Use the new input format by entering a negative 14 instead of a positive one to start Card Group 14. Each option is simply entered as a key-value pair entered on separate lines (1 pair per line). The key-value pairs may be entered in any order. Not entering a key-value pair will result in the default behavior being used. This card must be terminated with the phrase "end 14" as the final line of this card group input.

This card group is read by subroutine READ\_CARD\_14\_ALT.

The first line indicates the group number: NGROUP = -14

Key:	hdf5	[—]	[—]		
1 - Write to the HDF5 file (only valid if $IMESH = 1$ )					
0 - No HDF5 output					
Default: 1					
Integer	Conditional - if using New	Card Group 14 && IMESH	I = 1		

Key:	rods_vtk	[—]	[—]			
1 - Write to the rods VT 0 - No rod VTK output	K file (only valid if IMESH	= 1)				
Default: 0						
Integer	Conditional - if using New	Card Group 14 && IMESH	I = 1			
TZ	1, ,,,	r 1				
Key:	chan_edits					
<ul><li>1 - Write the channels.out file</li><li>0 - No channel output</li></ul>						
Default: 0						
Integer	Conditional - if using New	Card Group 14				
	~					
Key:	rod_edits	[—]	[—]			
1 - Write the conductor of 0 - No rod/conductor ou Default: 0 Integer		Card Group 14				
	onarronar ir asing rees	Cara Group 11				
Key:	gap_edits	[—]	[—]			
<ul><li>1 - Write the gaps.out fil</li><li>0 - No gap output</li><li>Default: 0</li></ul>	le					
Integer	Conditional - if using New	Card Group 14				
Key:	fluid_vtk	[—]	[]			
1 - Write the fluid VTK 0 - No fluid VTK file our  Default: 1 Integer		1) Card Group 14 && IMESH	r — 1			
meger	TOMORDONAI - II USING NEW	Card Group 14 && IMESH	1 — 1			

Key:	dnb_edits		[—]
1 - Write the DNB da	ta to the HDF5 file		
0 - No DNB data writ	ten		
0 - NO DIVID data WIII	00011		
Default: 0			
Integer	Conditional - if using New	w Card Group 14	
Key:	edit_dnb_text_file	[—]	[—]
1 - Write the legacy d	nb.out text file		
0 - File not written			
0 - Phe not written			
Default: 0			
Integer	Conditional - if using Nev	v Card Group 14	
Key:	convergence	[—]	[—]
<ul><li>1 - Write the converge</li><li>0 - No convergence pa</li><li>Default: 1</li></ul>	ence.out file		
Integer	Conditional - if using Nev	w Card Group 14	
Key:	balance	[—]	[—]
mesh 0 - No file written	rives the true mass/energy ba	lance in the system by sum	aming fluxes over all cells in
Default: 0			
Integer	Conditional - if using Nev	v Card Group 14	
Key:	pressure	[—]	[—]
0 - No file written  Default: 0	e_summary file that gives a r		pressure drop results.
Integer	Conditional - if using Nev	w Card Group 14	

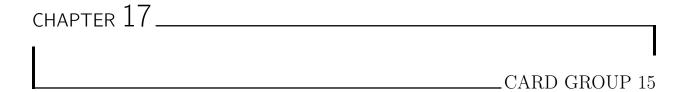
Key:	native_hdf5		[—]		[—]
arbitrary geometry	o a native HDF5 file (does n types	ot use VERA	A-CS format a	nd is therefo	ore applicable to
0 - No file written					
Default: 0					
Integer	Conditional - if using New	Card Grou	p 14		
Key:	main_text_output		[—]		[—]
1 - The main output file	from CTF (text based)				
0 - No file written	,				
Default: 1					
Integer	Conditional - if using New	Card Group	p 14		
		1		T	
Key:	mass_out		<u>[—]                                    </u>		<u>  </u>
1 - Write the mass.out f	ile (mass balance/storage)				
0 - No mass balance/sto	rage information written				
Default: 1					
Integer	Conditional - if using New	Card Grou	p 14		
		1		1	
Key:	heat_out		[—]		[—]
1 - Write the heat.out fi	le (heat balance/storage)				
0 - No heat balance/stor	rage information written				
Default: 1					
Integer	Conditional - if using New	Card Group	n 14		
8			F		
Key:	run_out		[—]		[—]
1 - Write the run.out file	e (timestep/iteration inform	ation)			
0 - No timestep/iteration	n information written				
Default: 1					
Integer	Conditional - if using New	Cord Crow	n 14		

If using the New Card Group 14, it <u>must</u> be terminated with end 14 as the last line

Example usage of the New Card Group 14:

\*NGR

-14
\*KEYS
hdf5 0
chan\_edits 1
rod\_edits 1
dnb\_edits 1
end 14



This card group is read by subroutine READ\_CARD\_15.

After all component data have been entered, the user must define the time domain for the simulation. The total time can be divided into several domains of specified duration. Each time domain can have different minimum and maximum time step sizes and different edit intervals.

If modeling a full transient (i.e.  $\mathtt{NOTRANS} = 0$ ), then any number of groups of time data may be entered, with each group specified by its own line in CARD 15.1. The last line of CARD 15.1 should give  $\mathtt{DTMIN}$  as a negative number. This will prompt CTF to stop reading. For example, if only specifying one time data group, there should be 2 lines; the first line gives the time data for the transient and the second line terminates the read of Card Group 15.

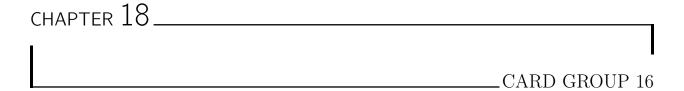
If modeling a pseudo-transient (i.e. NOTRANS = 1), then only one line of time data should be entered. The minimum and maximum timestep sizes and the RTWFP values will be used by the code, but the TEND and EDINT values will be ignored. Do not enter more than one line on CARD 15.1 if NOTRANS = 1, as this will cause a read-error by CTF.

The first line indicates the group number: NGROUP = 15

CARD 15.1: DTMIN, DTMAX, TEND, EDINT, DMPINT, RTWFP, MAXITS

15.1	DTMIN	[s]	[s]		
Minimum time step allowed for this time domain.					
Enter a negative value to terminate the calculation.					
Float	Required				
15.1	DTMAX	[s]	[s]		
Maximum time step allowed for this domain					
Float	Required				

15.1	TEND	[s]	[s]
End of this time domain			
(related to the <i>total</i> time	from the beginning at $t = 0$	) s	
Float	Required		
			·
15.1	EDINT	[s]	[s]
Print interval for this time	e domain		
	D GROUP 14 will be printe	d every EDINT seconds (rel	ated to the beginning of the
current time domain)			
Float	Required		
15.1	DMPINT	[—]	[—]
Restart dump interval			
	· ·		f the current time domain).
	n each time. One can start		
		ot used. In case $DUMPF = 1$	1 and DMPINT = $0$ , only one
restart file is written at the			
Float	Required		
15.1	RTWFP	[—]	[]
_	r heat conduction solution a		
To obtain <i>steady-state</i> consolution.	ditions, the conduction solu	tion can generally use time	e steps greater than the fluid
For the sign to a low lations	RTWFP should be 1.0		
For transient calculations	, 5110 0110 50 110		
	1 are not allowed. In this co	ase, the code automatically	$sets\ { t RTWFP}=1.0$
	<i>'</i>	ase, the code automatically	sets RTWFP = 1.0
Note: Ratios of RTWFP <	1 are not allowed. In this co	ase, the code automatically	sets RTWFP = 1.0
Note: Ratios of RTWFP <	1 are not allowed. In this co	use, the code automatically  [—]	sets RTWFP = 1.0
Note: Ratios of RTWFP < Float  15.1 Enter only if NOTRANS = 1	Required   MAXITS   (for modeling a pseudo-tra	[—] nsient). This gives the ma	sets RTWFP = 1.0  [—]  ximum number of iterations
Note: Ratios of RTWFP < Float  15.1 Enter only if NOTRANS = 1	1 are not allowed. In this concentration Required  MAXITS	[—] nsient). This gives the ma	[—]



This card group is currently not functional. It was put here to read in meshing data. Currently, meshing data are entered on Card Groups 2 and 3. It is intended that eventually the meshing information will be migrated from those card groups to this one for better organization of the input deck. Furthermore, the mesh is setup in CTF in a somewhat inefficient way right now: the cells are all free-standing, not knowing how they connect to one another. This leads to larger-than-necessary VTK files being produced. Eventually, this will be remedied and the necessary information for generating the new meshes will be put in this Card Group. Backwards compatibility will be retained for decks created using the old meshing technique using the IMESH option on Card Group 1.

CHAPTER 19	
I	
	CARD GROUP 17

This group gives a top-view of the map of channels and rods in the model. It is necessary for writing HDF5 edits. If the MAPS variable on Card 1.1 is set to 1, this Card Group must be entered. If MAPS was set to 0, do not enter this group. CTF only supports square rod lattice geometry. Special consideration must be given to which assemblies own which rods and channels. Special considerations are also required for symmetry cases. It is not intended that this card group should be generated by hand. Rather, the PWR preprocessor should be used to create the input deck when HDF5 output is desired.

This card group is read by subroutine READ\_CARD\_17.

The first line indicates the group number: NGROUP = 17

CARD 17.1: HDF5 NAME, VTK NAME

17.1	HDF5 NAME	[—]				
Desired name of the HDF5 file to be printed						
Character	Conditional - if MAPS $\neq 0$					
17.1	VTK NAME	[—]	[—]			
Desired name of the VTK file to be printed						
Character	Conditional - if MAPS $\neq 0$					
CARD 17.2: TOTRODSROW, TOTRODSCOL  17.2						
The number of rods in a single row of rods in the entire core model						
See Figure 19.1 for reference. This particular model would have 9 rods in a row.						
Integer	Conditional - if MAPS $\neq 0$					
17.2	TOTRODSCOL	[—]	[—]			
	ingle column of rods in the	core model				
Figure 19.1 would have 9 rods in a column.						
Integer	Conditional - if MAPS $\neq 0$					

CARD 17.3: TOTCHANSROW, TOTCHANSCOL

17.3	TOTCHANSROW	[—]	[—]		
The number of channels in a single row of channels in the entire core model					
The model in Illustration 1 would have 10 channels per row.					
Integer	Conditional - if MAPS $\neq 0$				
17.3	TOTCHANSCOL	[—]	[—]		
The number of channels in a single column of channels in the core model					
Integer	Conditional - if MAPS $\neq 0$				

#### CARD 17.4: Rod Map

The rod map is entered as zeros, positive integers, and negative integers. A positive integer represents the assembly index that owns a solved rod. A negative integer represents the assembly index that owns an unsolved rod, due to the rod existing on the unsolved side of a symmetry line. A "0" represents a blank region. The map must be square. When CTF reads these values in, it will assign each entity a unique ID. It will read from left to right, starting at the top row and working down. IDs are assigned in that order. The rod map for Figure 19.1 would be as shown in Table 3.

Table 3: Example Rod Map

0	0	0	1	1	1	0	0	0
0	0	0	1	1	1	0	0	0
0	0	0	1	1	1	0	0	0
2	2	2	3	3	3	4	4	4
2	2	2	3	3	3	4	4	4
2	2	2	3	3	3	4	4	4
0	0	0	5	5	5	0	0	0
0	0	0	5	5	5	0	0	0
0	0	0	5	5	5	0	0	0

If modeling this case using quarter symmetry, the assemblies that exist on the symmetry line must have both their solved and unsolved rods given in the map. The unsolved rods are shown using a negative number. A quarter symmetry map of this case would look as shown in Table 4.

Table 4: Example Quarter Rod Map

```
    -1
    -1
    -1
    -2
    -2
    -2
    -2

    -1
    -1
    -1
    -2
    -2
    -2
    -2

    -1
    1
    1
    2
    2
    2

    -3
    3
    3
    0
    0
    0

    -3
    3
    3
    0
    0
    0

    -3
    3
    0
    0
    0
```

Note that rod surface data will also be printed to the HDF5 file. CTF does not consider orientation of the

rod surfaces; only how they connect to adjacent channels. However, the PWR preprocessor has a specific way of setting up the connections that allows users to track the orientation of each rod surface. For square rod lattice geometry, there will be four quadrants of each rod (i.e. northeast, southeast, southwest, and northwest). To ensure the orientation is correctly printed to the HDF5 file, the PWR preprocessor must be utilized to generate the CTF input deck.

### CARD 17.5: Channel Map

The channel map is similarly given as positive and negative integers for solved and unsolved channels. Zeros represent blank regions in the map. Its indices are assigned in the same way as the rod map.

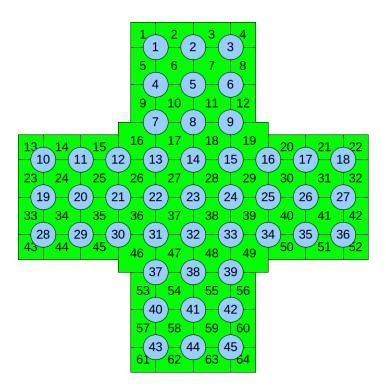


Figure 19.1: Core map of 3x3 assemblies

#### CARD 17.6 - Assembly Map Information: SYM\_OPT, NASSEM\_ROWS, NASSEM\_COLS

This card is optional, but required when modeling core symmetry. One must enter the following tag prior to the card so CTF reads the input correctly:

{assem map}

17.6	SYM_OPT	[—]	[—]							
The symmetry option being used:										
1 - full symmetry										
4 - quarter mirror symme	etry									
5 - quarter rotational syn	nmetry									
Integer	Conditional - if MAPS $\neq 0$	and {assem map} is entere	d							

17.6	NASSEM_ROWS	[—]	[—]				
Number of assemblies in each row of assemblies							
Integer Conditional - if MAPS $\neq 0$ and {assem map} is entered							
17.6	NASSEM_COLS	[—]	[—]				
Number of assemblies in each column of assemblies							
Integer	Conditional - if MAPS $\neq 0$ and {assem map} is entered						

# $\underline{\mathrm{CARD}\ 17.7}$ - Assembly Map

If CARD 17.6 is entered, then CARD 17.7 must be entered as well. This is a top view map of the core with each index representing an assembly in the core. For Figure 19.1, the assembly would look as shown in Table 5

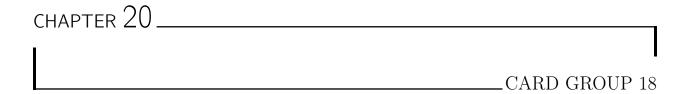
 Table 5: Example Quarter Rod Map

When modeled using quarter symmetry, this map would look as shown in Table 6:

Table 6: Example Quarter Rod Map

1 2

3 0



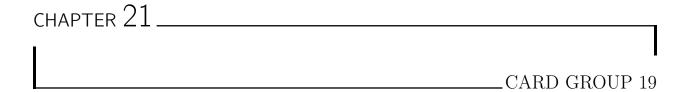
This group specifies a set of convergence criteria that will be used to judge when the transient has reached steady state. This only has an effect if the user specified NOTRANS=1 in Card 1.1. If this card is not specified, a default set of convergence criteria will be used. This card group specifies an older set of convergence criteria that use mass and energy balance and storage as the criteria. The meaning of these terms are not as intuitive as watching actual solution terms become steady. A new set of convergence criteria that are based on the  $l_{\infty}$  of several solution variables are provided in Card Group 19 and they are recommended over these criteria. Enter Card Group 18 to use the older criteria or enter Card Group 19 to use the newer criteria. Do not enter both card groups. A basic description of the terms may be found here. Consult the CTF Theory Manual [4] for a more in-depth description of these terms.

The first line indicates the group number: NGROUP = 18

CARD 18.1: Global energy balance

18.1	ENERGYBAL	[%]							
Energy in minus energy out, normalized to energy in.									
Float	Required								
18.2	MASSBAL	[%]							
Mass in minus mass out, normalized to mass in.									
Float	Required								
18.3	FESTOR	[%]	[%]						
Energy stored in the fluid	, normalized by system pow	rer							
Float	Required								
18.4	SESTOR [%] [%]								
Energy stored in the solid	, normalized by system pow	ver							
Float	Required								

18.5	MSTOR	[%]	[%]		
mass stored in the coolant	, normalized by inlet mass	flow			
Float	Required				



This group specifies a set of stopping criteria that will be used to judge when the transient has reached steady state. This only has an effect if the user specified NOTRANS=1 in Card 1.1. This set of stopping criteria is an alternative to the set that may be specified on Card 18. If neither Card 18 or 19 are specified, the code will default to using the stopping criteria of Card 18 (default values will be used). This card group specifies a newer set of stopping criteria that use the normalized  $l_{\infty}$ -norm and  $l_2$ -norm of the difference between solution vectors at set intervals of time in the simulation. The  $l_{\infty}$  and  $l_2$  is checked for:

- 1. Void
- 2. Pressure
- 3. Coolant temperature
- 4. Solid temperature
- 5. Liquid axial velocity
- 6. Vapor axial velocity
- 7. Droplet axial velocity

Each parameter (except void) is checked against both a relative and an absolute tolerance. Void is dimensionless, so it is only checked against an absolute tolerance. A steady-state run of the code will converge if the following conditions holds true.

$$\left[\frac{\max(|x^{n} - x^{n-1}|)}{x^{n-1}} <= R_{\infty}, \text{ or } \max(|x^{n} - x^{n-1}|) <= A_{\infty}\right], \text{ and}$$

$$\left[\sqrt{\frac{\sum_{k=1}^{N} \left(\frac{x^{n} - x^{n-1}}{x^{n-1}}\right)^{2}}{N}} <= R_{2}, \text{ or } \sqrt{\frac{\sum_{k=1}^{N} \left(x^{n} - x^{n-1}\right)^{2}}{N}} <= A_{2}\right]$$

In this equation, x is a vector of the solution for all mesh cells in the model (i.e., pressure, coolant temperature, etc.). The exponent, n, represents the point in time where a check has been made. Solution vectors are

compared at 0.05 s intervals in the transient solution. The first two equations are comparing the relative  $l_{\infty}$  and absolute  $l_{\infty}$  against relative, R, and absolute, A, stopping criteria. Note that A has units and R is dimensionless. The second two equations compare the relative  $l_2$  and absolute  $l_2$  against its separate relative and absolute stopping criteria. As the equations indicate, the  $l_{\infty}$  must meet either the relative or absolute criteria and the  $l_2$  must meet either its relative or absolute criteria. Both the  $l_2$  and  $l_{\infty}$  must satisfy at least one of its criteria for the simulation to be completed.

There are two caveats: first, the denominators have a floor value of  $1 \cdot 10^{-16}$  to prevent divide-by-zero errors and, second, the code will take one more iteration (one more timestep) and double-check that the solution is still below the tolerance. As long as the code still meets the stopping criteria, the code will exit.

The benefit of the relative stopping criteria is that it defines the number of digits to which each solution term is steady. For example, if CTF satisfies a relative stopping criteria of 1E-5 for solid temperatures, and a given temperature in the solution is 300.01, the user knows that those five digits will not change any further. The solution can oscillate in the sixth digit, but this small amount of wiggle in the solution may be acceptable for the user's application. The absolute tolerance is added as a safety measure to protect against situations where some solved quantity becomes very small. As seen in Equation 21.1, the relative stopping criteria is normalized by the old time solution value,  $x^n$ . If some velocity, for example, is close to zero, the relative change in the term can be huge from one checkpoint to another, even though the absolute change may be very small and the solution essentially steady.

The void difference does not check against a relative tolerance because, frequently, the void will be very small, leading to very large relative differences from one check to another; it is only checked against the absolute tolerance. In addition to Equation 21.1, CTF will also check the mass and energy balance of the solution domain. Normalized mass balance is defined in Equation 21.2 and normalized energy balance is defined in Equation 21.3.

$$M_{\text{balance}} = \frac{M_{\text{inlet}} - M_{\text{outlet}}}{M_{\text{inlet}}} \cdot 100 \tag{21.2}$$

$$E_{\text{balance}} = \frac{Q_{\text{rod}} + Q_{\text{fluid}} + Q_{\text{inlet}} - Q_{\text{outlet}} - Q_{\text{amb}}}{Q_{\text{rod}} + Q_{\text{fluid}} + Q_{\text{inlet}}} \cdot 100$$
(21.3)

The terms,  $M_{\text{inlet}}$  and  $M_{\text{outlet}}$  define the total mass entering the system and the total mass leaving the system, respectively. In the energy equation, the terms,  $Q_{\text{rod}}$ ,  $Q_{\text{fluid}}$ ,  $Q_{\text{inlet}}$ ,  $Q_{\text{outlet}}$ , and  $Q_{\text{amb}}$  represent the energy generated in the solid objects, energy generated in the fluid (gamma heating), energy entering the system due to advection, energy leaving the system due to advection, and energy loss to the environment, respectively. In a steady-state system, both of these terms should be driven to zero within some tolerance.

The following three cards specify the values of the tolerances for the steady-state check. Card 19.1 specifies the relative tolerances for  $l_{\infty}$ , Card 19.2 specifies the absolute tolerances for  $l_{\infty}$ , Card 19.3 specifies the tolerances for the mass and energy balances, Card 19.4 specifies the relative tolerances for  $l_2$ , and Card 19.5 specifies the absolute tolerances for  $l_2$ . If the user wishes to use only one type of stopping criteria (i.e., either  $l_{\infty}$  or  $l_2$ ), the other set may be disabled by setting it to an arbitrarily high value.

The first line of the card group indicates the group number: NGROUP = 19

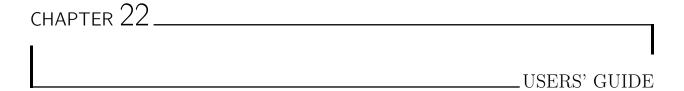
CARD 19.1: LIPRESS, LITCOOL, LITSOLID, LIVL, LIVV, LIVD

19.1	LIPRESS	[—]	
Relative tolerance for the	$l_{\infty}$ of pressure. Default 1E	-5.	
Float	Required		

19.1	LITCOOL	[—]	[—]
	$l_{\infty}$ of coolant temperature.	Default 1E-5.	L J
Float	Required		
	1 *		
19.1	LITSOLID	[—]	[—]
Relative tolerance for the	$l_{\infty}$ of solid temperature. D	efault 1E-5.	
Float	Required		
19.1	LIVL	[—]	[—]
Relative tolerance for the	$l_{\infty}$ of axial liquid velocity.	Default 1E-5.	
Float	Required		
	1		
19.1	LIVV	[—]	[—]
Relative tolerance for the	$l_{\infty}$ of axial vapor velocity.	Default 1E-5.	
Float	Required		
	T		
19.1	LIVD	[—]	[—]
Relative tolerance for the	$l_{\infty}$ of axial droplet velocity	. Default 1E-5.	
Float	Required		
CARD 19.2: LIVOID, LIA	PRESS, LIATCOOL, LIATSO	DLID, LIAVL, LIAVV, LIAV	D
19.2	LIVOID	[—]	[—]
Criteria for the $l_{\infty}$ of void	l. Default 1E-5.		
Float	Required		
	I		
19.2	LIAPRESS	[bar]	[psi]
	$l_{\infty}$ of pressure. Default 0.0	001 psi.	
Float	Required		
10.0	T. T. III 0007	[c]	[p]
19.2	LIATCOOL	[C]	[F]
	$l_{\infty}$ of coolant temperature	. Default 0.001 F.	
Float	Required		
19.2	LIATSOLID	[0]	[च]
		[C]	[F]
	$l_{\infty}$ of solid temperature. I	retault U.UUI F.	
Float	Required		
19.2	LIAVL	$[\mathrm{m/s}]$	$[\mathrm{ft/s}]$
Absolute tolerance for the	$e^{l}$ of axial liquid velocity.		. , ,
Float	Required		
	1 -		
19.2	LIAVV	[m/s]	[ft/s]
Absolute tolerance for the	$e^{l_{\infty}}$ of axial vapor velocity.	Default $0.001 \mathrm{ft  s^{-1}}$ .	
Float	Required		
	ı		

19.2	LIAVD	[m /a]	[ft /a]							
	ce for the $l_{\infty}$ of axial droplet v	m/s [m/s]	$[\mathrm{ft/s}]$							
Float	Required	elocity. Delauit 0.0011ts .								
rioat	Required									
<u>CARD 19.3</u> : ENE	RGYBAL, MASSBAL									
19.3	ENERGYBAL	[%]	[%]							
Energy in minus	energy out, normalized to ene	rgy in. Default $0.01\%$ .								
Float	Required									
		100	50.43							
19.3	MASSBAL	L J								
	nass out, normalized to mass in	n. Default 0.01 %.								
Float	Required									
CARD 19.4: L2P	RESS, L2TCOOL, L2TSOLID, I	.2VL, L2VV, L2VD								
19.4	L2PRESS	[—]	[—]							
Relative tolerand	ce for the $l_2$ of pressure. Defau	lt 1E-5.								
Float	Required									
19.4	L2TC00L	[—]	[—]							
Relative tolerance	e for the $l_2$ of coolant tempera	ture. Default 1E-5.								
Float	Required									
19.4	L2TSOLID	[—]	[—]							
Relative tolerand	e for the $l_2$ of solid temperature	re. Default 1E-5.								
Float	Required									
19.4	L2VL	[—]	[—]							
Relative tolerance	the for the $l_2$ of axial liquid velo	city. Default 1E-5.								
Float	Required									
19.4	L2VV									
	$\frac{1}{2}$ ee for the $l_2$ of axial vapor velo	city. Default 1E-5.								
Float	Required									
19.4	L2VD	[—]	[—]							
Relative tolerand	the for the $l_2$ of axial droplet vel	locity. Default 1E-5.								
Float	Required									
CARD 19.5: L2V	OID, L2APRESS, L2ATCOOL, I	.2ATSOLID, L2AVL, L2AVV, L2AV	/D							
19.5	L2V0ID		[—]							
Criteria for the $l$	2 of void. Default 1E-5.	[								
Float	Required									
	1 1 1 1 1 1									

19.5	L2APRESS	[bar]	[psi]							
Absolute tolerance for the	$l_2$ of pressure. Default 0.00	)1 psi.								
Float	Required									
19.5	L2ATCOOL	PATCOOL [C] [F]								
Absolute tolerance for the	$l_2$ of coolant temperature.	Default 0.001 F.								
Float Required										
19.5	L2ATSOLID	[C]	[F]							
Absolute tolerance for the	$l_2$ of solid temperature. De	efault 0.001 F.								
Float Required										
19.5	L2AVL	[m/s]	$[\mathrm{ft/s}]$							
Absolute tolerance for the	$l_2$ of axial liquid velocity.	Default $0.001 \mathrm{ft  s^{-1}}$ .								
Float	Required									
19.5	L2AVV	$[\mathrm{m/s}]$	$[\mathrm{ft/s}]$							
Absolute tolerance for the	$l_2$ of axial vapor velocity.	Default $0.001 \mathrm{ft  s^{-1}}$ .								
Float	Required									
19.5										
Absolute tolerance for the	$l_2$ of axial droplet velocity.	Default $0.001 \mathrm{ft  s^{-1}}$ .								
Float	Required									



# 22.1 General

This section is intended to aid the user in learning to set up the input deck for CTF. There are three basic tasks the user must perform:

- 1. setting the geometry data to describe the system;
- 2. specifying the fluid conditions and forcing functions to define the state of the system; and
- 3. running the code and interpreting the output.

The first two tasks are covered in this section.

Any vertical one-, two-, or three-dimensional component in the reactor vessel can be modeled with CTF. The exceptions are components such as pumps or pressurizers which have special boundary conditions not included in the code capabilities. In addition, the problem must be amenable to solution by the semi-implicit numerical algorithm used.

All geometries modeled are represented as a matrix of Eulerian mesh cells. The number of cells depends on the degree of detail required to resolve the flow field, the phenomena being modeled, and practical restrictions such as computing costs and storage limitations. In two-phase flow, the mesh cell should be large compared to the characteristic size of the two-phase flow pattern. For example, the mesh cell size should be large relatively to the bubble size, so that averaged quantities (bubble size, drag, etc.) used in the calculation will be valid. In slug or film flow, the mesh cell size should be on the order of the hydraulic diameter or larger since the physical models for these flow regimes are based on physical dimensions of the flow path.

The equations for flow field in the reactor core are solved using a staggered difference scheme on the Eulerian mesh. The velocities are obtained at the mesh faces and the state variables (e.g., pressure, density, enthalpy, and phasic volume fractions) are obtained at the cell center. The mesh cell is characterized by cross-sectional area, A, height,  $\Delta X$ , and the width of the connections to the adjacent mesh cells, S. The basic mesh cell is shown in Figure 22.1. This cell defines the control volume for the scalar continuity and energy equations. The momentum equations are solved on a staggered mesh with the momentum cell centered on the scalar cell face. The momentum cell for axial velocities is shown in Figure 22.2 and that for transverse velocities in Figure 22.3.

The input has been constructed to allow the user a great deal of flexibility in defining the mesh for the irregular geometries typical of reactor vessel internals. The mesh cells are defined by input in terms of subchannels. A subchannel is a vertical stack of mesh cells, as illustrated in Figure 22.4. A subchannel may represent a fluid volume between four fuel rods, a lumped region of the core, a segment of the downcomer, or any other vertical flow path appropriate to the geometry being modeled.

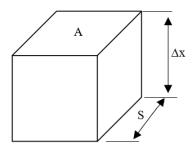


Figure 22.1: Basic mesh cell

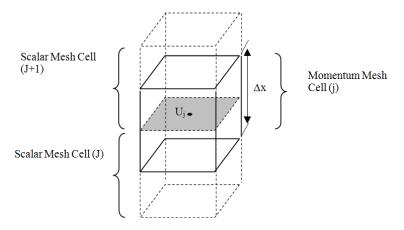


Figure 22.2: Mesh cell for axial momentum

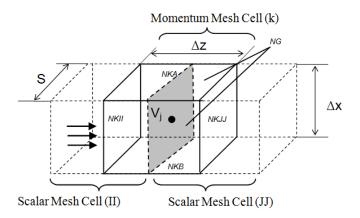


Figure 22.3: Mesh cell for transverse momentum

Boundary data for each subchannel are stored in phantom nodes at its top and bottom. Between these two

phantom nodes are NDX nodes which actually enter into calculation. Node J contains boundary conditions for the bottom of the subchannel and node J = NDX + 2 contains boundary conditions for the top of the subchannel. Boundaries between mesh cells are identified in Figure 22.4 with the lower case 'j' and refer to locations of the momentum mesh center, where velocities are obtained. The velocity corresponding to J = 1 is at the top of the continuity mesh cell corresponding to J = 1.

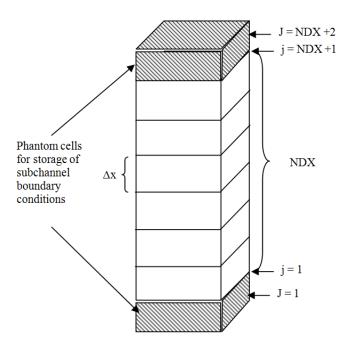


Figure 22.4: Basic Subchannel

Nominal flow area and wetted perimeter are specified for each subchannel by input. Each mesh cell within the subchannel is assumed to have nominal geometry unless variations for area or/and wetted perimeter are specified by the user. The mesh for a particular region of the vessel is developed by pacing a sufficient number of subchannels in the region to model the geometry and the important flow phenomena of the region. Transverse connections are specified between subchannels to complete the multidimensional mesh of the region. These connections are referred to as gaps. Gaps are defined by the width of the flow path between two subchannels and the distance between the subchannels centroids. The width of the gap between subchannels is assumed to be uniform along the total length of the subchannel unless an axial variation of the gap is specified. The centroid distance is always equal to the nominal value for the gap; no variation is allowed.

All regions composed of subchannels of same vertical length and beginning at the same level in the vessel are grouped together and referred to as a section. When all sections have been defined, the sections are joined together to form the omplete mesh by specifying connections to the subchannels in adjacent sections at the top and bottom of each subchannel. The process of connecting one or more subchannels to the top or bottom of a subchannel is referred to as subchannel splitting. Each subchannel and gap in the problem is assigned a unique identification number by the input and these subchannel numbers are used to identify the connections at the top and bottom of each subchannel. Connections are not specified for subchannels without a physical flow path at the inlet or outlet. Boundary conditions on the ends of subchannels that do not connect to subchannels in an adjacent section are specified by input.

Fuel rod simulators, nuclear fuel rods and the solid structures within the vessel can be modeled using the rod and unheated conductor model. Heat transfer through solid structures, thermal storage during transient, and quenching of hot dry surfaces can be calculated by the conduction model. The user has a great deal of

flexibility in modeling geometries of solid structures in the system. They can be represented as cylindrical rods, hollow tubes, or flat walls composed of any number of different materials with specified thermal properties.

To summarize, the basic building block for the vessel mesh is the subchannel, which is a vertical stack of single mesh cells. Several subchannels can be connected together by gaps to model a region of the reactor vessel. Regions that occupy the same level form a section. Sections are connected axially to complete the vessel mesh by specifying subchannel connections between sections. Heat transfer surfaces and solid structures that interact significantly with the fluid can be modeled with rods and unheated conductors.

# 22.2 Specification of the Geometry Data

This section provides the user with guidance how to model the geometry of the problem being simulated. It covers the code input requirements for modeling of the subchannels and gaps, the splitting of axial sections, and the spacer grid data.

# 22.2.1 Instructions to CARD GROUP 2

The input for CARD GROUP 2 specifies the number of subchannels in the model and their geometrical characteristics. The total number of subchannels, NCHANL, is specified on CARD 2.1. Each subchannel is uniquely identified on CARD 2.2 by an index number I, flow area AN(I), and wetted perimeter PW(I). The subchannel identification numbers are completely arbitrary and the user is not required to number them sequentially. The only constrain on subchannel numbers is that they must be unique.

The variables ABOT and ATOP on CARD 2.2 are optional input to model the area at the top and the bottom of the subchannel for the momentum equation solution. By default both are equal to the subchannel nominal flow area AN(I). When at the section boundaries there are sudden changes in the geometry of the system (for example, orifice or flow distribution plates), ABOT and ATOP are used to supply the correct area at the bottom and the top of the subchannel for the momentum solution. Nonzero values for ABOT and ATOP will replace the nominal subchannel area at those locations.

If the axial velocities for a subchannel can convect transverse momentum between sections, the location of those velocities must be specified by the user. If this input is not provided, transverse momentum will accumulate in the affected cells of subchannel I at the section boundaries, causing errors in the pressure solution. The number of gaps for which subchannel I convects transverse momentum between sections is specified as the variable NAMGAP on CARD 2.2.

The connections for axial convection of transverse momentum across the section boundary by a given subchannel I are specified on CARD 2.3. The connections are defined by the node number INODE(I) of the axial velocity convecting momentum and the index numbers of the gaps, above and below the section boundary, whose momentum is convected. The node numbering convention is shown in Figure 22.5. The center of the scalar mesh are identified by J=1,2,...(NONODE+2) and the centers of the momentum cells are identified by J=1,2,...(NONODE+1). The node number INODE(I,N) will be 1 for the subchannel velocity at the bottom of the boundary of the section, or NONODE+1 for the subchannel velocity at the top of the section. KGAPB(I,N) is the gap number on the lower side of the section boundary and KGAPA(I,N) is the gap number of the upper side of the section boundary. If the flow is positive, the axial velocity of subchannel I convects transverse momentum from KGAPB to KGAPA. If the flow reverses, momentum is convected from KGAPA to KGAPB. This input is fairly obvious for cases where there is a gap both above and below the section boundary. But even if one or other is nonexistent this input must be provided to tell the code how to dissipate transverse momentum that is transported out of a gap across a section boundary. If there is a gap below, but no gap above the section boundary, KGAPA is entered as zero and momentum convected out of KGAPB is considered

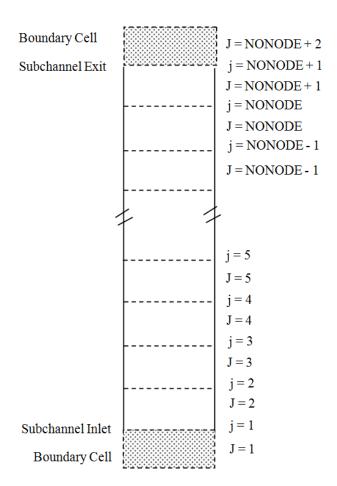


Figure 22.5: Subchannel node numbering convention

dissipated. Similarly, if there is a gap above, but no gap below, KGAPB is entered as zero. Reverse flow convects momentum out of KGAPA, which is considered dissipated. The area through which momentum is convected is calculated as the minimum of the  $S_k L_k/2$  for KGAPA and KGAPB and the flow area of subchannel I. A subchannel can convect transverse momentum to or from several gaps at the top and/or bottom of the section. For user convenience, when only one axial section is modeled, the input values of CARD 2.3 are ignored. They are created automatically by the code; which means NAMGAP(I) may be set to zero and CARD 2.3 may be omitted.

In defining the input for CARD 2.3, the user must understand how the momentum equation is solved at the section boundaries, to determine if the axial velocities will be solved in subchannel I at the section boundary. It does not do any good to specify the NAMGAP connections for subchannels that are not solved at the section boundary, and conversely, they must be specified for those that are. Regardless of the geometry being modeled, there are three basic patterns for subchannel splitting connections. These are:

- 1. one subchannel below connected to one subchannel above (see Figure 22.6 case (1));
- 2. one subchannel below connected to many subchannels above (see Figure 22.6 case (2)); and
- 3. multiple subchannels below connected to one subchannel above (see Figure 22.6 case (3)).

Transverse momentum convection input data are required only for subchannels where the momentum equation will be solved at the section boundary, and velocities so obtained can convect transverse momentum

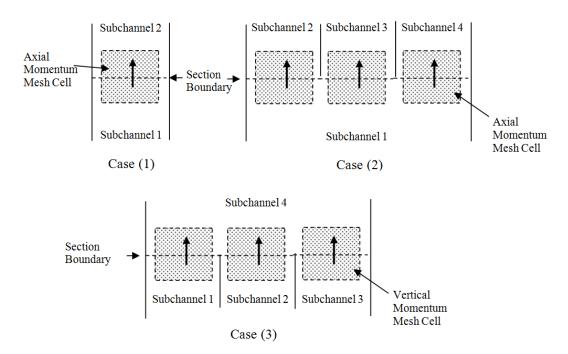


Figure 22.6: Subchannel connections at section boundaries allowed by the subchannel splitting logic

from or to adjacent gaps. The momentum equation is solved at the top of subchannel 1 for case (1) and is not solved at the bottom of subchannel 2. The momentum equation is solved at the bottom of subchannels 2, 3, and 4 in case (2), and at the top of subchannels 1, 2, and 3 in case (3).

The possible subchannel and gap configurations at section boundaries are outlined in Figure 22.7. The case where two subchannels below the section boundary connect to two subchannels above the section boundary with no change in the number of mesh cells between sections is illustrated in Figure 22.7 (a). The axial momentum equation is solved in the top nodes of subchannels 1 and 2 to obtain axial velocities at the section boundary. The velocity of subchannel 1 convects transverse momentum from the left side of the transverse momentum cell for gap 2. Similarly, the velocity of subchannel 2 convects momentum from the right side of the momentum cell for gap 1 to the right side of the momentum cell for gap 2. For this example, the input for CARD 2.3 must be supplied for both subchannels 1 and 2 as follows:

```
***********************************
* Axial section 1, subchannel 1
* Card 2.2
         AN
              PW ABOT ATOP NMGP
    Ι
    1
        xxxx xxxx 0.0 0.0
* Card 2.3
   INODE KGAPB KGAPA
NONODE+1
           1
* Axial section 1, subchannel 2
* Card 2.2
              PW ABOT ATOP NMGP
    Т
         AN
    2
        xxxx xxxx 0.0 0.0
   INODE KGAPB KGAPA
NONODE+1
          1
* Axial section 2, subchannel 3
* Card 2.2
    Ι
              PW ABOT ATOP NMGP
         AN
        xxxx xxxx 0.0 0.0
* Axial section 2, subchannel 4
* Card 2.2
   Ι
          AN
              PW ABOT ATOP NMGP
        xxxx xxxx 0.0 0.0
************************************
```

For the case shown in Figure 22.7 (b), there is no gap on the top side of the section boundary, so the axial velocities at the top of subchannels 1 and 2 convect momentum out of gap 1, but the momentum is assumed to be dissipated since there is no transverse momentum cell to receive it. For this case, the input will be:

```
* Axial section 1, subchannel 1
* Card 2.2
    Ι
          AN
               PW ABOT ATOP NMGP
    1
        xxxx xxxx 0.0 0.0
* Card 2.3
   INODE KGAPB KGAPA
NONODE+1
           1
* Axial section 1, subchannel 2
* Card 2.2
    Ι
          AN
               PW ABOT ATOP NMGP
    2
        xxxx xxxx 0.0 0.0
   INODE KGAPB KGAPA
NONODE+1
            1
* Axial section 2, subchannel 3
* Card 2.2
    Ι
               PW ABOT ATOP NMGP
          AN
        xxxx xxxx 0.0 0.0
**********************************
```

If a gap exists both above and below the section boundary as in Figure 22.7 (a), but there is a flow straightener (such as a core support plate) at the section boundary, then the transverse momentum can be dissipated by specifying two connectors for each subchannel, one for positive flow and the other for negative flow. Positive flow convects momentum from gap 1 and dissipates it; negative flow convects momentum from gap 2 and dissipates it:

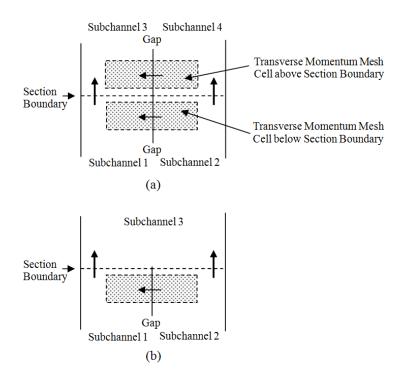


Figure 22.7: Typical configuration for convection of transverse momentum between sections

```
* Axial section 1, subchannel 1
 Card 2.2
                  PW ABOT ATOP NMGP
     Ι
          xxxx xxxx 0.0 0.0
* Card 2.3
    INODE KGAPB KGAPA
NONODE+1
              1
NONODE+1
              0
* Axial section 1, subchannel 2
 Card 2.2
     Ι
                  PW ABOT ATOP NMGP
            AN
     2
                      0.0 0.0
          xxxx xxxx
    INODE KGAPB KGAPA
NONODE+1
NONODE+1
              0
* Axial section 2, subchannel 3
* Card 2.2
     Ι
            AN
                  PW ABOT ATOP NMGP
     3
          xxxx xxxx 0.0 0.0
* Axial section 2, subchannel 4
 Card 2.2
     Ι
                  PW ABOT ATOP NMGP
            AN
          XXXX XXXX
                      0.0 0.0
```

If momentum is convected by axial velocity from one gap on one side of the section boundary to several gaps on the other side (variable mesh) then transverse momentum convection connections must be specified

for each gap. The momentum taken from the single gap is apportioned among the connecting gaps by the relative size of the areas through which the momentum is convected.

#### 22.2.2 Instructions to CARD GROUP 3

The gap input defines the control volume for the transverse momentum equation, on either a subchannel analysis basis or for a three-dimensional analysis of transverse flow.

The total number of gaps, NK, is read on CARD 3.1. The geometry data for each gap is read on CARD 3.2. Each gap is identified by a unique number K. Gap K connects subchannel IK(K) to subchannel JK(K). By convention, IK(K) is the lower-numbered subchannel of the pair, JK(K) is the higher-numbered subchannel of the pair, and the crossflow through the gap K is positive when passing from IK(K) to JK(K). Flow the other way is negative. The nominal gap width is specified as GAPN(K) and the distance between the centers of the connected subchannels is LENGTH(K). GAPN(K) and LENGTH(K) define the width and length of the transverse momentum mesh cell. The flow area between the connecting subchannels is given by the product of the gap width and the axial length increment for the mesh, DXS (to be read on CARD 4.2). The gap width, GAPN(K), is equal to the total width for transverse flow between the two adjacent regions modeled by the subchannels IK(K) and JK(K).

Form and wall drag in gap K is specified for the transverse momentum equations using the parameters WKR(K) and FWALL(K).

The form drag loss coefficient (velocity head), WKR(K), is specified by the user. The value of WKR(K) for a gap depends on the geometry of the flow path modeled. A value of 0.5 is typically used for flow across one row of rods or tubes. For lumped subchannels, this value is multiplied by the number of rod rows the gap momentum cell contains.

Wall friction in the gap is ordinarily included in the WKR parameter, but for gaps that are formed by vessel walls rather than rod arrays (as in the downcomer annulus, for example) the pressure loss in the gap is primarily a friction loss rather than a form loss and should be modeled accordingly. Wall friction factors in a gap are computed internally in the code according to the user-specified value of the FWALL(K) parameter. If FWALL(K) is zero, no wall friction is calculated (gap formed by a rods only). If FWALL(K) is set equal to 0.5, a wall with a surface of DXS\*LENGTH(K) is assumed (one side of the gap control volume is a solid wall). If FWALL(K) is set equal to 1.0, a wall with a surface of 2\*DXS\*LENGTH(K) is assumed (a solid wall on each side of the gap control volume). Both a form loss and a wall friction factor can be specified for a gap. One or the other can be set to zero depending of the geometry of the problem. Both can be set to zero if the gap is in an open fluid region where the fluid-to-fluid shear within the continuous phase adequately models the transverse drag forces.

The input required to describe transverse convection of axial momentum between gaps across section boundaries employs the same sort of reasoning as does the input in CARD GROUP 2 to describe the axial convection of transverse momentum across section boundaries. The user must specify which transverse gap velocities can transport axial momentum into or out of the subchannels associated with gap K at the top and bottom of the section. The transverse velocity of gap IGAPB(K) (at node NONODE+1 in the section below) can convect axial momentum into or out of the bottom cell of the subchannels associated with gap K. The transverse velocity of gap IGAPA(K) (at node 2 in the section above) can convect axial momentum into or out of the top cell of the subchannels associated with gap K. Transverse convection of axial momentum at a section boundary is illustrated in Figure 22.8. In this example, axial momentum is convected between the two axial mesh cells by the transverse velocity of gap 1 below the section boundary and by the transverse velocity of gap 2 above the section boundary. The input for IGAPA(K) identifies the gap in the section above gap K that convects axial momentum between the axial momentum cells associated with the top of gap K. For the example shown in Figure 22.8, IGAPA(K) = 2 and the corresponding input will have the following format:

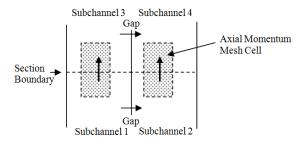


Figure 22.8: Axial momentum mesh cell at section boundary

*	**********************															
*	Card	3.2	âĂŞ	Gap 1												
*	K	IK	JK	GAPN	LNGT	WKR	FWAL	IGPB	IGPA	FACT	IGAP	JGAP	IGAP	JGAP	IGAP	JGAP
	1	1	2	X.X	<b>X.X</b>	X.X	X.X	0	2	X.X	Х	Х	Х	Х	Х	Х
*	*****	****	<b>**</b> **	*****	*****	****	k****	k****	k****	*****	k****	****	k****	****	k****	****

The remainder of the input for CARD 3.2 is required only for the three-dimensional form of the transverse momentum equation. If the subchannel formulation is desired for a specific problem, this input is omitted. It is used to define consistent transverse flow directions for the global coordinate system and to set up connections for orthogonal transport of momentum.

The normal crossflow sign convention is not always congruent with a global coordinate system. The user can convert the local transverse flow sign convention to an appropriate global system by specifying the variable FACTOR(K) to indicate the orientation of gap K in the global coordinate system. Figure 22.9 (a) gives two examples how FACTOR can be used. Example (a) shows an annular subchannel arrangement typically used in the downcomer. The annular flow subchannels are two-dimensional, so it is only necessary to define a transverse direction that is consistently positive. The normal convention for defining positive crossflow results in crossflow from subchannel 1 to subchannel 2 and from subchannel 3 to subchannel 4 is to have both being positive. Assuming that the clockwise direction is chosen as positive for the global coordinate system, if flow from IK(K) to JK(K) is clockwise, then FACTOR(K) is set to 1.0. If flow from IK(K) to JK(K) is counterclockwise, the sign convention must be reversed by setting FACTOR(K) to -1.0. In this example, FACTOR(2) is set to -1.0 and the input will have the following format:

```
ard 3.2 âĂŞ Gap 1
                GAPN
                             WKR FWAL IGPB IGPA FACT IGAP JGAP IGAP JGAP JGAP
                       LNGT
                 X.X
                        X.X
                             X.X
                                   X.X
                                           0
                                                0
                                                   1.0
                                                           X
                                                                 X
                                                                      X
                                                                            X
* Card 3.2 âĂŞ Gap 2
   K
       TK
                GAPN
                       LNGT
                             WKR FWAL IGPB IGPA FACT IGAP JGAP IGAP JGAP IGAP
   2
                 X.X
                        {\tt X.X}
                             X.X
                                   X.X
                                           0
                                                  -1.0
                                                           X
                                                                 X
                                                                      X
                                                                            X
                                                                                 X
                                                                                       X
 Card 3.2
           âĂŞ
                Gap 3
       ΙK
            JK
                GAPN
                       LNGT
                             WKR FWAL IGPB IGPA FACT IGAP
                                                             JGAP
                                                                   IGAP
   K
                                                                        JGAP
        2
   3
             3
                 X.X
                        {\tt X.X}
                             X.X
                                   X.X
                                                                 Х
* Card 3.2 âĂŞ Gap 4
       ΙK
           JK
                GAPN
                       LNGT
                             WKR FWAL IGPB IGPA FACT IGAP
                                                             JGAP IGAP JGAP IGAP JGAP
                 X.X
                        X.X
                             X.X
                                   X.X
                                                   1.0
                                                           X
                                                                 Х
                                                                      Х
                                                                                 Х
```

A three-dimensional subchannel array is shown in example (b) in Figure 22.9. For this case, the values of FACTOR(K) must correspond to the actual orientation of the velocity vector in the gap on the global coordinate system. The arrows in the gaps indicate the positive crossflow direction defined by the normal convection. For gaps 1, 3, 6, and 7, this is the same as the global coordinates, so FACTOR(1), FACTOR(3), FACTOR(6), and FACTOR(7) are 1.0 for this case. But the positive flow direction in gaps 2, 4, and 5 is exactly opposite of the global convention, so FACTOR(2), FACTOR(4) and FACTOR(5) must be specified as -1.0 to reverse the sign on the crossflow velocity. The input will have the following format:

*****	****	****	*****	*****	****	*****	****	****	*****	****	****	****	****	****	****
* Card	3.2	âĂŞ	Gap 1												
* K	IK	JK	GAPN	LNGT	WKR	${\tt FWAL}$	IGPB	${\tt IGPA}$	FACT	IGAP	JGAP	IGAP	JGAP	IGAP	JGAP
1	1	2	X.X	X.X	X.X	X.X	0	0	1.0	X	Х	Х	X	X	Х
* Card	3.2	âĂŞ	Gap 2												
* K	IK	JK	GAPN	LNGT	WKR	FWAL	IGPB	IGPA	FACT	IGAP	JGAP	IGAP	JGAP	IGAP	JGAP
2	1	4	X.X	X.X	X.X	Х.Х	0	0	-1.0	Х	Х	Х	Х	Х	Х
* Card	3.2	âĂŞ	Gap 3												
* K	IK	JK	GAPN	LNGT			IGPB		FACT				JGAP	IGAP	JGAP
3	2	3	Х.Х	Х.Х	X.X	Х.Х	0	0	1.0	Х	Х	Х	Х	Х	Х
• • •		u													
* Card		-	-												
* K	IK	JK	GAPN	LNGT					FACT						JGAP
4	2	5	X.X	X.X	Х.Х	Х.Х	0	0	-1.0	Х	Х	Х	Х	Х	Х
• • •		. 🗸													
* Card		_	-												
* K	IK	JK	GAPN	LNGT					FACT						
5	3	6	X.X	X.X	X.X	Х.Х	0	0	-1.0	Х	Х	Х	Х	Х	Х
• • •		. Y													
* Card		_	-	T 110m		T11.4.T	Tabb	Tabl	D 4 G M	TOAR	7015	TOAR	7015	TOAR	7015
* K	IK	JK	GAPN	LNGT					FACT						
6	4	5	X.X	X.X	х.х	Х.Х	0	0	1.0	Х	Х	Х	Х	Х	Х
		^ ¥ a	0 7												
* Card		,	-	TNOT	THE	TTTAT	TADD	Taba	г л от	TOAD	TOAD	TOAD	TOAD	TOAD	TOAD
* K	IK	JK	GAPN	LNGT					FACT						
7	5	6	X.X	X.X	X.X		0	0	1.0	Х	Х	Х	X	X	X
*************************															

For the three-dimensional form of the transverse momentum equation, the user must specify the gaps facing a given gap K. This information is required in the code to calculate the normal components  $(\rho V^2 A)$  of transverse momentum. The arrays IGAP(K,N) and JGAP(K,N) on CARD 3.2 are used to supply this data. The IGAP array holds the identification numbers of up to three gaps on the IK (or lower-numbered subchannel) side of gap K. The JGAP array holds the identification numbers of up to three gaps on the JK (or higher-numbered subchannel) side of gap K. This is illustrated in Figure 22.10. In this example, gap 4 connects subchannels 1 and 2, with the conventional positive flow direction shown by the arrow. Gaps 1 and 3 face the JK side of gap 4, and gaps 5 and 6 face the JK side. The input will have the following format:

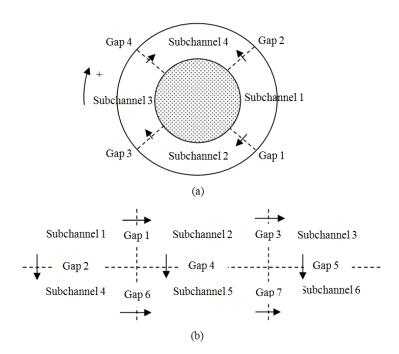


Figure 22.9: Global coordinate systems

***************************************									****							
*	Card	3.2	âĂŞ	Gap 1												
*	K	IK	JK	GAPN	LNGT	WKR	FWAL	IGPB	IGPA	FACT	IGAP	JGAP	IGAP	JGAP	IGAP	JGAP
	1	1	3	Х.Х	Х.Х	X.X	Х.Х	0	0	-1.0	4	-1	0	0	0	0
	•															
*	Card	3.2	âĂŞ	Gap 2												
*	K	IK	JK	GAPN	LNGT	WKR	FWAL	IGPB	IGPA	FACT	IGAP	JGAP	IGAP	JGAP	IGAP	JGAP
	2	3	4	Х.Х	Х.Х	X.X	Х.Х	0	0	-1.0	-1	-1	0	0	0	0
	•															
*	Card	3.2	âĂŞ	Gap 3												
*	K	IK	JK	GAPN	LNGT					FACT		JGAP	IGAP	JGAP	IGAP	JGAP
	3	1	4	Х.Х	Х.Х	X.X	Х.Х	0	0	1.0	4	-1	0	0	0	0
	•															
*	Card		âĂŞ	Gap 4												
*	K	IK	JK	GAPN	LNGT			IGPB	IGPA	FACT	IGAP		IGAP	JGAP	IGAP	JGAP
	4	1	2	Х.Х	Х.Х	X.X	Х.Х	0	0	1.0	1	5	3	7	0	0
	•															
*	Card	3.2	âĂŞ	Gap 5												
*	K	IK	JK	GAPN	LNGT	WKR	FWAL	IGPB	IGPA	FACT	IGAP	JGAP	IGAP	JGAP	IGAP	JGAP
	5	2	5	Х.Х	Х.Х	X.X	Х.Х	0	0	1.0	4	-1	0	0	0	0
	•		u													
			-	Gap 6												
*	K	IK	JK	GAPN	LNGT					FACT						JGAP
	6	5	6	Х.Х	X.X	X.X	Х.Х	0	0	-1.0	-1	-1	0	0	0	0
	•		u													
*	Card	3.2	âĂŞ	Gap 7												
*	K	IK	JK	GAPN	LNGT	WKR	FWAL	IGPB	IGPA	FACT	IGAP	JGAP	IGAP	JGAP	IGAP	JGAP
	7	2	6	<b>X.X</b>	<b>X.X</b>	X.X	Х.Х	0	0	1.0	4	-1	0	0	0	0
**	*************************															

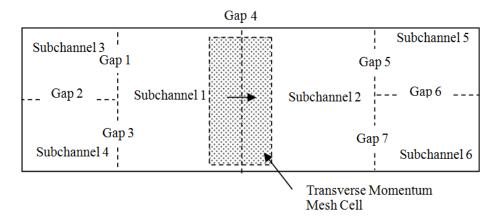


Figure 22.10: Axial momentum mesh cell at section boundary

CARD 3.3 must be read for each gap, following CARD 3.2. This input is designed to model gaps between subchannels that model large regions in the vessel. The nominal gap width, GAPN(K) (supplied on CARD 3.2), is the total width for crossflow between subchannels. The characteristic width for two-phase flow through the gap, however, is the distance between the physical structures and not necessarily the total width of the connection between mesh cells. The parameter GMULT(K), input on CARD 3.3, is the number of uniform spaces between structures in the region modeled by the gap. The characteristic dimension for two-phase flow is given by:

$$S' = \frac{\text{GAPN(K)}}{\text{GMULT(K)}}$$

The dimension S' determines the maximum bubble size that can pass between structures. If GAPN(K) is the actual width available for flow, GMULT(K) is set to 1.

ETANR(K), which is also read on CARD 3.3, is the fraction of transverse droplet flow de-entrained on structures within the mesh cell. In a reactor vessel or test section the array of rods in the core and support columns and guide tubes in the upper plenum can contribute significantly to de-entrainment of droplets from the transverse component of the flow. The de-entrainment fraction, ETANR(K), for a gap modelling an array of tubes is:

ETANR(K) = 
$$1 - (1 - \eta_R)^N$$

where  $\eta_R$  is the de-entrainment fraction of a single row of tubes and N is the number of tubes. The de-entrainment fraction of a single row of tubes is:

$$\eta_R = \eta_I (1 + 4.5\beta^2)$$

where  $\eta_I = 0.19$  for cylindrical tubes and  $\eta_I = 0.27$  for square tubes. N is the pitch-to-diameter ratio of the array.

ETANR(K) can be used to model the crossflow de-entrainment rate for geometries other than tube banks and rod bundles, but the user must determine the appropriate value for each particular application.

CARDS 3.4 and 3.5 are required only for the three-dimensional form of the transverse momentum equation. The total number of gaps that convect orthogonal transverse momentum is specified as the NLMGAP parameter on CARD 3.4. The velocity of the gap identified in KGAP1(N) convects transverse momentum from KGAP3(N) to KGAP2(N) if the velocity is positive and from KGAP2(N) to KGAP3(N) if negative. This is illustrated in Figure 22.11. In this example, the velocity of gap 1 convects momentum from the left half of the momentum mesh cell for gap 4 to the left half for gap 2. The velocity of gap 3 convects momentum from the right half of the momentum mesh cell for gap 4 to the right half for gap 2. Gaps 2 and 4 also convect momentum

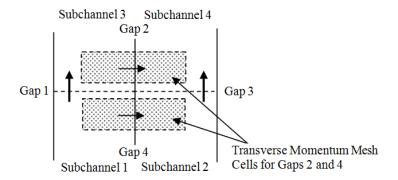


Figure 22.11: Convection of transverse momentum by an orthogonal transverse velocity

from gap 1 to gap 3. Then, KGAP1(1) = 1; KGAP2(1) = 2; KGAP1(2) = 3; KGAP3(2) = 4; KGAP1(3) = 2; KGAP2(3) = 3; KGAP3(3) = 1; KGAP1(4) = 4; KGAP2(4) = 3; and KGAP3(4) = 1. The input will have the following format:

In general, the convection of transverse momentum by transverse velocities in the gap between local and global (lumped) mesh cells is neglected. The transverse flow between local mesh and global mesh is assumed to be axisymmetric. The shape of the transverse momentum cell is idealized and requires the user's judgment in selecting values for the gap width and length. The gap width should be chosen such that the product of the gap width and the axial mesh length increment is equal to the physical area for the flow path between the local and global mesh cells. The shape of the momentum mesh cell is assumed rectangular and its length should be chosen to give a physically meaningful time constant for accelerating flow through the gap. A distance equal to the diameter of the local mesh is recommended for the centroid distance (LENGTH(K)).

Setting up the gap geometry data is relatively simple even for peculiar geometries if the user bears in mind that GAPN(K) and LENGTH(K) define the transverse control volume for the gap. The size and shape of such a control volume should bear some logical resemblance to the physical structure being modeled, within the constraints of the nodding philosophy used in the code. The control volume is important not only for defining the location and magnitude of the transverse flow, but also for determining axial transport of transverse momentum through the top and bottom surfaces of the control volume. Both aspects of the momentum solution must be considered in defining the gap width and centroid length.

# 22.2.3 Instructions to CARD GROUP 4

The input data for CARD GROUP 4 describes the axial subchannel connections between sections and defines the rebalancing and simultaneous solution group options.

CARD 4.1 provides the input for the total number of axial sections in the problem, NSEC; the total number of the simultaneous solution groups, NSIM, in the iterative pressure matrix solution; and the flag IREB

for selection of the rebalancing option for enhancement of the convergence rate for the iterative solution (parameters NSIM and IREB will be discussed later in the text).

CARD 4.2 provides the input data describing each axial section in the problem. It contains the section number, ISEC; the number of subchannels, ICHN, in the section; the number of axial nodes, NONODE, in the section; the axial node length, DXS, in the section; and the flag IVARDX for variable node length in the section. The section boundaries are uniform at a given axial level, so all subchannels within the section have the same total length. The axial node length DXS defines the length of both the continuity and momentum control volumes. DXS is constant within the section and therefore the total axial length of the section is DXS\*NONODE. DXS can vary between sections and each section may have one or many axial nodes. The only constraint is that the change in DXS between adjacent sections should not be greater than factor of two.

CARD 4.3 specifies a variable axial node length within a section. The input data contains the last axial level JLEV(I) in a section to have a node length of VARDX(I), where I is the number of pairs to be read and it is equal to the parameter IVARDX read on CARD 4.2. For the example shown in Figure 22.12, the input will have the following format:

```
* NSEC NSIM IREB NDM4 NDM5 NDM6 NDM7 NDM8 NDM9 NM10 NM11 NM12 NM13 NM14
          1
                    0
                         0
                              0
                                   0
    1
* Card 4.2
 ISEC NCHN NONO
                      DXS TVAR
                  0.02556
* Card 4.3
 JLEV VRDX JLEV
                  VRDX JLEV
                             VRDX
    5 0.04
                 0.01
                          10 0.02
```

CARD 4.4 specifies the axial connections for each subchannel in a section. A subchannel may connect up to six subchannels in the section above and six subchannels in the section below. The array KCHANA contains the index numbers of subchannels connected to the top of the subchannel and the array KCHANB contains the index numbers of the subchannels connected to the bottom. If a subchannel does not have a connection above or below, KCHANA or KCHANB is specified with the subchannel's own identification number. The code will not accept a subchannel with only zeros in the KCHANA or KCHANB arrays.

There are three basic patterns possible for vertical connections between subchannels:

- (a) one subchannel below connected to one subchannel above;
- (b) one subchannel below connected to many subchannels above; and
- (c) multiple subchannels below connected to a single subchannel above

As previously discussed, the momentum equation for the axial velocity at the section boundary is solved at the top node of the bottom subchannel for case (a); in the bottom node of each of the upper subchannels in case (b); and in the top node of each lower subchannel in case (c). The momentum mesh cells at the section boundary for these three cases are shown in Figure 22.13. Velocities are solved for in the momentum cells at the top of subchannel 1 for case (a); at the bottom of subchannels 2, 3, and 4 for case (b); and at the top of subchannels 1, 2, and 3 for case (c).

Velocities at the momentum cell faces, which are used to calculate the momentum flux terms, are obtained by averaging the velocities at the cell centers of adjacent momentum cells. At section boundaries where the

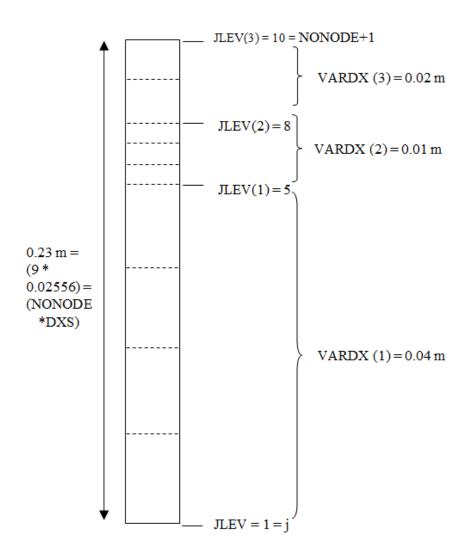


Figure 22.12: Diagram of the variable axial node length

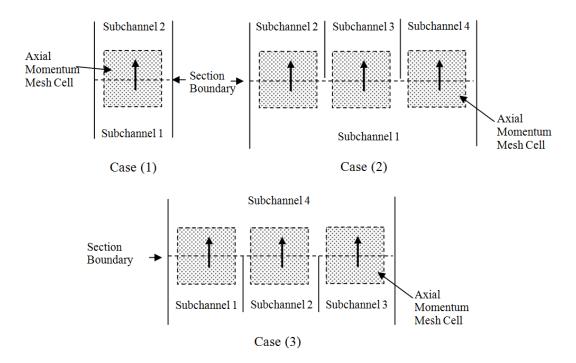


Figure 22.13: Allowable vertical connections between subchannels at section boundaries

connections are like those shown in cases (a) and (c) in Figure 22.13, the velocity at the top face of the momentum cell is obtained by averaging the velocities at the centers of the momentum mesh cells on the boundary with the velocity at the center of the first momentum mesh cell in the subchannel above. For case (a), the velocity of subchannel 2, node 2, is averaged with the velocity calculated in subchannel 1 at the boundary. For case (c), the velocity of subchannel 4, node 2, is averaged with the velocities calculated in subchannels 1, 2, and 3 at the boundary. For case (b), the velocity calculated at the last cell of subchannel 1 is averaged with velocities calculated in subchannels 2, 3, and 4 at the boundary. A subchannel with multiple connections to both the bottom and top must contain at least two mesh cells. If the subchannel had only one cell, the code would not be able to determine which velocities to use when obtaining average velocity at the cell face for momentum cells at the bottom of the subchannel or at the top of the subchannel.

Figure 22.14 shows two examples of subchannel connections that are not permitted by the subchannel splitting logic of the code. In example (a), the subchannels below overlap in their connections to the subchannels above. In example (b), subchannel 3 is only one cell long. The input for the correct subchannel splitting example shown in Figure 22.14 (b) will have the following format:

```
* Card 4.1
 NSEC NSIM IREB NDM4 NDM5 NDM6 NDM7 NDM8 NDM9 NM10 NM11 NM12 NM13 NM14
                                               0
          1
               0
                          0
                               0
                                     0
                                          0
                                                     0
                                                          0
* Axial Section 1
* Card 4.2
 ISEC NCHN NONO DXS IVAR
     1
          2
                  x.x
* Card 4.4
     I KCHA KCHA KCHA KCHA KCHA KCHB KCHB KCHB KCHB KCHB
          3
               0
                     0
                          0
                               0
                                               0
                                                     0
                                                          0
                                                                0
                                                                     0
     1
                                     0
                                          1
               0
                                          2
     2
          3
                     0
                          0
                               0
                                     0
                                               0
                                                     0
                                                          0
                                                               0
                                                                     0
* Axial Section 2
* Card 4.2
 ISEC NCHN NONO
                  DXS IVAR
          1
                  x.x
* Card 4.4
     I KCHA KCHA KCHA KCHA KCHA KCHB KCHB KCHB KCHB KCHB KCHB
                     6
                          0
                               0
                                                     0
                                     0
                                          1
                                               2
                                                          0
* Axial Section 3
* Card 4.2
 ISEC NCHN NONO
                  DXS IVAR
     3
          3
                  x.x
* Card 4.4
     I KCHA KCHA KCHA KCHA KCHA KCHB KCHB KCHB KCHB KCHB KCHB
                     0
                          0
                               0
                                     0
                                          3
                                               0
                                                     0
     5
          5
               0
                     0
                          0
                               0
                                     0
                                          3
                                               0
                                                     0
                                                                0
                                                                     0
                     0
                                                                     0
     6
          6
                          0
                               0
                                     0
```

The remainder of CARD GROUP 4 specifies the numerical solution procedure to be used for solving the linearized continuity and energy equations. The equations may be solved by a direct inversion of the system pressure matrix or a combination of direct inversion and Gauss-Seidel iterative method. (The pressure matrix can be solved using different Krylov iterative solvers, but this option is controlled by the flag ISOL entered in CARD GROUP 1, ISOL > 0). In the combination solution, the pressure matrix for groups of cells is solved by direct inversion while the pressure in cells surrounding each group is held constant. Gauss-Seidel iteration is then performed over the groups of cells to obtain a converged solution for the entire system pressure matrix.

The solution procedure used is determined by the parameter NSIM read on CARD 4.1. If the number of the simultaneous solution groups NSIM is set to one the solution is obtained by direct inversion. If NSIM is set to N (where N is an integer greater than one) the solution is obtained with a combination of direct inversion and iteration with N simultaneous solution groups. The convergence rate for the iterative solution can be enhanced by specifying the rebalancing option, IREBAL = 1. When rebalancing is done, a one-dimensional solution for the linear pressure variation at each axial level is obtained by direct inversion. This value is used as an initial estimate for the linear pressure variation in each cell in the group-by-group iteration.

A direct inversion is recommended for relatively small problems (up to 100 mesh cells). The Gauss-Seidel iterative technique is recommended for larger problems. However, experience indicates that the significant speed-up with group-by-group iteration is observed only when stationary conditions or transients not involving mass flow rate variation are simulated. For flow transients, the Gauss-Seidel solver converges slowly, leading to a tremendous increase of the CPU time (such lack of convergence cannot be overcome by using the rebalancing option). Therefore, for larger problems, the Krylov solvers are highly recommended because of their competitive efficiency and better accuracy comparing to the Gauss-Seidel method.

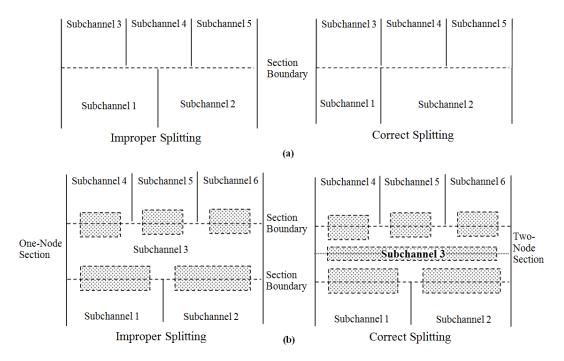


Figure 22.14: Common subchannel splitting errors

The bandwidth of the pressure matrix is defined by the maximum difference between the index numbers of adjacent cells in a group that is being solved simultaneously. This maximum difference is specified by input as parameter IWIDE on CARD 4.5, and the maximum bandwidth of the pressure matrix is calculated as 2\*IWIDE. The cell numbers used to define IWIDE are not subchannel numbers. Cell numbers are assigned in the following manner: beginning at the first subchannel in the first axial section, the first node in the subchannel (node J=2) is assigned cell number one (1). The cells of all the subchannels in that section are numbered sequentially at the same level (J=2). The cell numbering is continued on the next level, starting in the same first subchannel. This process continues until all cells for all axial sections have been assigned unique cell numbers. An example illustrating this numbering convention is shown in Figure 22.15. In the example, IWIDE is equal to 2 in simultaneous solution groups 1 and 5. In groups 3 and 4, IWIDE = 1. In group 2 IWIDE is 3. The value of IWIDE entered in CARD 4.5 will then be 3 (the matrix must accommodate the largest bandwidth in all groups).

The MSIM array—the number of the last cell in each simultaneous solution group—is provided by CARD 4.6. The input for the example shown in Figure 22.15 will have the following format:

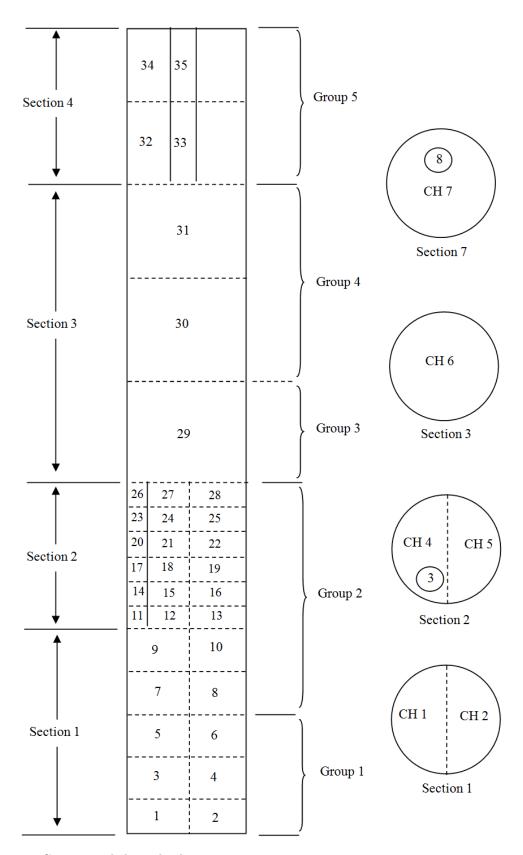


Figure 22.15: Common subchannel splitting errors

### 22.2.4 Instructions to CARD GROUP 5 and 6

The geometry of a subchannel or a gap is assumed to remain constant along the entire axial length at the nominal values specified by input. But CARD GROUPS 5 and 6 permits the user to specify axial variation in the geometry. This is an optional input and it is omitted if axial variation data is not needed. The input data is extremely general. The user provides tables of variation factors in CARD GROUP 5. Variation tables can be supplied for the continuity flow area, momentum flow area, and wetted perimeter of subchannels and the width of gaps. Variation tables are read on CARD 5.3 as tables of node number, JAXL(I,N), versus variation factor AFACT(I,N). The node number JAXL(I,N) refers to the continuity cell for continuity area and gap width and to the momentum cell for the momentum area and wetted perimeter. The variation factor is defined as:

$$\texttt{AFACT(I,L)} = \frac{local value}{nominal value}$$

The value AFACT(I,N) is applied in the code as a multiplier on the nominal value of the quantity being varied at the corresponding JAXL(I,N) node. Figure 22.16 shows two examples of subchannels with area variations that can be modeled with variation tables. For example (a), variations of the continuity area occur at nodes 7, 8, and 9. Variations in the momentum area also occur at nodes 7, 8, and 9, but because the geometry of the subchannel is tapered and the continuity and momentum nodes are staggered, the areas vary by different amounts. The tapered region in the subchannel is approximated by a stack of cells that have a uniform cross section along their individual lengths. The continuity areas of the cells should be defined so that the volumes of the cells are equal to the volumes of the regions they are intended to model. In the continuity solution, cells 7 and 8 model the tapered region. The momentum cells use the actual area at the location of the momentum cell center. In this example, the momentum area variation is modeled in cell 7. The wetted perimeter is defined in the momentum cell, so variations in wetted perimeter must be located relative to the momentum cells. In example (a) of Figure 22.16, the gradually changing wetted perimeter must be approximated by step changes, as is the gradually changing flow area in the taper. For the circular cross section of this example, the simplest approach is to define the wetted perimeter as the perimeter of the momentum cells:

$$P_w = \pi D'$$

where:

$$D' = \sqrt{\frac{4A_{momcell}}{\pi}}$$

The input for CARD GROUP 5 to model the variations in example (a) of Figure 22.16 consists of three variation tables (NAFACT = 3, read on CARD 5.1). The tables themselves are read on CARD 5.2 (specifying the number of entrees in a table) and on CARD 5.3 (filling the arrays for the node indices, JAXL, and variation factors, AFACT). The tables are numbered sequentially in the order in which they are read.

Example (b) in Figure 22.16 shows a different sort of area variation. The orifice plate in the straight pipe affects the momentum solution, but because it occurs over a relatively short distance compared to the node length, the continuity solution is largely unaffected. This could be modeled with a momentum area variation alone without any variation in the continuity cell area.

Gap width variations are specified in the same manner as axial variations in the wetted perimeter. The nodes that differ from nominal are identified in the JAXL array and the corresponding variation factors in the AFACT array.

The input for the example shown in Figure 22.16 (a) will have the following format:

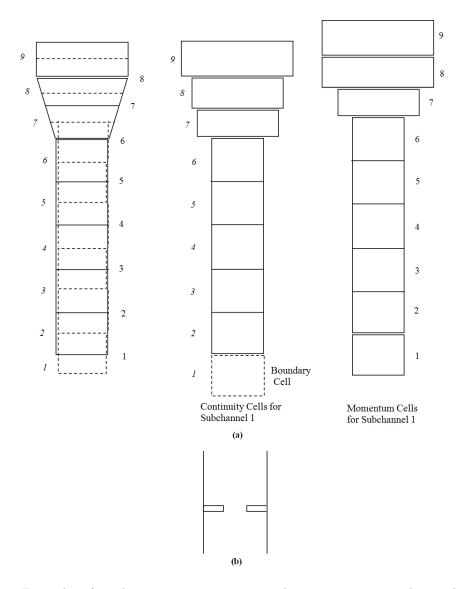


Figure 22.16: Examples of axial variation in continuity and momentum area and wetted perimeter of a subchannel

```
* GROUP 5.0 - Geometry Variation Data
***************
* NGR
   5
* Card 5.1
* NFCT NAXL NDM3 NDM4 NDM5 NDM6 NDM7 NDM8 NDM9 NM10 NM11 NM12 NM13 NM14
    3
        5
                 0
                      0
                               0
                                    0
                                        0
* Card 5.2
 JAXL AFACT JAXL AFACT JAXL AFACT JAXL AFACT
    1 1.000
              6 1.000
                        7 x.xxx
                                  8 у.ууу
    1 1.000
              6 1.000
                        7 x.xxx
                                  8 y.yyy
                                            9 y.yyy
    1 1.000
              6 1.000
                        7 v.vvv
                                            9 w.www
```

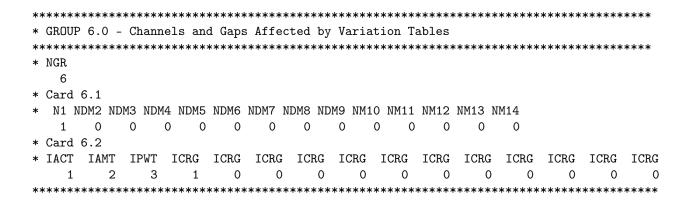
Variations encountered in most problems for CTF will not be as neat, generally, as the two examples shown in Figure 22.16. The user will have to make approximations appropriate for the particular geometry involved and noding selected. Some general guidelines in setting up axial variations are:

- 1. In complex geometries, the user should preserve the same fluid volume as in the actual system.
- 2. The momentum area should approximate the actual flow area in the system at sudden changes.
- 3. The code interpolates linearly in the table to obtain variation factors for cells within the range of the table but not named explicitly. (Cells with identification numbers lower than the first element of a variation table or greater than the last element in the table remain at nominal values).
- 4. The area in either the continuity cells or the momentum cells should not change by more than a factor of two between adjacent cells, even if the area in the system actually changes more abruptly. The code must be led gradually through a large change in a series of steps.

The input for CARD GROUP 6 specifies the subchannels or gaps to which the variation tables described in CARD GROUP 5 apply. Variation table assignments are read on CARD 6.2 for either subchannels or gaps. The variation tables are identified by sequence number; i.e., the first table read in CARD GROUP 5 is Table 1; the second table read in CARD GROUP 5 is Table 2; etc. The index number of the table to be used for continuity area variations is specified by IACT. The index number of the table to be used for momentum area variations is specified by IAMT. The index number of the table to be used for wetted perimeter variations is specified by IPWT. The numbers of the subchannels using the variation tables named by IACT, IAMT, and IPWT are listed in array ICRG(M). Axial variation in gap width is specified by setting IACT to the negative of the variation table number, and naming the indices of the gaps using the table in array ICRG(M). When gap variations are specified, IAMT and IPWT are not used.

The input for CARD 6.2 is repeated until all subchannels and gaps having variations have been identified. It is not necessary to specify variations in continuity area, momentum area and wetted perimeter in a subchannel simultaneously. Any or all can be used, as is appropriate for a given problem. It is possible to specify only a continuity variation, only a momentum area variation, only a wetted perimeter, or any combination of the three.

The input for the example shown in Figure 22.16 (a) will have the following format:



### 22.2.5 Instructions to CARD GROUP 7

The input for CARD GROUP 7 is designed to model local pressure losses in the axial flow due to spacer grids, orifice plates and other local obstructions in the flow field.

In the COBRA/TRAC code version [5], the local pressure losses in vertical flow are modeled as velocity head loss given by

$$\Delta P = \zeta \rho \frac{v^2}{2g_c},\tag{22.1}$$

where  $\zeta$  is the pressure loss coefficient,  $\rho$  is the density, v is the flow velocity in vertical direction, and  $g_c$  is the gravitational conversion constant. The pressure loss coefficients are defined assuming positive upflow in a channel and specified for a momentum (not continuity) cell containing the obstruction.

Later, grid heat transfer models for convective enhancement downstream of the spacers, grid rewet during bottom reflood, and droplet breakup on spacers were included in Paik et al. [2]. In this code version, rather than using input specified values for the spacer loss coefficients, the spacer loss coefficients are calculated from grid dimensions using

$$\zeta_{grid} = \min(20, 196Re_{mix}^{-0.333}) f_{loss} (A_{blocked}^{spacer} + A_{blocked}^{springs})^2, \tag{22.2}$$

where

 $f_{loss}$  is the pressure loss coefficient multiplier (input value),

 $A_{blocked}^{spacer}$  is the fraction of channel area blocked by the grid (input value),

 $A_{blocked}^{springs}$  is the fraction of channel area blocked by the grid springs (input value), and

 $Re_{mix}$  is the drops/bubbles mixture Reynolds number.

In the current code version, pressure losses must be specified either by  $\zeta$  values or by geometrically modeled flow blockages. If pressure loss coefficients, CDLs, are specified in CARD 7.2, then cards CARD 7.3 to CARD 7.9 must be omitted.

CARD 7.1 specifies the total number of loss coefficients, NCD; the number of grid types, NGT; the flag for grid quench front model IFGQF; the flag for small droplets model, IFSDRP; the flag for grid convective enhancement, IFESPV; the flag for two-phase enhancement of dispersed flow heat transfer, IFTPE; the flag for grid quench calculations, IGTEMP; the flag for flow blockages calculations NFBS; and IXFLOW, the flag for grid spacer influences on lateral exchange over gaps.

If the parameter NCD is equal to N, loss coefficient values CDL(N) are specified on CARD 7.2. CARD 7.2 contains also the identification number J of the axial node at which the pressure loss coefficient is applied and the ICDUM array of identification numbers of the subchannels for which the pressure loss coefficient is applied at axial node J. The pressure losses will be calculated according to Equation 22.1. The models for quench front and grid convective enhancement are not applicable when NGT = 0 (NCD > 0) and the flags IFGQF and IFESPV have to be set to zero in CARD 7.1. The input for the configuration shown in Figure 22.17 will have the following format:

***************************												
* GROUP 7.0 - Local Pressure Loss Coefficient and Grid Spacer Data												
****************************												
* NGR												
7												
* Card 7.1												
* NCD NGT IFGQ IFSD IFES IFTP IGTP NFBS IFCD IXFL NM11 NM12 NM13 NM14												
3 0 0 0 0 0 0 0 0 0 0 0 0												
* Card 7.2												
* CDL J ICD01 ICD02 ICD03 ICD04 ICD05 ICD06 ICD07 ICD08 ICD09 ICD10 ICD11 ICD12												
x.xxx 3 1 2 0 0 0 0 0 0 0 0 0												
x.xxx 5 1 2 0 0 0 0 0 0 0 0 0												
x.xxx 6 1 2 0 0 0 0 0 0 0 0 0												
*************************************												

There are two important points to remember:

- 1. The location of a loss coefficient is determined by node and subchannel number. The node refers to the momentum cell, not the continuity cell. This must be kept in mind when determining the node that corresponds to the location of the local loss in the system being modeled. Care must be used when placing a loss coefficient in a momentum cell at a section boundary. The loss coefficients must be defined in the subchannel where the momentum equation is solved.
- 2. The loss coefficients are defined assuming positive upflow in the subchannel. If the loss coefficient of a particular structure changes significantly when flow reverses through it, the code does not see the change. If reverse flow is the dominating pattern for such a situation, the user should specify the loss coefficient corresponding to reverse flow than the value for positive flow.

If the parameter NGT is greater than zero (0), the pressure loss coefficients will be calculated according to the methodology described in Paik et al. [2]. This option requires input for CARDS 7.3, 7.4, and 7.5. The physical meaning of the parameters that have to be specified is given in the input description for CARD 7. An example of the input for the arrays NGROD and NGSURF is given below. It is a four subchannels problem with three spacer grids of same type ING along the axial length. The axial locations of the grids, NNGL, refer to the momentum cells containing the grids.

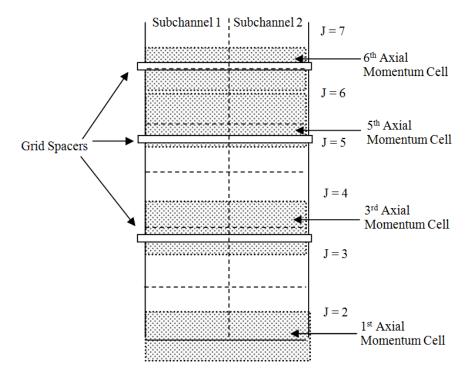


Figure 22.17: Example of subchannels with local form losses due to spacer grids

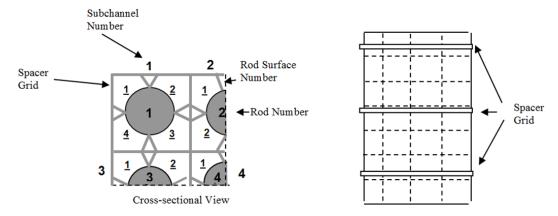


Figure 22.18: Example of rod surface numbering in CARD 7.5

*************************
* GROUP 7.0 - Local Pressure Loss Coefficient and Grid Spacer Data
***************************************
* NGR
7
* Card 7.1
* NCD NGT IFGQ IFSD IFES IFTP IGTP NFBS IFCD IXFL NM11 NM12 NM13 NM14
0  1  0  0  0  0  0  0  0  0
* Card 7.3
* ING NGAL NGCL IGMT GLOSS GABLOC GLONG GPERIM SPBLOC TPROBE
1 3 4 1 1.0 x.xxx x.xxx x.xxx x.xxx x.xxx
* Card 7.4
* NNGL1 NNGL2 NNGL3
2 5 7
* Card 7.5
* NCGL GMLT NGRD NGSF NGRD NGSF NGRD NGSF NGRD NGSF NGRD NGSF
1 0.25
$\begin{array}{cccccccccccccccccccccccccccccccccccc$
3 0.50 1 4 3 1 0 0 0 0 0 0 0
$4\ 1.00  1  3  2  2  3  2  4  1  0  0  0$
***************************************

CARDS 7.6 and 7.7 are needed when flow blockages are modeled. CARD 7.8 provides input data for the grid quench calculations. For a detailed description of the flow blockages modeling, the user should refer to Paik et al. [2].

CARD 7.8 is used if the updated spacer grid quench modeling is desired, which uses the grid re-wet model to calculate grid temperatures.

CARD 7.9 is used if the spacer grid crossflow effects are modeled. When  $\mathtt{IXFLOW} \neq 0$  then this card is required. The user must be able to supply data files on the spacer grids depending on the modeling option chosen. If  $\mathtt{IXFLOW} = 1$ , then the directed crossflow modeling option is activated. If this model is invoked, then files 'xflow data' and 'direct data.inp' must be provided in the working directory.

The first file, 'xflow\_data', must contain a 2D table for the lateral flow rates corresponding to the spacer grid in use. This 2D table should contain vane angles (degrees) in the x-direction and axial distance (m) from the bottom of the rod in the y-direction.

The first non-commented line for the file must contain 4 values:

- 1. the number of axial distances at which data are specified;
- 2. the number of angles for which lateral flow data are provided;
- 3. dummy variable (suggested value of 0); and
- 4. dummy variable (suggested value of 0)

The second non-commented line for the file must contain 1 value: the number of data sets (2D tables) to be read.

The third non-commented line for the file must contain N2 values (as specified by the number of mixing vane angles in the first line): the angles at which data are supplied.

The remaining lines in the document should contain N2+1 values each. The first column corresponds to the axial position (m) and the remaining columns are the spacer multiplier values corresponding to the current axial location and angle.

A sample of this file is provided below. This sample provides data for 2 different angles (0.0 and 15.0 degrees) at 9 different axial locations from 0.0000 m to 2.0000 m. The '0.0000' values would be replaced by the values of the spacer multiplier at the given angle.

```
*******************************
* 2D table for the spacer multiplier
**********************************
* N1 N2 N3 N4
  9
     2
        0
           0
* NSET
        N2_1
              N2_2 (these are the angles)
         0.0
* V_N1 V_N2_1 V_N2_2
0.0000
      0.0000
             0.0000
0.2500 0.0000
             0.0000
0.5000
      0.0000
            0.0000
0.7500 0.0000
             0.0000
1.0000
      0.0000 0.0000
1.2500 0.0000 0.0000
1.5000 0.0000 0.0000
1.7500 0.0000
             0.0000
2.0000 0.0000
             0.0000
```

The data for these files are generally based on CFD calculations and are specific to each spacer grid design. The code linearly interpolates between the angles at which the data are specified to match that of VAL\_ANG in the input deck, CARD 7.9.

# 22.3 Specification of the Conductors' Data

This section provides the user with guidance how to set up the input data required for the solution of the heat transfer and heat conduction. It covers the input for the heated and unheated conductors and specification of their material properties.

#### 22.3.1 Instructions to CARD GROUP 8

The input for CARD GROUP 8 identifies the rods and unheated conductors modeling the solid structures that interact significantly with the fluid in a particular problem.

Rods and unheated conductors are both used to model solid structures in the vessel. There are two significant differences between them. First, rods can model either active or passive elements, but unheated conductors are always passive. Unheated conductors cannot have internal heat sources. Second, the quench front model with fine-mesh re-noding can be applied to rods if needed, but unheated conductors are assumed to never require it.

Examples of vessel structure that can be modeled with rods are:

- 1. an array of nuclear fuel pins;
- 2. an array of electrically-heated fuel pin simulators;
- 3. an electrically-heated annular test section with both cylinders heated;
- 4. a test section with an electrically heated flat plate as the heat source; etc.

Examples of vessel structure that can be modeled with unheated conductors are:

- 1. a control rod guide tube;
- 2. a support column;
- 3. a section of downcomer annulus;
- 4. a canister of a BWR fuel assembly; etc.

CARD 8.1 identifies the number of rods in the problem, NROD; the number of unheated conductors, NSROD; the conduction solution flag, NC; the number of temperature initialization tables, NRRAB; the number of radiation channels, NRAD; the flag for steady-state calculation of rod temperature, NSTATE; the number of time steps between radiation calculations, NXF; the flag for Yamanouchi canister quench model, NCAN; the flag for the radiation heat transfer calculations, RADFLG; and the flag for critical heat flux calculation option.

The conduction solution flag, NC, must be set to select among no conduction (NC = 0); radial conduction only (NC = 1); radial and axial conduction (NC = 2); and radial, axial, and azimuthal conduction (NC = 3). The option for radial and axial conduction is recommended only for problems where a significant axial temperature gradient in the rods is expected. Similarly, if significant axial and azimuthal temperature gradients are expected, the NC parameter is set to three (3). Problems involving reflooding and quenching of very hot surfaces should not use the radial conduction option only. Problems involving relatively gradual heating or cooling of the system can be handled adequately with radial conduction only.

The simulation time can be speed-up by selecting the option for a steady state fuel rod temperature calculation (NSTATE is set greater than zero).

The rod identification parameters are read on CARDS 8.2, 8.3, and 8.4. If NROD is set to zero these cards are omitted. The rods are numbered sequentially from 1 to NROD. Each rod is uniquely identified by its index number, N, and geometry type number IFTYP(N). The geometry type number corresponds to a set of descriptive geometry data specified in CARD GROUP 9. An individual rod may have a unique geometry type, or several rods may be of the same geometry type.

Each rod has identified with it the number of an axial power profile table, IAXP(N). If IAXP(N) is left blank, an axially uniform table with a factor of unity is assumed. An axial profile table can serve any number of rods. The axial power profile tables (and the radial power factors) are specified in CARD GROUP 11. Together, the axial profile tables and the radial power factors define the local power generation in the individual rod.

In modeling large geometries it is sometimes convenient to represent regions of the vessel by average rods. A fuel pin array might be represented by a single average rod. The number of actual rods modeled by an average rod is specified in the variable RMULT(N). The values specified for RMULT can include fractional parts of rods.

The fine-mesh re-noding capability developed to resolve the quench front in reflooding requires some extra input for the rods. If fine-mesh re-noding is to be used for a particular rod, the flag NRENODE(N) is set to

the number of calculational time steps to elapse between re-noding. How often the rod should be re-noded is primarily a function of the reflood rate and the size of time step expected during the reflood portion of the transient. In general, NRENODE(N) should be set that the quench front will not progress further than 1/2 of the minimum node size, DAXMIN(N), between re-noding. For example, if the quench front velocity is 0.5 cm/sec, the maximum time step is 0.05 seconds, and the minimum node size is 0.1 cm, then:

$$NRENODE = INT(\frac{1/2*(0.1)}{(0.5)(0.05)}) = 2$$

If the rod is a tube quenching on the inside surface, then NRENODE(N) should be specified as a negative number and the absolute value of NRENODE(N) is used to determine the re-noding interval.

Variable HGAP(N), entered in CARD 8.2, specifies a constant value for the fuel rod gap conductance. (If a dynamic gap conductance modeling is desired, the required input will be provided by CARD GROUP 9). If the rod N does not model a nuclear fuel rod, HGAP(N) should be set to zero.

The code allows a rod to be included in more than one axial section. The total number of sections containing rod N is specified by the parameter ISECR(N).

The last two variables in CARD 8.2, HTAMB(N) and TAMB(N), provide input data for the heat transfer coefficient for heat loss to ambient from a surface not connected to a subchannel and the sink temperature for ambient heat loss. This option is normally not related to the real fuel rod simulation. It is used in some simulations of experiments when the user wants to account for the heat loss from a non-water-cooled rod to the ambience or to the medium inside the tube.

CARD 8.3 describes the thermal connections between rods and subchannels for heat transfer between fluid and solid surfaces. All rods (and unheated conductors) must be connected to at least one subchannel. (However, not all subchannels have to be connected to a rod or unheated conductor). Parameter NSCHC(IS,K) gives the subchannel number with thermal connection to rod N and parameter PIE(N,K) supplies the azimuthal fraction of rod N thermally connected to subchannel NSCHC(IS,K). Index IS varies from 1 to the total number of sections containing rod N, ISECR(N). Index K varies from 1 to 8 (up to 8 sets of (NSCHC, PIE) may be entered. If the rod N is thermally connected to fewer than 8 channels, enter (0;0.0) until 8 sets of (NSCHC, PIE) have been entered).

If an inside surface of rod N exists, the number of the subchannel connected to the connected to the inside of azimuthal section K of rod N, NISCHC(N,IS,K) is provided by CARD 8.4.

The unheated conductors' identification parameters are read on CARD 8.5. If NSROD on CARD 8.1 is set to zero, CARD 8.5 is omitted. The unheated conductors are numbered sequentially from 1 to NSROD. Each unheated conductor is uniquely identified by its index number, N, and geometry type number ISTYP(N). The geometry type number corresponds to a set of descriptive geometry data specified in CARD GROUP 9. An individual unheated conductor may have a unique geometry type, or several unheated conductors may be of the same geometry type.

The number of actual unheated conductors modeled by a single one unheated conductor is specified in the RMULS(N) variable. The values specified for RMULS can include fractional parts of rods.

Unheated conductors do not generate heat but can transfer heat to and from the fluid and store thermal energy during a transient. For each unheated conductor N the user must specify the heated perimeter of the surface, HPERIM(N). If the conductor is a tube or wall that has contact with the fluid on its inner surface, then the heated perimeter of the inner surface, HPERIMI(N), must be specified as well. The parameter NOSLCH corresponds to the identification number of the subchannel adjacent to the outside surface of the unheated conductor N. Respectively, the parameter NSLCHC corresponds to the identification number of the subchannel adjacent to the inside surface of the unheated conductor N. If no subchannels are connected to the outside/inside surfaces, NOSLCH/NSLCHC are set to zero.

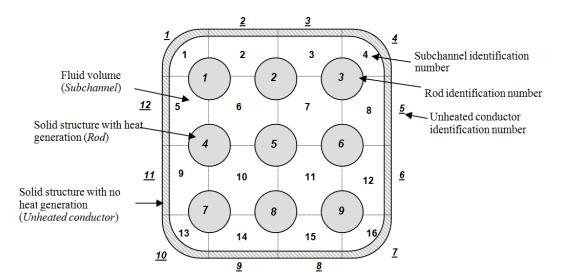


Figure 22.19: 3x3 BWR Bundle

Similar to the rod modeling, the last two variables in CARD 8.6, HTAMBS(N) and TAMBS(N), provide input data for the heat transfer coefficient for heat loss to ambient from surface not connected to a subchannel and the sink temperature for ambient heat loss.

CARDS 8.6 through 8.9 set up the initial surface temperate of the rods and unheated conductors. On CARD 8.9, the temperatures are specified in tables of initial temperatures, TRINIT(I,L), versus axial distance, AXIALT(I,L), relative to the bottom of the vessel. The code interpolates linearly in the table along the axial length of the rod or unheated conductor to which it is applied, so the first element of the AXIALT(I,L) array for a given table I must be at or below the bottom of the rod, and the last element must be at or below the top of the rod. A total of NRTAB (as specified on CARD 8.1) tables must be supplied, but table I can be applied to more than one rod or unheated conductor. On CARD 8.6 the user must specify the number of rods, NRT1(I), and the number of unheated conductors, NST1(I), that use table I. The user must also identify how many pairs of TRINT(I,L) and AXIAL(I,L) elements make up this particular table. The NRAX1(I) pairs of entrees for table I are read on CARD 8.9.

If table I is used to initialize any rods, CARD 8.7 is read to fill array IRTAB(I,L) with the identification numbers of the rods. The steady-state conduction equation is solved for these rods using the temperatures from table I as a boundary condition on the rod surfaces. If any of the rods are tubes or walls and the table is to be used to initialize the temperatures on the inside surface, the rod identification number, N, is entered in the IRTAB array as its negative.

If table I is used to initialize any unheated conductors, CARD 8.8 is read to fill array ISTAB(I,L) with the identification numbers of the unheated conductors. The code assumes an initially flat temperature profile in unheated conductors, so it is not necessary to specify whether the temperature applies to the inner or outer surface.

CARDS 8.10 and 8.11 are needed when radiation heat transfer must be modeled. For a detailed description, the user should refer to [2].

CARD 8.12 provides input data for the Yamanouchi canister quench model (see the CTF Theory Manual [4].

An example of CARD GROUP 8 for a 3x3 BWR bundle follows. The geometry of the bundle is shown in Figure 22.19.

```
************************************
* GROUP 8.0 - Rod and Unheated Conductor Data
* CARD GROUP 8
* NGR.
  8
* Card 8.1
* NRRD NSRD NC NRTB NRAD NLTY NSTA NXF NCAN RADF W3 NM12 NM14
       12 1 1 0 0 0 1 0 0 0 0
* Card 8.2
  N IFTY IAXP NRND DAXMIN RMULT HGAP
1 1 1 0 0.00000 1.000 0.00000
                                HGAP ISECR HTAMB
                                                   TAMB
                                        1
                                           0.000
* Card 8.3
* NSCH PIE             2 0.250 6 0.250 5 0.250
                                                    0 0.000
                                                                        0 0.000
   1 0.250
                                           0.000
                                                               0.000
* Cards 8.2 and 8.3 continued
   2 1 1 0 0.00000 1.000 0.00000 1 0.000 0.000 2 0.250 3 0.250 7 0.250 6 0.250 0 0.000 0 0
                                                    0 0.000 0 0.000
                                                                          0 0.000
                                       1 0.000
       1 1 0 0.00000 1.000 0.00000
                                                   0.000
    3 0.250
            4 0.250 8 0.250
                                7 0.250
                                           0 0.000
                                                     0.000
                                                                0 0.000
                                                                          0 0.000
               0 0.00000 1.000 0.00000
                                        1 0.000
                                                    0.000
    5 0.250
            6 0.250 10 0.250 9 0.250
                                           0 0.000
                                                    0 0.000
                                                                0 0.000
                                                                          0.000
           1 0 0.00000 1.000 0.00000
                                       1 0.000 0.000
    6 0.250
             7 0.250
                      11 0.250
                                10 0.250
                                            0 0.000
                                                      0 0.000
                                                                0.000
                                                                          0 0.000
    6 1 1 0 0.00000 1.000 0.00000 1 0.000 0.000
    7 0.250
            8 0.250 12 0.250 11 0.250 0 0.000
                                                    0 0.000
                                                                0 0.000
                                                                          0 0.000
    7 1 1 0 0.00000 1.000 0.00000 1 0.000
            10 0.250 14 0.250 13 0.250
                                                    0 0.000
                                                                          0 0.000
    9 0.250
                                           0.000
                                                                0.000
               0 0.00000 1.000 0.00000
                                        1 0.000 0.000
   10 0.250
           0 0.000
                                                                0.000
                                                                          0 0.000
    9 1 1 0 0.00000 1.000 0.00000
                                        1 0.000 0.000
   0 0.000
                                                    0 0.000
                                                                0.000
                                                                          0 0.000
* Card 8.5
  N ISTY HPERIM PERIMI RMULT NOSLCHC NSLCHC HTAMBS TAMBS
       2 0.01698 0.00000 1.00
                             1 0
                                          0.000
                                                 0.000
       3 0.01875 0.00000 1.00
                                          0.000
                                                 0.000
       3 0.01875 0.00000 1.00
                            3 0 0.000
4 0 0.000
                                                 0.000
       2 0.01698 0.00000 1.00
                                                 0.000
                              5
8
9
       3 0.01875 0.00000 1.00
                                      0
                                          0.000
                                                 0.000
       3 0.01875 0.00000 1.00
                                      0
                                          0.000
                                                 0.000
                             9
       3 0.01875 0.00000 1.00
                                          0.000
                                     0
                                                 0.000
       3 0.01875 0.00000 1.00
                                          0.000
                                                 0.000

    3 0.01678
    0.00000
    1.00
    13
    0
    0.0000

    3 0.01875
    0.00000
    1.00
    14
    0
    0.000

    3 0.01875
    0.00000
    1.00
    15
    0
    0.000

    2 0.01698
    0.00000
    1.00
    16
    0
    0.000

   9
                                                 0.000
                                                 0.000
  10
                                                 0.000
  11
                                                0.000
  12
* Card 8.6
   I NRT1 NST1 NRX1
       9 12
* Card 8.7
* IRTAB                             6
         2
             3 4
                       5
                                  7
                                       8
                                            9
                                                 0
* Card 8.8
* IRTAB     1 2 3 4 5
                            6
                                  7
                                       8
                                            9
* Card 8.9
  T.TATXA
           TRINIT
0.0000000 300.00000
1.8300000 300.00000
```

#### 22.3.2 Instructions to CARD GROUP 9

The input for CARD GROUP 9 describes the characteristics of geometry types identified in the IFTYP(N) and ISTYP(N) arrays for rods and unheated conductors in CARD GROUP 8. Geometry types fall into two basic classes âĂŞ nuclear fuel rods and all other conductors. The nonnuclear conductors can be characterized as solid cylinders, hollow tubes or flat plates.

On CARD 9.1, the user must specify total number of geometry types, NFUEL. This must be equal to the number of unique entries in the IFTYP(N) and ISTYP(N) arrays. Two flags for fuel relocation, IRELF and ICONF, are also input on CARD 9.1. Relocation is an option in the dynamic gap conductance model. The last entree is the flag for metal-water reaction, IMWR. The option is applicable only for Zirconium dioxide.

CARDS 9.2 through 9.5 are related to the nuclear fuel rod geometry only and must be omitted if fuel rods are not modeled. More than one nuclear fuel geometry type can be specified. The input on CARD 9.2 sets the geometry type index, I, which corresponds to the IFTYP(N) value of the rods that are of this geometry type. The geometry flag FTYPE(I) for nuclear geometry types is NUCL. The geometry of the nuclear fuel rod type is defined by the outside diameter, DROD; the pellet diameter, DFUEL(I); the diameter of the central core, DCORE (zero if the fuel is solid); and the clad thickness, TCLAD. The code assumes that the rod is uniform in axial direction, so these data completely characterize the physical dimensions of the rods being modeled by this geometry type.

On CARD 9.2, NFUEL defines the number of radial heat transfer nodes in the fuel pellet. NFUEL must be large enough for the code to resolve the temperature profile in the fuel pellet adequately, but more nodes specified will increase the time and space expenses when solving the heat conduction calculations. NFUEL between 5 and 10 is a good approximation for a fuel pellet diameter in the range of 1.0 cm, but the user must exercise a degree of judgment in determining the number of radial nodes appropriate for the problem.

The remainder of the CARD 9.2 sets flags and options for the fuel material properties and the gap conductance model. The code contains properties for  $UO_2$  fuel derived from MATPRO-11 (Revision 1). These can be flagged by setting IMATF to zero. The correlations from MATPRO-11 use the fuel theoretical density, FTDENS(I), as a parameter in the calculation of the  $UO_2$  properties, which must be supplied by the user even if IMATF is zero. FTDENS(I) is a fractional value, that depends on the properties of  $UO_2$  pellets being modeled and is usually on the order of 0.95. Alternatively, the user may select to specify a different material for the fuel, in which case IMATF must be set to the numerical index of a material property type specified in CARD GROUP 10. For cladding material properties, the code contains properties of Zircaloy and Zircaloy oxide from MATPRO-11 (Revision 1) and these can be flagged by setting IMATC(I) and IMATOX(I) to zero. If the cladding or cladding surface oxide is some other material for the nuclear rods being modeled by geometry I, IMATC(I) and IMATOX(I) must be assigned the numerical index of an appropriate material type specified in CARD GROUP 10.

For the example shown in Figure 22.20, the input data for a nuclear fuel rod with constant gap conductance will have the following format:

There are three options available for the gap conductance model for geometry type I:

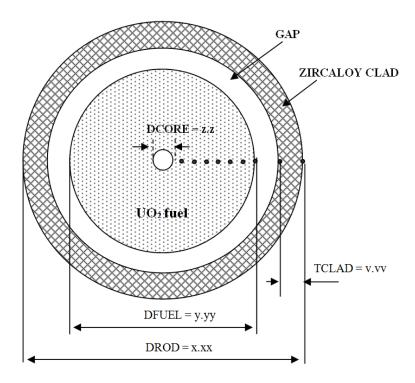


Figure 22.20: Nuclear fuel rod geometry

- 1. constant uniform gap conductance;
- 2. axially non-uniform user-specified gap conductance; and
- 3. a dynamic gap conductance model

The simplest of the three alternatives, and the easiest to implement is the constant uniform gap conductance. It is flagged by setting IGPC to zero. The gap conductance value specified on CARD 8.2 for HGAP(N) is used for rods with geometry type I and no further input is required.

The second option for a user-specified non-uniform gap conductance is a variation of the first option. The user specifies the value for gap conductance but can vary it axially and in time by means of input tables. The user specifies a table of pairs of values of axial location, AXJ(L), and gap conductance, AGFACT(L,I) on CARD 9.4. The number of pairs in the table is defined by the positive integer assigned to IGAPC on CARD 9.2. The code determines the gap conductance in each axial node of the rods with geometry type I by linear interpolation in the table specified on CARD 9.4 and applies the gap conductance temporal forcing function, if one has been specified (the temporal forcing function is specified on CARD 11.6). An example is given below:

```
Card 9.1
 NFUELT IRELF ICONF IMWR NDM5 NDM6 NDM7 NDM8 NDM9 NM10 NM11 NM12 NM13 NM14
                                    0
                    0
                         0
                               0
                                         0
                                               0
                                                    0
                                                         0
* Card 9.2: Nuclear Fuel Rod
 I FTYPE
              DROD
                      DFUEL NFUEL IMATF IMATC IMTOX DCORE
                                                               TCLAD FTDENS IGPC IGFORC IRADP
   nucl
                                       0
                                             0
                                                                       0.95
                                                                                3
                                                                                       0
                                                        z.z
              x.xx
                       у.уу
 Card 9.4: Gap conductance versus axial location
          AGFACT
                                    AXL
                                          AGFACT
    AXL
                    AXL
                          AGFACT
            u.uu
                                            u.uu
                            u.uu
```

In the dynamic gap conductance model, the code calculates the gap conductance based on the thermal and geometric properties in the gap. The model takes into account the thermal expansion of the fuel pellet and cladding and elastic deformation of the cladding in determining the size of the gap. The dimensions of the gap and the thermodynamic properties of the fill gas are used to calculate the gap conductance. The user must specify the initial cold gap width in a table of axial location, AXJ(L), versus cold gap width, AGFACT(L,I), read on CARD 9.4. The cold gap width is the gap width before the fuel pin is brought to full power. But since the gap conductance model does not calculate the effects of the power history of the rod, the cold gap width specified by input must indicate any changes from as-built conditions due to burnup-dependent factors. The number of entrees in the table read on CARD 9.4 is the absolute value of IGPC (IGPC is entered as a negative number to flag the dynamic gap conductance model).

In addition to the cold gap width, the user must supply data on composition of the fill gas and the internal characteristics of the gap for the dynamic gap conductance model. These data are read on CARD 9.3. The cold pin pressure, PGAS(I), is the gas pressure at 295 K. The gas plenum volume, VPLEN(I), is the volume of extra space between the top of the fuel pellet stack and the top of the rod. Both PGAS(I) and VPLEN(I) should be determined from the manufacturing specifications of the particular fuel rods being modeled by geometry I. The surface roughness of the fuel pellet and the cladding inner surface, ROUFF(I) and ROUFC(I), are used in the temperature jump discontinuity correlation in the gap conductance model. The correlation was optimized with values of  $0.2x10^{-4}$  inches and  $0.39x10^{-4}$  inches for ROUFC and ROUFF, respectively, and even if the user knows the correct values, the optimized values will likely yield a more accurate answer.

The fuel relocation model [6] can be included in the dynamic gap conductance calculations. Relocation allows fuel to move radially into the fuel-cladding gap. Cracks formed in the fuel by relocation reduce the effective conductivity of the fuel. When IRELF on CARD 9.1 is set to one, both radial relocation and conductivity degradation are included in the calculation. Since the relocation model is an empirical correlation, the surface roughnesses should correspond to those used to derive the correlation. ROUFC and ROUFF are  $0.45x10^{-4}$  inches and  $0.85x10^{-4}$  inches, respectively in [5]. (These values are different from the surface roughnesses used to optimize the temperature jump discontinuity correlation). Conductivity degradation can be calculated without relocation by setting ICONF = 1 and IREFL = 0. In this case the effect of radial relocation should be included in the specified cold gap width.

The code expects the fill gas to be composed of helium, xenon, argon, krypton, hydrogen and nitrogen. The composition of the gas for a given geometry type is specified by filling the GSRAC(I) array on CARD 9.3 with values for the molar fractions ( $GSFRAC \le 1$ ) of the gases present. The GSFRAC(I) value for any one component can be any value between zero and one, inclusive, but the array must sum to unity:

$$\sum_{L=1}^{6} \mathtt{GSFRAC(L)} = 1.000$$

For example, if the fill gas were 100% helium, GSFRAC(1) would be 1.00 and GSFRAC(2) through GSFRAC(6) would all be 0.00.

An example for a nuclear fuel rod with dynamic gap conductance, a gap filled only with helium, and no fuel relocation is given below:

\* Card 9.1 \* NFUELT IRELF ICONF IMWR NDM5 NDM6 NDM7 NDM8 NDM9 NM10 NM11 NM12 NM13 NM14 0 \* Card 9.2: Nuclear Fuel Rod I FTYPE DROD DFUEL NFUEL IMATE IMATC IMTOX DCORE TCLAD FTDENS IGPC IGFORC IRADP 1 nucl 7 0 0 0.95 -1 0 0 x.xу.у 7. . 7. \* Card 9.3: Fill gas composition **PGAS VPLEN** ROUFF ROUFC GSFR1 GSFR2 GSFR3 GSFR4 GSFR5 GSFR6 OXIDET d.d 1.000 0.000 0.000 0.000 0.000 0.000 b.b c.c a.a Card 9.4: Gap conductance versus axial location AXL AGFACT

The code assumes that the radial power profile in a fuel pellet is uniform, but the user has the option of specifying a non-uniform radial profile in the pellet. If this option is selected, the pin radial power profile is read on CARD 9.5 as a table of relative radial location, RODP(L), versus relative power factor, POWR(L). The relative radial location is defined as:

$$RODP(L) = \frac{2(R(L) - \frac{DCORE}{2})}{DFUEL-DCORE}$$

where DCORE is the diameter of center void (in inches); DFUEL is the pellet diameter (in inches); and R(L) is the radial location (in inches).

The relative power fraction is defined as:

$$\texttt{POWR(L)} = \frac{power a tradial location R(L)}{average pin power}$$

The profile constructed by the table of (RODP(L), POWR(L)) is automatically normalized to unity.

CARDS 9.6 and 9.7 are related to the nonnuclear rod and unheated conductor geometry types. The geometry types are classified as solid cylinders, hollow tubes or flat plates. The physical elements of a reactor vessel or test section can be modeled with these geometries, even if they do not conform exactly to the ideal shape. A guide tube, for example, can be modeled as a flat wall, even though it is a tube. In modeling a solid element as a rod or unheated conductor, the important aspects to preserve are:

- 1. the surface area in contact with the fluid;
- 2. the mass available for thermal storage;
- 3. the thickness of any element (wall or tube) that has fluid contact on both surfaces

The physical dimensions of the geometry types should be defined so that these three characteristics of the solid elements of the system are modeled with reasonable accuracy.

The nonnuclear geometry types are defined on CARD 9.6 by index number I and an alphanumeric value for FTYPE(I):

HROD for solid cylinder;

TUBE for hollow tube, and

WALL for a flat plate

The dimensions of the rods or unheated conductors modeled by a geometry type I are defined by DROD and DIN. For solid cylinder geometry, DROD is the outside diameter and DIN is zero. For a tube geometry, DROD is the outside diameter and DIN is the inside diameter. For flat plate geometry, DROD is the perimeter of the plate surface that is in contact with the fluid and DIN is the plate thickness.

In many problems, DROD and DIN will correspond exactly to the physical dimensions of the elements being modeled. However, consider a square guide tube that is being modeled. It could be represented by an equivalent conductor, such a flat plate. The guide tube, with width W and thickness t, has an outer diameter equal to 4W. A flat plate with equivalent outer perimeter would have DROD = 4W and DIN = t. (In this case the user has to remember that an unheated conductor can be connected to only one fluid subchannel. If the guide tube from the above example is connected to more than one subchannel, it must be split to several walls with corresponding dimensions. See Figure 22.18).

The number of different material regions modeled in the rods or unheated conductors of geometry I is specified by the value of NFUEL on CARD 9.6. The number of heat transfer nodes in a region is specified by input on the subsequent CARD 9.7 input. A region defines a radial ring (or layer, for walls) of uniform material properties for the conduction solution. For example, a heater—composed of an electrical heater wire made from nichrome, a boron silicate filler material, and a stainless steel cladding—consists of three different materials: nichrome; boron silicate; and stainless steel. Four separate regions of materials are formed by the rod geometry: the central core of boron silicate; the nichrome heater wire; the outer layer of boron silicate; and the stainless steel cladding. Therefore, four material regions (NFUEL = 4) are required to describe this heater rod geometry type.

The remaining input for CARD 9.6 defines the material properties used to determine the minimum film boiling temperature for the heat transfer surfaces of the geometry type. Most physical systems in contact with water, particularly at high temperatures, develop an oxide layer that significantly affects the thermal properties of the surface. The user may specify the thermal properties of the oxide material on the outside and inside surface by setting IMATOX(I) and IMATIX(I) to the index number of a material properties table specified in CARD GROUP 10. (The variable IMATIX(I) applies only to TUBE or WALL geometries with fluid contact on the inside surface. For solid cylinders (HROD) and TUBEs or WALLs with fluid contact only on the outside surface, it is ignored). If a particular geometry type does not have oxide scale on its heat transfer surfaces, IMATOX(I) and IMATIX(I) should be assigned values that correspond to the material properties table for the material the rods or unheated conductors are composed of at the surface. If IMATOX(I) and IMATIX(I) are set to zero, the oxide property index defaults to the built-in zirconium dioxide property table.

The material composition and radial noding in the NFUEL regions identified for a geometry type I are read on CARD 9.7. The user must specify a material property table index for the region MATR(L), corresponding to a properties table supplied in CARD GROUP 10. The physical thickness of each region must be specified in the TREG(L) variable, with the number of radial models in that region, NODER(L). The code automatically calculates the size of each node within the region by dividing the region into NODER(L) nodes of equal thickness.

The material temperature is calculated at the center of mass of each node. The only exception to this placement occurs for a region where one edge forms a heat transfer surface in contact with the fluid. In this

case, the node closest to the surface is given only one-half the nominal thickness. The node temperature is calculated at the surface rather than the center of the half-width node.

For the above described heater rod with four material regions (NFUEL = 4), Figure 22.21 shows a typical temperature profile in the heater rod and the radial nodding in the various region required to resolve it. Region 1, the boron silicate in the center of the rod, must of necessity be at a uniform temperature in steady state, and, since it is a relatively small region, it can be modeled adequately with a single node. The nichrome wire fills a thin enough region in the heater rod to be considered almost a line source and cannot have a significant temperature gradient (because of the thermal properties of nichrome), so this region too can be modeled with a single node. The region between the heater wire and the cladding has a very gradual slope because of the thermal properties of boron silicate, so two nodes are enough to resolve it adequately. The region defined by the steel cladding is relatively thin, but because of the thermal properties of steel, the temperature profile may have a relatively steep slope here, so two nodes are required in this region. However, because of special data handling inside the code, the following restriction holds:  $2 \le NODER(L) \le NFUEL$ .

The number of radial nodes to use within a region of geometry is governed by two factors:

- 1. the temperature gradient expected in the structure; and
- 2. the calculational cost for large number of heat transfer conditions being modeled.

Beyond a certain limit, however, doubling the number of nodes does not double the accuracy of the solution, but it nearly quadruples the computational time required for elements with that geometry type. The user must exercise some judgment as to what constitutes an adequate but not excessive number of radial nodes for the heat transfer solution in a particular problem.

The radial power factor, QREG(L), specified for each region, defines which regions of a nonnuclear rod generate heat internally. In heated rods, the QREG(L) values for all the regions of the rod are automatically normalized to unity, relative to the total power generated in a rod. Nonzero values of QREG are ignored in the unheated conductors.

The input for the heater rod shown in Figure 22.21 will be:

```
* Card 9.1
* NFUELT IRELF ICONF IMWR NDM5 NDM6 NDM7 NDM8 NDM9 NM10 NM11 NM12 NM13 NM14
             0
                              0
                                   0
                                        0
                                              0
                                                   0
                                                         0
* Card 9.6: Heater Rod
 I FTYPE
            DROD
                  DIN NFUEL IMATOX IMATIX
                                             NDUM8
                                                    NDUM9 NDUM10 NDUM11 NDUM12 NDUM13 NDUM14
  1 hrod
             x.x 0.0
                                  0
                                          0
                                                 0
                                                        0
                                                                0
                                                                       0
* Card 9.7: RegionsâĂŹ description ; L = 1, NFUEL
*NODER(L) MATR(L)
                    TREG(L) QREG(L)
    2
                1
                         r1
    2
                2
                     (r2âĂŞr1)
                                   1.0
                1
                     (r3âĂŞr2)
                                   0.0
                     (r4âĂŞr3)
    2
                3
                                   0.0
```

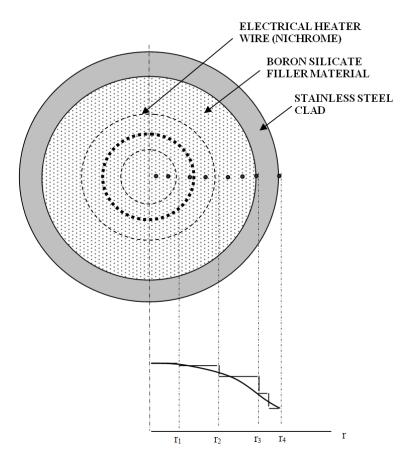


Figure 22.21: Heater rod geometry

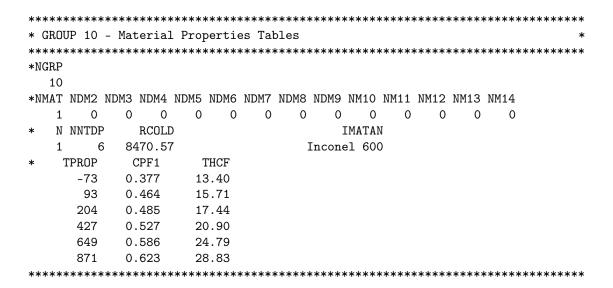
#### 22.3.3 Instructions to CARD GROUP 10

The input for CARD GROUP 10 supplies the material properties selected by numerical indices in the geometry type input on CARD GROUP 9. If all the geometry types specified in CARD GROUP 9 use the build-in nuclear fuel rod properties for  $UO_2$ , Zircaloy, and  $ZiO_2$ , the input for CARD GROUP 10 is omitted. Otherwise, properties tables must be supplied for NMAT materials, where NMAT is the number of unique indices specified in CARD GROUP 9 input for the variables IMATF, IMATC, MATR(L), IMATOX(I), and IMATIX(I).

The material properties tables are entered on CARDS 10.2 and 10.3. Both cards are read for each table. CARD 10.2 identifies the table by its numerical index and specifies the number of entrees, NNTDP, in the table. The cold-state density of the material, RCOLD(N), is also entered on this card. RCOLD(N) is the density at normal (as-built) conditions and is used to calculated the mass of the nodes. This approach eliminates the tedious process of calculating the relocation of the radial nodes due to thermal expansion during the transient.

The properties table itself is read on CARD 10.3. The material properties included in the table are specific heat, CPF(I,N), and thermal conductivity, THCF(I,N), as functions of the temperature, TPROP(I,N). The temperature range of the table must be great enough to encompass the temperature extremes expected in the calculation; otherwise the run will abort due to property lookup failure. The user may specify constant properties for a material type. This is done by entering only one element in the table; i.e., setting NNTDP to 1, and reading CARD 10.3 for only one set of CPF(I,N), THCF(I,N), and TPROP(I,N). The values specified in TPROP(I,1) is in this case is superfluous. The values specified in CPFF(I,1) and THCF(I,1) will be used for material type I regardless of the material temperature.

An example of the material properties input for a heated tube made from Inconel 600 is given below:



# 22.4 Specification of the Initial and Boundary Conditions

The next sections provide instructions to the specification of the initial and boundary conditions as well as the axial and radial power distribution. The initial operating conditions are read on CARD GROUP 1. The power distribution is prescribed on CARD GROUP 11. The boundary conditions are read on CARD GROUP 13.

#### 22.4.1 Instructions to CARD GROUP 1

The rod friction correlation, the entrainment and deposition model, the turbulent mixing and void drift models, and the pressure matrix solver to be utilized are selected via input on CARD 1.1. The number of non-condensable gases to be considered is specified by the parameter NGAS (maximum of eight). The rod friction correlation is chosen by the input flag IRFC. The modeling of entrainment and deposition at annular flow regime is activated by setting the input flag EDMOD equal to one. The turbulent mixing and void drift models are chosen by the input flag IMIX. The input flag ISOL selects the numerical solver for the system of pressure equations.

The initial operating conditions are specified in CARDS 1.2 and 1.3. The bundle (total) inlet flow rate, GTOT; the average linear heat rate, AFLUX; and the fraction of the heat rate generated directly in the coolant, DHFRAC, are read on CARD 1.2. The system pressure, PREF; the inlet fluid enthalpy, HIN (or the inlet temperature, TIN); the mass flux into the system, GIN; and the initial volume fraction of liquid in the liquid-vapor-gas mixture, VFRAC(1); and the initial volume fraction of vapor in the vapor-gas mixture, VFRAC(2) are read on CARD 1.3.

The system pressure, PREF, and enthalpy, HIN (or the enthalpy derived from the temperature: TIN), are used to determine the initial properties of the fluid. During the calculation the enthalpy and pressure change depending on the state of fluid and the local heat generation rate. The initial mass flux GIN is not, in general, compatible with the initial pressures, which are calculated based on gravity head losses only. The code must iterate to a correct solution. It is usually most expedient to specify the initial mass flux as zero; i.e., a "standing start".

If GTOT is set to non-zero value and the inlet boundary condition type (see CARD 13.4) is inlet mass flow rate and inlet enthalpy (BC type 2), the code will calculate subchannel mass flow rates according to the subchannels flow areas as specified on CARD 2.1 and will ignore the subchannel mass flow rates in CARD 13.4. If GTOT is set to zero and the inlet boundary condition type is inlet mass flow rate and inlet enthalpy (BC type 2), the user must specify subchannel mass flow rates in CARD 13.4.

If non-condensable gases are considered, their types (names) and volumetric fractions are specified on CARD 1.4 with the parameters GTYPE(I) and VFRAC(I+2), respectively. Valid non-condensable gas names include 'air', 'argo', 'heli', 'hydr', 'kryp', 'nitr', 'oxyg', and 'xeno'. Even without modeling non-condensable gases, at least one type (usually air) should be specified with negligibly small volumetric fraction. An example of CARD GROUP 1 is given below.

```
******************************
* GROUP 1 - Calculation Variables and Initial Conditions
*****************************
* NGR.
   1
* Card 1.1
 NGAS IRFC EDMD IMIX ISOL NDM6 NDM7 NDM8 NDM9 NM10 NM11 NM12 NM13 NM14
                   0
   1
           1
               2
                       0
                           0
                               0
                                   0
                                       0
       1
 Card 1.2
    GTOT
           AFLUX
                  DHFRAC
      0.
             10.
                      .0
* Card 1.3
    PREF
             HIN
                    HGIN
                          VFRAC1
                                  VFRAC2
   1000.
           485.5
                   124.0
                             1.0
                                   .9999
* Card 1.3
* GTYPE
            VFRAC
            .0001
   air
**********************************
```

#### 22.4.2 Instructions to CARD GROUP 11

The input for CARD GROUP 11 defines the steady state and transient two-dimensional power distribution. In addition, the user may specify temporal forcing functions on the total power generation rate and the nuclear fuel rods gap conductance. If no rods are specified in CARD GROUP 8 input, the input for CARD GROUP 11 is unnecessary and may be omitted entirely.

The CARD 11.1 specifies the number of transient changes of the axial profile, NQA (NQA must be equal to 1 for steady state calculations); the number of axial profile tables to be read for each transient point, NAXP (the identification numbers must correspond to the values specified in the IAXP array on CARD 8.2 and at least one table must be supplied); the maximum number of pairs of elements in any axial power profile table, MNXN (a minimum of two values, for the top and bottom of the rod, must be provided for the simplest case of an uniform power profile); the number of pairs of elements in the total power forcing function table, NQ (NQ is set to zero if total power is constant); the number of pairs of elements in the gap conductance forcing function table, NGPFF (NGPFF is set to zero if gap conductance is constant); and the number of radial profile tables, NQR (NQR corresponds to the transient points of time, for which different radial profiles will be read).

CARDS 11.2 through 11.4 are read NQA times. CARD 11.2 specifies the transient time, YQA, for which the axial power profile table will be read on the CARD 11.4. CARD 11.3 gives the identification number, I, and the number of pairs of elements, NAXN(I), in the axial power profile table I to be read in CARD 11.3. The table itself is read as NAXN(I) pairs of Y(I,N) and AXIALZ(M,I,N), where Y(I,N) is the axial location relative to the bottom and AXIALZ(M,I,N) is the power factor at that location. (Index M varies from 1 to NQR). AXIALZ(M,I,N) is defined as the ratio of the local power to average power in the rod (rods) the table is applied to. Figure 22.22 shows a nuclear fuel rod with a 3.66 m active (heated) length and a chopped cosine power profile.

The axial power table is used to interpolate linearly for axial power factors at the boundaries of the axial heat transfer cells, and the profile is re-integrated over each cell to obtain on average linear heat rate for the cell. When re-noding occurs, the axial power profile is re-integrated to obtain the average linear heat rate in the new cells. In the example shown in Figure 22.23, an electrical heater rod with step approximation to a cosine power profile in modeled in a subchannel with 0.305 m continuity nodes. The linear heat rate in the cell is modeled as a uniform  $Q_i n$  along entire length of the cell.

With subchannel splitting it is possible to extend a rod through several sections of channel geometry. A rod can be used to model an average heater rod in a test section, as illustrated in Figure 22.24. Channel 3 in Section 1 models a portion of the region penetrated by the cold ends of the rods modeled by the average heater rod 1. Channel 3 connects to Channel 10 above. Channel 10 in Section 2 models a portion of the core containing the rods modeled by Heater Rod 1. Channel 10 connects to Channels 15, 16, and 17 above, in Section 3, but the heater rod ends at the boundaries between Sections 2 and 3.

In general, a rod can pass through any number of contiguous vertical sections with fluid connections in different subchannels in each section, but the rod must begin and end at a section boundary. The vertical locations of the beginning and end of each rod must be considered when defining section boundaries in CARD GROUP 4. Unheated conductors do not require axial power profiles and may not cross section boundaries, but they must also be considered in the section boundary locations. Unheated conductors are assumed by the code to have the same axial length as the sections that contain them.

The total power forcing functions are read on CARD 11.5 in NQ pairs of transient time YQ(N) and power factor FQ(N). The power factor FQ(N) is the ratio of power at time YQ(N) and the initial power. The code interpolates linearly in the power forcing function table to determine the current value of total power at each step.

The gap conductance forcing functions are specified on CARD 11.6 as NGPFF pairs of transient time YGPFF(N) and gap conductance factor FGPFF(N). The input convention for this table follows the logic as the other tables in CARD GROUP 11. The gap conductance factor FGPFF(N) is defined as the ratio of the gap conductance

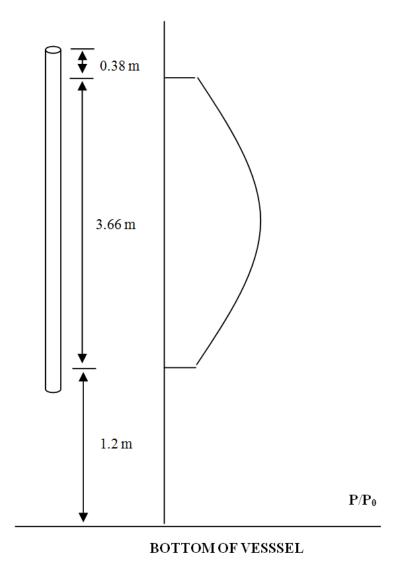


Figure 22.22: Nuclear fuel rod power profile

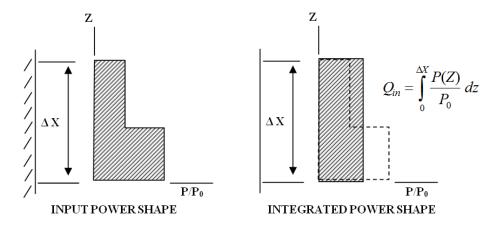


Figure 22.23: Heat input over one fluid node

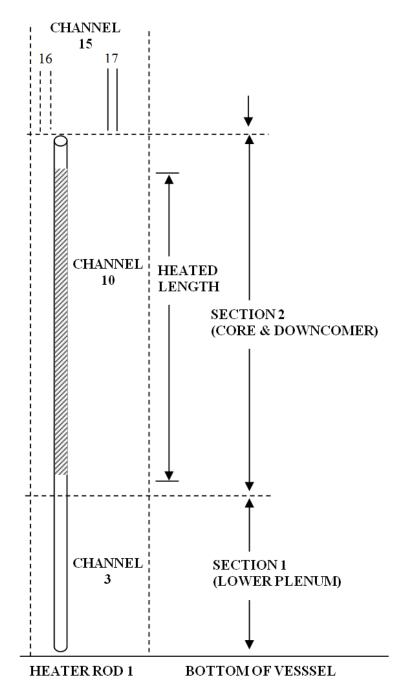


Figure 22.24: Heater rod crossing section boundaries

value at the time YGPFF(N) and the initial value. The forcing function is applied uniformly along the axial length for nuclear fuel rods with a gap conductance forcing function specified on CARD 9.2 (IGPC > 0).

The next two cards, CARDS 11.7 and 11.8, provide the radial power profile and its transient behavior. CARD 11.7 specifies the transient time, YQR, for which an array FQR(N), containing the radial power factors of rods (the local power generation in each individual rod), will be read in the CARD 11.8.

The code calculates the local linear heat rate (which is transferred through the rod surface to heat up the coolant) as follows:

$$\dot{q}(rod, x, t) = (1 - d)\dot{\bar{q}}f_{time}(t)f_{axial}(x, t)f_{radial}(rod, t)$$

where  $\dot{q}$  is the AFLUX read on CARD 1.2; d is the DHFRAC read on CARD 1.2;  $f_{time}(t)$  is the FQ read on CARD 11.5;  $f_{axial}(x,t)$  is the AXIALZ read on CARD 11.4; and  $f_{radial}(rod,t)$  is the FQR read on CARD 11.8.

An example of input for CARD GROUP 11 for the configuration shown in Figure 22.19 with a constant uniform axial and radial power distributions and transient total power is given below.

```
*************************************
* GROUP 11.0 - Axial Power Tables and Forcing Functions
**************************************
* CARD GROUP 11
* NGR
  11
* Card 11.1
* NQA NAXP MNXN
                NQ NGPF NQR NDM7 NDM8 NDM9 NM10 NM11 NM12 NM13 NM14
   1
        1
                     0
                          1
                              0
                                   0
                                       0
                                            0
                                                0
* Axial Power Forcing Functions
 Card 11.2
      YQA
      0.0
* Card 11.3
  I NAXN
       2
  1
* Card 11.4
       Υ
              AXIAL
 0.000000
           1.000000
 1.830000
           1.000000
 Total Power Forcing Functions
 Card 11.5
        YQ
                  FQ
    0.0000
              0.0000
    1.0000
              1.0000
  100.0000
              1.0000
* Radial Power Forcing Functions
 Card 11.7
      YQR
      0.0
* Card 11.8
   FQR(1)
            QR(2)
                   FQR(3)
                            FQR(4)
                                    FQR(5)
                                            FQR(6)
                                                    FQR(7)
                                                            FQR(8)
           1.0000
                   1.0000
                            1.0000
                                    1.0000
                                            1.0000
                                                    1.0000
                                                            1.0000
   1.0000
   FQR(9)
   1.0000
```

### 22.4.3 Instructions to CARD GROUP 13

Two main types of boundary conditions can be specified by the input in CARD GROUP 13. These are:

- 1. inlet or outlet boundary conditions on subchannels, and
- 2. specific boundary conditions on particular cells within the mesh for both the vertical and transverse control volumes

Inlet or outlet boundary conditions must be specified for subchannels that do not have vertical connections to other subchannels in the system at their inlet or outlet. This, obviously, must include the inlet of

subchannels in the first axial section and the outlet of subchannels in the last axial section. But other subchannels within the stack of sections can be unconnected at the top or bottom to other subchannels. Mesh boundary conditions may be applied wherever needed, on any face of any cell within the vessel mesh. But only one boundary condition can be applied to any one node.

The number of subchannel mesh cell boundary conditions, including inlet, outlet, and internal mesh cells, in a given problem is specified by the value of NIBND on CARD 13.1. The NIBND boundary conditions are set by reading CARD 13.4 for each boundary, specifying the subchannel number, IBOUND(1,M); axial node number, IBOUND(2,M); boundary type number, ISPEC(M); the first boundary value, BCVALUE1; the second boundary value, BCVALUE2; and the third boundary value, BCVALUE3. The forcing functions for a given boundary condition type are identified by table number in parameters N1FN, N2FN, and N3FN for the first, second and third boundary values, respectively. The boundary conditions that can be specified for subchannel mesh cells are listed in Table 7.

The general application of the boundary conditions is summarized in Table 8.

If boundary condition type 1 is selected, the pressure in subchannel I = IBOUND(1,M) at node J = IBOUND(2,M) is set to BCVALUE3 and the enthalpy is set to BCVALUE2. BCVALUE1 is not used. For boundary condition type 2, the flow at the top face of node IBOUND(2,M) in the subchannel IBOUND(1,M) is set to BCVALUE1 and the enthalpy is set to BCVALUE2. BCVALUE3 is not used. The flow may be positive or negative. For boundary condition type 3, the flow is set to BCVALUE1. BCVALUE2 and BCVALUE3 are not used. This boundary condition is used to shut off vertical flows within the mesh or to specify flow into or out of the top of a subchannel, where it is not desirable to specify the node enthalpy, as is done for boundary condition type 2. Boundary condition type 4 is used to specify a mass source in any cell. BCVALUE1 is set to the mass flow rate, BCVALUE2 is set to the enthalpy, and BCVALUE3 is set to pressure. Boundary condition type 5 is used to specify a pressure sink connected to any mesh cell. BCVALUE3 is the pressure sink, and BCVALUE2 is the enthalpy of the fluid in the sink. BCVALUE1 is not used.

Tab	le 7:	Bound	lary	Cond	lition	Types
-----	-------	-------	------	------	--------	-------

E	Boundary condition type	Internal mesh cells	Boundary cells	Input in CARDs
1	Pressure and enthalpy	-	X	13.4, 13.6, 13.7
2	Mass flow rate and enthalpy	-	X	13.4, 13.6, 13.7
3	Mass flow rate	X	-	13.4
4	Mass source	X	-	13.4, 13.5, 13.6, 13.7, 13.8
5	Pressure sink	X	-	13.4, 13.6, 13.7, 13.9
Ŀ	Crossflow set to zero	X	-	13.10

In addition to the axial boundary conditions, the transverse flow between subchannels connected by a gap can be shut off at any elevation within the vessel by specifying input for CARD 13.7. This input is read if parameter NKBND on CARD 13.1 is greater than 0. The input for CARD 13.7 consists simply of the gap identification number K and the contiguous axial levels JSTART to JEND where the crossflow will be set to zero. This format makes it very easy to zero out a sequence of crossflows, from J = 2 to J = 15, for example.

If flow is to be shut off in more than one segment of a gap, CARD 13.7 input can be specified for gap K more than once, until all of the axial levels to be zeros have been specified.

Unless otherwise indicated, the special boundary conditions applied to cells as specified in CARD GROUP 13 are constant in time. The user may, however, specify forcing functions on BCVALUE1, BCVALUE2, and BCVALUE3 for any or all of the special boundary conditions. The number of forcing functions is specified by the value of NFUNCT on CARD 13.1.

The forcing functions on the special boundary conditions are specified in the same way as any other forcing function in the vessel input—as tables of forcing function factors versus time. The number of elements in a given table is read in CARD 11.2 as NPTS; then NPTS pairs of ABSCIS(K,I) and ORDINT(K,I) are read on CARD 13.3. The ABSCIS array contains the transient time, in seconds, when the corresponding forcing function factor ORDINT is to be applied. ORDINT is defined as the ratio of the value at time ABSCIS(K,I) of the parameter being forced and its initial value. The factor is determined at any given time step by linear interpolation in the forcing function table. The forcing function tables are numbered sequentially from 1 to NFUNCT in the order they are read in.

In the code, the initial mass flow rate is specified as zero in the whole flow domain by default. That means a "standing start" is performed. (Otherwise there could be severe problems with convergence). For this reason the inlet mass flow rate is also highly recommended to start as zero and to be run up by means of a ramp which can be specified by a forcing function (see variable N1FN). This concerns boundary condition types 2, 3, and 4.

If a mass source boundary condition (type 4) has been specified for a cell on CARD 13.4, additional information must be supplied on CARD 13.5 specifying the droplet diameter DROPS(N) and droplet mass flow rate FDROPRS(N) at the injection boundary. The index number of the forcing function table, NDFN(N), by which the droplet diameter will vary and the index number of the forcing function table, NDFFN(N), by which the droplet mass flow rate will vary must be provided as well.

For all types of boundary conditions except for type 3 the enthalpy of the non-condensable gases mixture, HMGA(N), and their volumetric fractions, GVALUE(NGA,N), have to be specified on CARD 13.6. The corresponding forcing functions NHMFN(N) and NGFN(NGA,N) are read on CARD 13.7. CARDS 13.6 and 13.7 are not needed and can be omitted by setting INITGAS on CARD 13.4 to 1, when the inlet conditions for the non-condensable gases are considered equal to the initial conditions entered on CARDS 1.3 and 1.4 (HMGA = HGIN; GVALUE = VFAC; NHMFN = 0; and NGFN = 0).

If a pressure sink boundary condition (type 5) has been specified for a cell on CARD 13.4, additional information describing the sink geometry must be supplied on CARD 13.9. Flow in or out of the sink is defined for a special control volume, as shown in Figure 22.25. The flow area of the control volume is ASINK and the length of the cell is DXSINK. The sink boundary condition usually models a real hole in the system and in this case ASINK is the area of the hole. The control volume length should be on the order of the centroid length of the channel in which the sink lies. The flow in and out of the sink generally encounters some unrecoverable pressure loss due to expansion at the orifice and the loss coefficient associated with the sink flow is defined by SINKK on CARD 13.9. The type 5 boundary conditions are numbered sequentially within the code in the order they are read in on CARD 13.4. CARD 13.9 should be read for the sink in the same sequential order. CARD 13.8 is read for the flow area of mass sources in the same manner.

An example of input for CARD GROUP 13 for the configuration shown in Figure 22.19 at steady state conditions is given below.

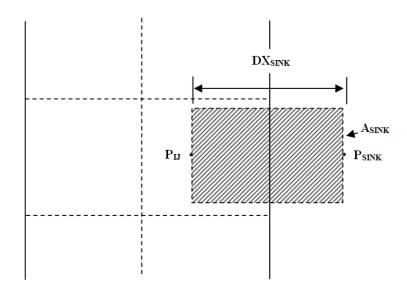
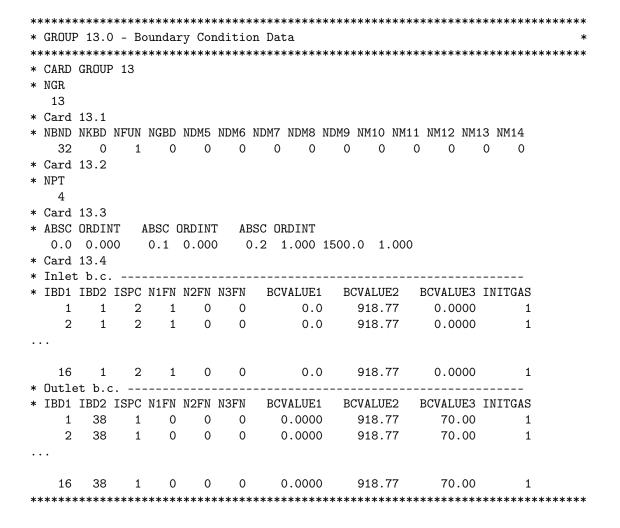


Figure 22.25: Control volume for pressure sink boundary conditions



March 24, 2019

 Table 8: General Application of the Boundary Conditions - Summary

BC type	BCVALUE1	BCVALUE2	BCVALUE3	DROPS	FDROPS	HMGA	GVALUE (1:NGA)	AINJT	ASINK/SINKK/DXSINK
1	0.0	enthalpy (of flow into the system across this boundary)	pressure	-	-	enthalpy (of non-condensible gas mixture)	volume fractions	-	-
2	mass flow rate	enthalpy	0.0	-	-	enthalpy (of non-condensible gas mixture)	volume fractions	-	-
3	mass flow rate	0.0	0.0	-	-	-	-	-	-
4	mass flow rate (of the mass source)	enthalpy (of the mass source)	pressure (of the mass source)	diameter (of droplets	mass flow rate (of droplets)	enthalpy (of non-condensible gas mixture)	volume fractions	flow area (of the mass injection	-
5	0.0	enthalpy (of the fluid in the sink)	pressure (in the sink)	-	-	enthalpy (of non-condensible gas mixture)	volume fractions	-	flow area, pressure, loss coeff., length of control vol. (of the pressure sink)
crossflow set to zero	-	-	-	-	-	-	-	-	-
forcing function identifier	N1FN	N2FN	N3FN	NDFN	NDFFN	NHMFN	NGFN (on 1:NGA)	-	-

## 22.4.4 Boron Tracking and Precipitation Modeling

### 22.4.4.1 Application of the Boron Tracking/Precipitation Model

The boron tracking model is controlled in the CTF input deck along with the rest of the CTF user-defined parameters. All of the parameters related with the boron tracking and precipitation model are conditioned to the variable IBTM (defined in CARD 1.1), the flag of the model. The user can choose between the First Order Accurate Upwind Scheme and the Second Order Accurate Modified Godunov Scheme (explained in Section A).

Initial conditions of the boron are defined in CARD 1.3 where the initial boron concentration (BRIN) of the system is uniformly defined, and the boron physical diffusion coefficient (RDIF). The boron physical diffusion coefficient is only used when is applied the Second Order Accurate Modified Godunov Scheme (IBTM = 2). The suggested values are: 0.0 (IBTM = 1) and 1.0 (IBTM = 2), due to RDIF is the result of experimental data.

The boron tracking model boundary conditions are defined in CARD 13.11. As with other CTF boundary conditions, forcing functions may be defined for the inlet boron concentration (CARDs 13.2 and 13.3) and the total number of vertical mesh cell boundary conditions, number of boron inlet boundary conditions (NIBNDB, in CARD 13.1); this value may be not the same as the number of inlet boundary conditions entered in CARD 13.4. This allows the user to control the inlet boron with respect to time as the CTF-modeled transient progresses.

CARD 13.11 follows the same scheme as CARD 13.4: IBOUNDB(1,N), index number of channel at which boundary condition N is applied; IBOUNDB(2,N), vertical node number at which boundary condition N is applied; N4FNB(N), index number of the forcing function table by which the forth parameter of the boundary condition (BCVALUE4B) will be varied; and BCVALUE4B(N), forth boundary condition represents the boron concentration [ppm]; where N=1:NIBNDB.

There must be a match between where the inlet boundary conditions and the boron inlet boundary conditions are applied, (IBOUND(L,N), L=1:2) and (IBOUNDB(L,N), L=1:2).

The boron tracking/precipitation model is applied when ISPEC(N) = 1, 2or3, what that means, except when a source (ISPEC = 4) or a sink (ISPEC = 5) is the boundary condition. The boron inlet boundary condition, BCVALUE4B, and its associated forcing function, N4FNB; are reassigned to the variables BCVALUE4(N) and N4FN(N), N=1:NIBND, as an extra boundary condition, based on the subchannel and axial level where the boron inlet boundary conditions are applied.

#### 22.4.4.2 Example of Card Group 1 and Card Group 13 (Boron Tracking/Precipitation Model)

```
* GROUP 1 - Calculation Variables and Initial Conditions
************************************
* NGR
  1
* Card 1.1
* NGAS IRFC EDMD IMIX ISOL GINT NTRN MESH MAPS IPRO MFLX IBTM NM13 NM14
 1 1 1 2 0 0 0 0 0 0 2 0
* Card 1.2
 GTOT
        AFLUX DHFRAC
          10. .0
     0.
* Card 1.3
 PREF
          HIN HGIN VFRAC1 VFRAC2 BRIN
                                               RDTF
   1000. 485.5 124.0 1.0 .9999 0.0
                                              1.0
* Card 1.3
* GTYPE
           VFRAC
           .0001
air
* GROUP 13.0 - Boundary Condition Data
************************************
* CARD GROUP 13
* NGR
 13
* Card 13.1
* NBND NKBD NFUN NGBD NBNDB NDM6 NDM7 NDM8 NDM9 NM10 NM11 NM12 NM13 NM14
  32 0 1 0 16 0 0 0 0 0 0
* Card 13.2
* NPT
* Card 13.3
* ABSC ORDINT ABSC ORDINT ABSC ORDINT
 0.0 0.000 0.1 0.000 0.2 1.000 1500.0 1.000
* Card 13.4
* Inlet b.c. -----
* IBD1 IBD2 ISPC N1FN N2FN N3FN BCVALUE1 BCVALUE2 BCVALUE3 INITGAS

    1
    1
    2
    1
    0
    0
    0.0
    918.77
    0.0000
    1

    2
    1
    2
    1
    0
    0
    0.0
    918.77
    0.0000
    1

      1 2 1 0 0 0.0 918.77 0.0000
* Outlet b.c. -----
* IBD1 IBD2 ISPC N1FN N2FN N3FN BCVALUE1 BCVALUE2 BCVALUE3 INITGAS
   1 38 1 0 0 0 0.0000 918.77 70.00 1
2 38 1 0 0 0 0.0000 918.77 70.00 1
  16 38
          1 0 0 0 0.0000
                                918.77 70.00
* Card 13.11
* Boron (Inlet) b.c. -----
* IBD1 IBD2 N4FN BCVALUE4
   1 1 1 2000.0
   2
               2000.0
  16 1 1 2000.0
************************************
```

## 22.5 Turbulent Mixing and Void Drift Modeling

CARD GROUP 12 provides input for the turbulent mixing and void drift model as specified by IMIX on CARD 1.1. This input is optional and should be used only for problems where is a significant effect of these phenomena on the behavior of the system.

In the subchannel codes, the exchange of momentum, energy, and mass due to turbulence, or the so-called turbulent diffusion or turbulent mixing, is commonly modeled in analogy to the molecular diffusion under the assumption of a linear dependence between the exchange rate of the given quantity and its gradient in the medium. The proportionality coefficients are called turbulent diffusion coefficients. Unlike molecular diffusion coefficients, which are material dependent, the turbulent diffusion coefficients depend only on the location in the flow domain. The turbulent kinematical viscosity and turbulent temperature diffusivity are in the same order of magnitude (turbulent Prandtl number approaches unity). The latter allows applying the same turbulent diffusion coefficient to all momentum, mass, and energy exchanges:  $C_t = D_t = v_t = a_t$ . In the case of a gradient in the y direction, the aforementioned assumption takes the form of:

Turbulent mixing of mass:

$$\dot{m}_k = -\rho D_t \frac{dc}{dy} A = -C_t \frac{d(\alpha_k \rho_k)}{dy} A$$

Turbulent mixing of momentum:

$$\dot{I}_k = -\rho \nu_t \frac{dU}{dy} A = -C_t \frac{d(\alpha_k \rho_k U_k)}{dy} A = -C_t \frac{dG_k}{dy} A$$

Turbulent mixing of energy:

$$\dot{Q}_k = -\rho c_p \alpha_t \frac{dT}{dy} A = -C_t \frac{d(\alpha_k \rho_k c_{p,k} T_k)}{dy} A = -C_t \frac{d(\alpha_k \rho_k h_k)}{dy} A$$

Here, the index k stands for the given field (liquid, vapor, and droplets);  $D_t$  is the turbulent diffusion coefficient for mass transfer;  $\nu_t$  is the turbulent kinematical viscosity;  $\alpha_t$  is the turbulent temperature conductivity; C is the concentration;  $c_p$  is the specific heat capacity; A is the area relevant for lateral exchange;  $\alpha$ ,  $\rho$ , U, and h are, respectively, the volume fraction of given field, density, vertical velocity, and enthalpy.

In the subchannel analyses, very often the ratio  $C_t/dy$  is substituted with the ratio of the turbulent kinematic viscosity  $\epsilon$  and the mixing length l,  $\epsilon/l$ , and the mixing length is commonly approximated as the centroid distance between the adjacent subchannels. Regarding turbulent diffusion coefficients, a dimensionless parameter can be defined:

$$\beta = \frac{C_t}{\Delta y \bar{U}}$$

where  $\bar{U}$  is the area-averaged veritical velocity of the adjacent subchannels and is defined as:

$$\bar{U} = \frac{A_i U_i + A_j U_j}{A_i + A_j}$$

Using the definition of the turbulent mixing coefficient, the exchange rate of mass, momentum and energy (Equations -35— through -37—) can be written as:

Turbulent mixing of mass:

$$\dot{m}_k = -\beta \frac{\bar{G}}{\rho} \Delta(\alpha_k \rho_k) A$$

Turbulent mixing of momentum:

$$\dot{I}_k = -\beta \frac{\bar{G}}{\rho} \Delta G_k A$$

Turbulent mixing of energy:

$$\dot{Q}_k = -\beta \frac{\bar{G}}{\rho} \Delta(\alpha_k \rho_k h_k) A$$

where:

$$\bar{G} = \frac{A_i G_{tot,i} + A_j G_{tot,j}}{A_i + A_j}$$

As concluded from Equation -39-, the turbulent mixing coefficient is a function of the particular geometry and the flow conditions. Under single-phase flow conditions, it is usually correlated to the flow Reynolds number, subchannel hydraulic diameter, heated rod diameter, gap width, and the centroid between the adjacent subchannels:  $\beta_{SP} = f(Re, d_{hyd}, d_{gap}, drod, \Delta Y)$ .

It is experimentally observed that in a two-phase flow the turbulent mixing is much higher than in a single-phase flow. Most often, the dependence of the mixing rate on the flow regime is modeled by the Beus correlation -(Beus, S.G., 1970). The two-phase turbulent velocity is assumed to be a function of the single-phase turbulent velocity:  $(\frac{\epsilon}{l})_{TP} = \theta_{TP}(\frac{\epsilon}{l})_{SP}$ , where the "two-phase multiplier",  $\theta_{TP}$ , depends on the quality. The approach by Faya (Faya, A.J.G., 1979) has been adopted in the subchannel analysis codes. Faya has slightly modified the Beus approach by applying the two-phase multiplier  $\theta_{TP}$  to the single-phase mixing coefficient:

$$\beta_{TP} = \theta_{TP}\beta_{SP}$$

where  $\theta_{TP} = f(x)$ .

The mixing rate, and hence the turbulent velocity, reaches its maximum at the slug-annular regime transition point. According to the model of Wallis (Wallis, G.B., 1969), this point can be obtained by an expression for the corresponding quality:

$$x_{max} = \frac{\frac{0.4\sqrt{g\rho_{liq}(\rho_{liq} - \rho_{vap})d_{hyd}}}{G_{tot}}}{\sqrt{\frac{\rho_{liq}}{\rho_{vap}} + 0.6}}$$

The function for  $\theta_{TP}$  is assumed to increase linearly for  $x \leq x_{max}$  and to decrease hyperbolically for  $x > x_{max}$ .

$$\theta_{TP} = \begin{cases} 1 + (\theta_{max} - 1) \frac{x}{x_{max}}, & \text{if } x \le x_{max} \\ 1 + (\theta_{max} - 1) \frac{x_{max} - x_0}{x - x_0}, & \text{if } x > x_{max} \end{cases}$$

where:

$$Re = \frac{G_{tot}d_{hyd}}{\mu_{mix}}$$

and:

$$\mu_{mix} = (1 - \alpha_{vap})\mu_{liq} + \alpha_{vap}\mu vap$$

The parameter  $\theta_{max}$ , which is the maximum of the ratio  $\frac{\beta TP}{betaSP}$ , is treated as a constant and can be estimated experimentally.

In the COBRA/TRAC code version, only a single-phase mixing coefficient (single-phase liquid for void fractions below a value of **0.6**, single-phase vapor for void fractions above a value of **0.8**, and a ramp between the two) has been modeled by means of the traditional mixing coefficient approach. Later, in the FLECHT SEASET code version, a void drift model based on the work of Lahey and Kelly has been employed. Void drift was only assumed to occur when the liquid is continuous phase and its modeling has been not applied to the hot wall flow regimes.

The current turbulent mixing and void drift models assume that the net two-phase mixing (including void drift) is proportional to the non-equilibrium void fraction gradient. At an annular film flow regime a void drift offset is assumed and only the turbulent mixing of vapor and entrained droplets is modeled. In other words, the void drift is only modeled in bubbly, slug, and churn flow, where liquid is the continuous phase and vapor is the dispersed phase.

The lateral exchange due to turbulent mixing is modeled as follows:

Turbulent mixing of mass in phase k:

$$\dot{m}_k^{TM} = -\beta \frac{\bar{G}}{\rho} \Delta(\alpha_k \rho_k) A$$

Turbulent mixing of momentum in phase k:

$$\dot{I}_k^{TM} = -\beta \frac{\bar{G}}{\bar{\rho}} \Delta G_k A$$

Turbulent mixing of energy in phase k:

$$\dot{Q}_k^{TM} = -\beta \frac{\bar{G}}{\rho} \Delta(\alpha_k \rho_k h_k) A$$

In Equations -46- through -48-,  $\beta=\theta_{TP}\beta_{SP}$  is the two-phase turbulent mixing coefficient.

Currently the single phase mixing coefficient  $\beta_{SP}$  may be either specified as a single input value or internally calculated with an empirical correlation by Rogers and Rosehart (Rogers, J.T. and Rosehart, R.G., 1972):

$$\beta_{SP} = \frac{1}{2}0.0058 \left(\frac{d_{gap}}{d_{rod}}\right)^{-1.46} Re^{-0.1} \left[1 + \left(\frac{d_{hyd,j}}{d_{hyd,i}}\right)\right]^{1.5} \frac{d_{hyd,i}}{d_{rod}}$$

The two-phase multiplier  $\beta_{TP}$  is calculated using the Beus approach for two-phase turbulent mixing.

The lateral exchange due to void drift is modeled as follows:

Mass exchange in phase k by void drift:

$$\dot{m}_{k}^{VD} = -\beta \frac{\bar{G}}{\rho} (\alpha_{k,j,EQ} \rho_{k,j,EQ} - \alpha_{k,i,EQ} \rho_{k,i,EQ}) A$$

Momentum exchange in phase k by void drift:

$$\dot{I}_{k}^{VD} = -\beta \frac{\bar{G}}{\bar{\rho}} (G_{k,j,EQ} - G_{k,i,EQ}) A$$

Energy exchange in phase k by void drift:

$$\dot{Q}_{k}^{VD} = -\beta \frac{\bar{G}}{\rho} (\alpha_{k,j,EQ} \rho_{k,j,EQ} h_{k,j,EQ} - \alpha_{k,i,EQ} \rho_{k,i,EQ} h_{k,i,EQ}) A$$

According to Levy (Levy, S., 1963) the equilibrium density distribution is related to the mass flux distribution by the expression:

$$\rho_{mix,EQ} = \alpha_{liq}\rho_{liq} + \alpha_{vap}\rho_{vap} = \alpha G_{tot,EQ} + b$$

For phase k this yields:

$$(\alpha_{k,j,EQ}\rho_{k,j,EQ} - \alpha_{k,i,EQ}\rho_{k,i,EQ}) = f_{sign} \frac{\alpha_{vap}\rho_k}{\bar{G}_{tot}} \left( G_{tot,j,EQ} - G_{tot,i,EQ} \right)$$

where:

$$f_{sign} = \begin{cases} 1 & \text{for vapor} \\ -1 & \text{for liquid} \end{cases}$$

Other parameters are:

$$\bar{G}_{tot} = \frac{A_i G_{tot,i} + A_j G_{tot,j}}{A_i + A_i}$$

$$\alpha_{vap}\bar{\rho}_{vap} = \frac{A_i \alpha_{vap,i} \rho_{vap,i} + A_j \alpha_{vap,j} \rho_{vap,j}}{A_i + A_j}$$

$$\alpha_{vap}^{-}\rho_{liq} = \frac{A_i \alpha_{vap,i} \rho_{liq,i} + A_j \alpha_{vap,j} \rho_{liq,j}}{A_i + A_j}$$

The equilibrium vapor phase distribution is given by:

$$(\alpha_{vap,j,EQ}\rho_{vap,j,EQ} - \alpha_{vap,i,EQ}\rho_{vap,i,EQ}) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} (\alpha_{vap,i}\rho_{vap,i} + \alpha_{vap,j}\rho_{vap,j})$$

and the equilibrium liquid phase distribution is given by:

$$(\alpha_{liq,j,EQ}\rho_{liq,j,EQ} - \alpha_{liq,i,EQ}\rho_{liq,i,EQ}) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} (\alpha_{vap,i}\rho_{liq,i} + \alpha_{vap,j}\rho_{liq,j})$$

The corresponding equilibrium momentum flux distribution for the vapor phase becomes:

$$(G_{vap,j,EQ} - G_{vap,i,EQ}) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} \left(\alpha_{vap,i}\rho_{vap,i}U_{vap,i} + \alpha_{vap,j}\rho_{vap,j}U_{vap,j}\right) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} \left(G_{vap,i} + G_{vap,j}\right) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} \left(G_{vap,i} + G_{vap,j}\right) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} \left(G_{vap,i} + G_{vap,j}\right) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} \left(G_{vap,i} + G_{vap,j}\right) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} \left(G_{vap,i} + G_{vap,j}\right) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} \left(G_{vap,i} + G_{vap,j}\right) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} \left(G_{vap,i} + G_{vap,j}\right) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} \left(G_{vap,i} + G_{vap,j}\right) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} \left(G_{vap,i} + G_{vap,j}\right) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} \left(G_{vap,i} + G_{vap,j}\right) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} \left(G_{vap,i} + G_{vap,j}\right) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} \left(G_{vap,i} + G_{vap,j}\right) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} \left(G_{vap,i} + G_{vap,i}\right) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} \left(G_{vap,i} + G_{vap,i}\right) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} \left(G_{vap,i} + G_{vap,i}\right) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} \left(G_{vap,i} + G_{tot,i}\right) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} \left(G_{vap,i} + G_{tot,i}\right) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} \left(G_{vap,i} + G_{tot,i}\right) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} \left(G_{vap,i} + G_{tot,i}\right) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} \left(G_{vap,i} + G_{tot,i}\right) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} \left(G_{tot,j} - G_{tot,i}\right) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} \left(G_{tot,j} - G_{tot,i}\right) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} \left(G_{tot,j} - G_{tot,i}\right) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} \left(G_$$

and for the liquid phase:

$$(G_{liq,j,EQ} - G_{liq,i,EQ}) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} (\alpha_{vap,i}\rho_{liq,i}U_{liq,i} + \alpha_{vap,j}\rho_{liq,j}U_{liq,j}) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} (G_{liq,i} + G_{liq,j})$$

The corresponding equilibrium energy distribution for the vapor phase becomes:

$$(\alpha_{vap,j,EQ}\rho_{vap,j,EQ}h_{vap,j,EQ} - \alpha_{vap,i,EQ}\rho_{vap,i,EQ}h_{vap,i,EQ}) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} (\alpha_{vap,i}\rho_{vap,i}h_{vap,i} + \alpha_{vap,j}\rho_{vap,j}h_{vap,j})$$

and for the liquid phase:

$$(\alpha_{liq,j,EQ}\rho_{liq,j,EQ}h_{liq,j,EQ} - \alpha_{liq,i,EQ}\rho_{liq,i,EQ}h_{liq,i,EQ}) = K_M \frac{G_{tot,j} - G_{tot,i}}{G_{tot,j} + G_{tot,i}} (\alpha_{liq,i}\rho_{liq,i}h_{liq,i} + \alpha_{liq,j}\rho_{liq,j}h_{liq,j})$$

#### 22.5.1 Instructions to CARD GROUP 12

The choice of the turbulent mixing and void drift modeling is controlled by the parameter IMIX specified on CARD 1.1. If IMIX is set to zero, neither turbulent mixing nor void drift are modeled and CARD GROUP 12 is omitted.

If IMIX is set to one (1), a constant mixing coefficient BETA (used at both single- and two-phase conditions) and the equilibrium distribution weighting factor AAAK ( $K_M$  in the above equations) are specified on CARD 12.1. A suggested value of AAAK is 1.4 according to Kelly, et al. (1981). An example is given below.

www.mne.psu.edu/rdfmg

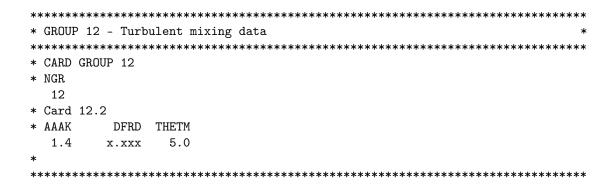
If IMIX is set to two (2), the single-phase mixing coefficient will be calculated with the correlation by Rogers and Rosehart and the two-phase multiplier will be calculated using the Beus approach. The input on CARD 12.2 includes the equilibrium distribution weighting factor AAAK; the outside rod diameter DFROD ( $d_{rod}$  in Equation -36-); and the ratio between the maximum two-phase turbulent mixing and the single-phase liquid mixing THETM ( $\theta_{max}$  in Equations -33- and -34-). The value of the outside rod diameter, DFROD, should be consistent with DROD read on CARD 9.2 or CARD 9.6. A suggested value of THETM is 5.0 according to Sato (1992). An example is given below.

#### 22.5.2 Instructions to CARD GROUP 12

The choice of the turbulent mixing and void drift modeling is controlled by the parameter IMIX specified on CARD 1.1. If IMIX is set to zero, neither turbulent mixing nor void drift are modeled and CARD GROUP 12 is omitted.

If IMIX is set to one (1), a constant mixing coefficient BETA (used at both single- and two-phase conditions) and the equilibrium distribution weighting factor AAAK ( $K_M$  in the above equations) are specified on CARD 12.1. A suggested value of AAAK is 1.4 according to Kelly, et al. (1981). An example is given below.

If IMIX is set to two (2), the single-phase mixing coefficient will be calculated with the correlation by Rogers and Rosehart and the two-phase multiplier will be calculated using the Beus approach. The input on CARD 12.2 includes the equilibrium distribution weighting factor AAAK; the outside rod diameter DFROD ( $d_{rod}$  in Equation -36-); and the ratio between the maximum two-phase turbulent mixing and the single-phase liquid mixing THETM ( $\theta_{max}$  in Equations -33- and -34-). The value of the outside rod diameter, DFROD, should be consistent with DROD read on CARD 9.2 or CARD 9.6. A suggested value of THETM is 5.0 according to Sato (1992). An example is given below.



# 22.6 Results Reporting

The output information to be printed out can be selected by input on the legacy CARD GROUP 14. In total, ten output files can be generated. A short description of the CTF output files (produced by legacy Card Group 14) is provided in Table 9. Guidance to CARD GROUP 14 is given in the following section.

 Table 9: List of Generated Output Files

File Name	Description	SI Units	US Units
deck.cdm	A dump file for restarting the calculation; generated if $\mathtt{DUMPF} = 1$ in CARD INPUT .2; empty if $\mathtt{DUMPF} = 0$	X	X
deck.crs	An empty restart file; it must be overwritten by deck.cdm before a restart calculation is executed	X	X
deck.out	Input repitition, geometry information Subchannel, gap, rod, and unheated conductor results	-	X
deck.run	Convergence history	-	X
dnb.out	DNB information	-	X
heat.time	Global transient heat balance	X	X
krysolv.out	Debug output for Krylov solver convergence. Generated if the flag IOPT read on CARD 14.1 is set to 2.	-	-
massflow.time	Global transient mass balance	X	X
mixture_temp.out	Average coolant temperature in node (i,j) (simplified output needed for validation purposes). Generated if the flag IOPT read on CARD 14.1 is set to 3.	X	-
results_channel.out	Subchannel results	X	-
results_gap.out	Gap results	X	-
time.out	Debug output for inner/outer iterations information. Generated if the flag IOPT read on CARD 14.1 is set to 2.	-	-
void.out	Subchannel void fraction (simplified output needed for uncertainty analyses). Generated if the flag IOPT read on CARD 14.1 is set to 4.	-	-

## 22.6.1 Instructions to CARD GROUP 14 (Legacy)

The user can request a complete printout of the calculated results by setting the parameter N1 on CARD 14.1 equal to 5 and the parameters NOUT1, NOUT2, NOUT3 and NOUT4 equal to zero. In this case results for all

subchannels, gaps, heated and unheated conductors will be printed out in the corresponding output files. If the user needs output information for selected subchannels, gaps, or conductors only, that can be specified by different combinations of values for N1, NOUT1, NOUT2, NOUT3 and NOUT4 and additional input on CARDS 4.2 through 4.5.

The property table will be printed in the deck.out file if the parameter IPROPP on CARD 4.1 is set to 1. The parameter IOPT controls the printout of additional debug information.

For the example given below, the output will contain results for subchannels 5, 13, 17, and 20; for gaps 9, 25, and 36; for heated conductors 4, 10, and 12; and for unheated conductors 5 and 8. In addition, the subchannel void fractions (for all subchannels) will be provided in the output file void.out.

```
* Group 14 - Output Options
*NGRP
   14
* Card 4.1
  N1 NOUT1 NOUT2 NOUT3 NOUT4 IPROPP IOPT NDM8 NDM9 NM10 NM11 NM12 NM13 NM14
                3
                      3
                             2
                                    0
                                              0
                                                    0
                                                         0
                                                              0
                                                                   0
                                                                        0
* Card 4.2: Subchannels to be printed
*PRCH
    5 13
            17
                 20
* Card 4.3: Gaps to be printed
*PRTG
    9 25
            36
* Card 4.4: Heated conductors to be printed
*PRTR
    4 10
            12
* Card 4.5: Unheated conductors to be printed
*PRTS
    5
```

## 22.7 Main Problem Control and Time Domain Data

A brief description of the required input for the main problem control data and time domain data is provided in the following sub-sections.

#### 22.7.1 Instructions to Input of Main Problem Control Data

The main problem control data is read on CARDS INPUT 1 through 3 and CARD COBRA 1. CARD INPUT 1 selects the units in input and output files (SI or US). CARD INPUT 2 specifies if an initial or restart calculation will be performed and whether a restart file, deck.cdm, will be generated. The inner and outer iterations limits are read on CARD INPUT 4. CARD COBRA 1 gives alphanumeric information to identify the simulation.

In the example of input for main problem control data given below, the input deck will be written in US units, but the output decks will be in SI units (ICOBRA = 0); it will be an initial run (INITIAL = 1) and

a restart file will not be generated (DUMPF = 0); the outer iteration convergence criterion (EPSO) is set to  $10^{-3}$ ; the maximum number of outer iterations (DITMAX) is equal to 5; and the maximum number of inner iterations (IITMAX) is equal to 40.

```
* MAIN PROBLEM CONTROL DATA
 CARD INPUT 1
       ICOBRA
* CARD INPUT 2
      INITIAL
                    DUMPF
                       0
 CARD INPUT 3
        EPSO
                   OITMAX
                               IITMAX
       0.001
                                   40
* CARD COBRA 1
*<---->
         **2sub-channels**
```

## 22.7.2 Instructions to Input of Time Domain Data

After all component data have been entered, the user must define the time domain for the simulation. The total time can be divided into several domains of specified duration. Each time domain can have different minimum and maximum time step sizes and different edit intervals as entered in Card 15.1. The card must be repeated for each time domain desired. A final time domain with a negative value for DTMIN must be entered to terminate the calculation.

In the example below, the whole calculation of 12 seconds is divided into two time domains. The first time domain has a duration of 2 seconds (TEND = 2), a minimum time step size of  $10^{-6}$  seconds (DTMIN =  $10^{-6}$ ) and a maximum time step size of  $10^{-3}$  seconds, (DTMAX =  $10^{-3}$ ). The results will be printed out every half second (EDINT = 0.5) and the dump file deck.cdm will be written every second. The second time domain has a duration of 10 seconds (TEND = 10), a minimum time step size of  $10^{-6}$  seconds, (DTMIN =  $10^{-6}$ ), and an increased maximum time step of  $10^{-2}$  seconds, (DTMAX =  $10^{-2}$ ). The results will be printed out only once at the end (EDINT = 10.0) and the dump file deck.cdm will be written every 5 seconds.

The parameter RTWFP is a ratio of time step sizes for heat conduction solution and fluid solution. To obtain steady-state conditions, the conduction solution can generally use time steps greater than the fluid solution. For transient calculations RTWFP should be 1.0.

***	*******	*******	******	******	*******	******	***
* G	roup 15 - Tl	IME DOMAIN DATA					*
***	*******	*******	******	*****	******	*****	***
* C	ARD GROUP 15	5					
* N	GR						
	15						
* C	ard 15.1						
*	DTMIN	DTMAX	TEND	EDINT	DMPINT	RTWFP	
	0.000001	0.001	2.0	0.5	1.0	1.0	
*	DTMIN	DTMAX	TEND	EDINT	DMPINT	RTWFP	
	0.000001	0.01	10.0	10.0	5.0	1.0	
* C	ard 15.2						
*	DTMIN	(if negative stop)					
	-0.01	0.0	0.0	0.0	0.0	0.0	
***	*******	********	******	******	*******	******	***

## 22.7.3 Instructions to Preparation of Input Files for Restart Calculations

When the parameter DUMPF read on CARD INPUT 2 is not set to zero, a dump file 'deck.cdm' is generated. In order to restart the calculation, a restart file 'deck.crs' must be read. Because the initial calculation will generate an empty 'deck.crs' file, the dump file 'deck.cdm' must be renamed to 'deck.crs' before restarting the calculation.

Two dump/restart options are possible: "simple" and "full" restart. During the so-called "simple" restart run the user is allowed to change the time domain data, but not the power distribution and the flow conditions. The input file for the "simple" restart must contain CARDS INPUT.1, INPUT.2, INPUT.3, COBRA.1, and CARD GROUP 15; and must not contain CARD GROUP 1 through CARD GROUP 14.

During a "full" restart run, the user can specify changes in the operating conditions, power distribution, boundary conditions, printout options, and the time domain data. The input file for the "full" restart must contain CARDS INPUT.1, INPUT.2, INPUT.3, COBRA.1, and CARD GROUP 15; may contain CARD GROUP 1, CARD GROUP 11, CARD GROUP 12, CARD GROUP 13, and CARD GROUP 14; and must not contain CARD GROUP 2, CARD GROUP 3, CARD GROUP 4, CARD GROUP 5, CARD GROUP 6, CARD GROUP 7, CARD GROUP 8, CARD GROUP 9, or CARD GROUP 10.

Examples for "simple" and "full" restart input files follow.

"Simple" Restart:

```
*************************************
* 9 Channel PWR Core Model; Embedded Mesh Structure
* Loss of Coolant Transient
* Ref.: Avramova, M., et al., "Comparative Analysis of PWR Core Wide and Hot
      Channel Calculations", ANS Winter Meeting, 2002
*************************************
* CARD INPUT 1
      ICOBRA
* CARD INPUT 2
    INITIAL
                DUMPF
                 1
* CARD INPUT 3
      EPS0
                OITMAX
                           IITMAX
      0.001
                   1
                              40
* CARD COBRA 1
*<---->
   ****9_chan_PWR_core***
**********************************
* Group 15 - TIME DOMAIN DATA
*NGRP
  15
    DTMIN DTMAX TEND EDINT DMPINT RTWFP
  0.000001 0.01 4.0
                   1.0
                         1.0
                               1.0
    DTMIN DTMAX TEND EDINT DMPINT RTWFP
  0.000001 0.01 14.8
                         14.8
                     0.2
   DTMIN (if negative stop)
           0.0 0.0
                                 0.0
                     0.0
                           0.0
*************************************
*************************************
"Full" Restart:
*************************************
* 9 Channel PWR Core Model; Embedded Mesh Structure
* Loss of Coolant Transient
* Ref.: Avramova, M., et al., "Comparative Analysis of PWR Core Wide and Hot
      Channel Calculations", ANS Winter Meeting, 2002
*************************************
* CARD INPUT 1
      ICOBRA
* CARD INPUT 2
                 DUMPF
     INITIAL
         2
                    1
* CARD INPUT 3
                CAMTIO
       EPS0
                           IITMAX
      0.001
                    1
                              40
* CARD COBRA 1
```

```
*<---->
  ****9_chan_PWR_core***
*******************************
*GROUP 1 - Calculation Variables and Initial Conditions
************************************
*NGRP
  1
* NGAS IRFC EDMD IMIX ISOL NDM6 NDM7 NDM8 NDM9 NM10 NM11 NM12 NM13 NM14
           1 1
                  0 0
                          0
                              0
                                  0
                                      0
                                         0
    GTOT
           AFLUX
                  DHFRAC
     0.
            0.0
    PREF
            HIN
                   HGIN
                         VFRAC1
                                 VFRAC2
   160.00 1251.255
                 288.233
                          1.0
                                  .9999
*GTYPE
           VFRAC
           .0001
***********************************
* END GROUP 1
***********************************
********************************
* Group 11 - Axial Power Tables and Forcing Functions
************************************
*NGRP
  11
* NQA NAXP MNXN
             NQ NGPF NQR NDM7 NDM8 NDM9 NM10 NM11 NM12 NM13 NM14
  1 1 30
             30 0
                      2
                        0
                             0 0 0 0
                                           0 0
* Axial Power Distribution/Forcing Functions
     ΥQ
    10.0
  I NAXN
   1 30
      Y
           AXIAL
  1.0000
          1.0000
  0.0820
        1.0000
  0.1665
          1.0000
  0.2485
          1.0000
  0.3330
          1.0000
  0.4150
        1.0000
  0.4995
        1.0000
  0.5815
          1.0000
  0.6660
        1.0000
  0.7480
        1.0000
  0.8325
          1.0000
  0.9145
          1.0000
  0.9990
          1.0000
  1.0810
        1.0000
  1.1655
          1.0000
  1.2475
          1.0000
  1.3320
        1.0000
        1.0000
  1.4140
  1.4985
          1.0000
   1.5805
          1.0000
```

```
1.6650
            1.0000
   1.7470
            1.0000
   1.8315
            1.0000
   1.9135
            1.0000
   1.9980
            1.0000
   2.0800
           1.0000
   2.1645
            1.0000
   2.2465
            1.0000
   2.3310
            1.0000
   2.4130
            1.0000
* Total Power Forcing Functions
      YQ
               FQ
                        ΥQ
                                         YQ
                                 FQ
                                                  FQ
                                                           YQ
                                                                    FQ
      0.0
            0.0000
                       1.0
                             0.1000
                                         2.0
                                               1.0000
                                                         10.0
                                                                1.0000
     10.0
            1.0000
                      11.8
                             1.0000
                                        11.9
                                               0.9815
                                                         12.0
                                                                0.9629
     12.1
            0.9444
                      12.2
                             0.9259
                                        12.3
                                               0.9073
                                                         12.4
                                                                0.8888
                                        12.7
                                                         12.8
     12.5
            0.8702
                      12.6
                           0.8517
                                               0.8303
                                                                0.8096
     12.9
            0.7908
                           0.7734
                                        13.1
                                               0.7571
                                                         13.2
                                                                0.7418
                      13.0
     13.3
            0.7273
                      13.4
                             0.7135
                                        13.5
                                               0.7003
                                                         13.6
                                                                0.6877
                           0.6639
     13.7
            0.6756
                      13.8
                                        14.8
                                               0.5638
                                                         15.8
                                                                0.4835
     16.8
            0.4146
                      17.8
                             0.3546
                                        18.8
                                               0.3029
                                                         19.8
                                                                0.2589
     20.8
            0.2226
                      21.8
                             0.2162
* Radial Power Distribution/Forcing Functions
      YQR
      0.0
              FQR
      FQR
                       FQR
                                FRQ
                                         FQR
                                                 FRQ
                                                          FQR
                                                                   FRQ
  1.55769
           1.55769
                    1.55769
                            1.55769
                                     1.52627
                                              1.50229
                                                       1.51086
                                                                1.51087
  0.95895
      YQR
     10.0
      FQR
              FQR
                       FQR
                                FRQ
                                         FQR
                                                 FRQ
                                                          FQR
                                                                   FRQ
                            1.55769
                                              1.50229
  1.55769
           1.55769
                    1.55769
                                     1.52627
                                                       1.51086
                                                                1.51087
  0.95895
* END GROUP 11
************************
*GROUP 12 - Turbulent mixing data
*NGRP
  12
* Card 12.1
* AAAK
       BETA
  1.0 0.051
*********************
* END GROUP 12
* Group 13 - Boundary Conditions Data
***********************
```

```
*NGRP
   13
*NBND NKBD NFUN NGBD NDM5 NDM6 NDM7 NDM8 NDM9 NM10 NM11 NM12 NM13 NM14
         0 1 0 0 0 0 0 0 0 0 0 0
  18
*NPTS
  32
*ABSC
        ORDINT ABSC
                          ORDINT ABSC
                                          ORDINT
 0.0 0.0000 0.1 1.0000 10.0 1.0000 10.1 0.9976 10.2 0.9929
10.3 0.9867 10.4 0.9795 10.5 0.9716 10.6 0.9632 10.7 0.9547
10.8 0.9459 10.9 0.9371 11.0 0.9283 11.1 0.9196 11.2 0.9109

    11.3
    0.9023
    11.4
    0.8938
    11.5
    0.8855
    11.6
    0.8773
    11.7
    0.8693

    11.8
    0.8615
    11.9
    0.8537
    12.0
    0.8462
    13.0
    0.7817
    14.0
    0.7363

15.0 0.6960 16.0 0.6593 17.0 0.6259 18.0 0.5953 19.0 0.5672
20.0 0.5413 21.8 0.5413
* Inlet b.c. -----------------
*IBD1 IBD2 ISPC N1FN N2FN N3FN BCVALUE1 BCVALUE2 BCVALUE3 INITGAS
        1 2 1 0 0 .24961 1251.255 0.0 1

    2
    1
    2
    1
    0
    0
    .24961
    1251.255
    0.0

    3
    1
    2
    1
    0
    0
    .28699
    1251.255
    0.0

    4
    1
    2
    1
    0
    0
    .24961
    1251.255
    0.0

    5
    1
    2
    1
    0
    0
    3.1446
    1251.255
    0.0

    6
    1
    2
    1
    0
    0
    9.5544
    1251.255
    0.0

                                                           0.0
    7 1 2 1 0 0 32.872 1251.255
    8 1 2 1 0 0 75.777 1251.255
                                                            0.0
                                                                         1
    9
       1
            2
                  1 0 0 1522.97 1251.255
                                                           0.0
                                                                          1
* Outlet b.c. ----------------
*IBD1 IBD2 ISPC N1FN N2FN N3FN BCVALUE1 BCVALUE2 BCVALUE3
        42 1 0 0 0 0.0 1251.255 156.3042
    1
    2
        42 1 0 0 0
                                      0.0 1251.255 156.3042
    3
       42 1 0 0 0
                                      0.0 1251.255 156.3042

      3
      42
      1
      0
      0
      0
      0
      0
      1251.255
      156.3042

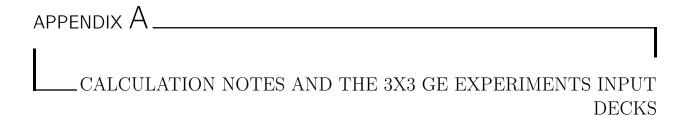
      5
      42
      1
      0
      0
      0
      0
      1251.255
      156.3042

      6
      42
      1
      0
      0
      0
      0
      1251.255
      156.3042

      7
      42
      1
      0
      0
      0
      0
      1251.255
      156.3042

        42 1 0 0 0
                                      0.0 1251.255 156.3042
                             0
                                       0.0 1251.255 156.3042
       42
            1
                          0
***********************************
* END GROUP 13
**********************************
***********************************
* Group 14 - Output Options
*************************************
*NGRP
   14
  N1 NOU1 NOU2 NOU3 NOU4 IPRP IOPT IRWR NDM9 NM10 NM11 NM12 NM13 NM14
       0 0 0 0 0 2 1 0 0 0 0 0
    5
*PRCH
*PRTG
```

*							
*P	*PRTR						
*							
**	*******	*****	*****	*****	*****	*****	**********
*	* END GROUP 14						
**	*******	*****	*****	*****	*****	*****	**********
*							
*							
**	********	*****	*****	*****	*****	*****	**********
*	Group 15 -	- TIME D	OMAIN	DATA			*
**	********	*****	*****	*****	*****	*****	***********
*N	IGRP						
	15						
*	DTMIN	DTMAX	TEND	EDINT	DMPINT	RTWFP	
*	0.000001	0.01	3.0	3.0	3.0	1.0	
*	DTMIN	DTMAX	TEND	EDINT	DMPINT	RTWFP	
	0.000001	0.01	14.8	0.2	14.8	1.0	
*	DTMIN	(if neg	ative	stop)			
	-1.0	0.0	0.0	0.0	0.0	0.0	
*							
************************							
* END GROUP 15							
			dealers and a dealers			Acatember 1. Acatember 1.	



## A.1 GE 3x3 Experimental Parameters

Table 10: Geometrical characteristics of the GE 3x3 test section

Parameter	Value
Number of Rods	9
Rod Diameter	14.7066 mm
Rod-Rod Clearance	4.2672 mm
Row-Wall Clearance	3.429 mm
Channel Corner Radius	10.16 mm
Heated Length	1828.8 mm
Hydraulic Diameter	12.0396 mm

Table 11: Experimental parameters of the GE 3x3 tests

Name of the test	GE 3x3	
Reference in	[9]	
literature	[3]	
	Mixing tests, yeilding:	
	mass flow rate	
Measured	equilibrium quality at the bundle outlet	
quantities	for the total bundle,	
quantities	for 3 subchannels each representing a characteristic class	
	of subchannels (corner, side, central) in non-simultaneous	
	sampling—only one subchannel sampled at a certain measurement	
Measuring	Total bundle mass flow rate: $\pm 8\%$	
uncertainties	Subchannel flow rate and subchannel enthalpy: $\pm 3\%$	

Table 12: GE 3x3 Test Points

Number of test point			Total rod power [kW]	Exit pressure [bar]
1B	1.231	93.7	0	70
1C	2.538	93.7	0	70
1D	3.871	93.7	0	70
1E	5.050	93.7	0	70
2B2	1.359	918.7	532	70
2B3	1.372	1014.6	532	70
2B4	1.372	1144.6	532	70
2C1	2.717	1134.4	532	70
2C2	2.738	1185.8	532	70
2D1	1.384	664.5	1064	70
2D3	1.384	978.1	1064	70
2E1	2.769	936.2	1064	70
2E2	2.769	1042.5	1064	70
2E3	2.717	1199.7	1064	70
2G1	2.743	742.0	1596	70
2G2	2.769	825.9	1596	70
2G3	2.743	926.2	1596	70
3B2	1.372	1237.4	532	70
3D1	1.397	2474.9	1064	70
3E1	2.769	2474.9	1064	70
3E2	2.717	2312.0	994	70

## A.2 CTF Model of the GE 3x3 Test Section

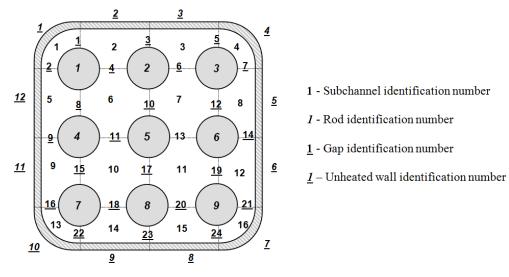


Figure A.1: Cross section of the CTF model of the GE 3x3 rod bundle

Table 13: CTF modeling parameters for the GE 3x3 Tests

Units in input file	US units				
Entrainment / deposition model	Original CTF model (Paleev and Filippovich correla-				
	tion for entrainment, Cousins model for deposition)				
Direct heating of the coolant (by neutron mod-none					
eration)					
Initial pressure of the fluid	equal to the outlet pressure				
Initial enthalpy of the fluid	equal to the inlet enthalpy				
Heat loss to ambient	none				
Initial rod temperature	$149~{\rm \mathring{A} \check{r} C}$ (137 K below the saturation temperature at				
	70 bars)				
Radiation	none				
Forcing functions for rod power	Test points 1B-1E: noneTest points 2B2-2G3:				
	noneTest points 3B2-3E2: non-uniform radial peaking				
	pattern (but not time-dependent)				
Mixing model	Standard mixing and void drift model according to				
	user specified mixing coefficient BETA. BETA=0.02, KM=1.4				
Boundary condition type	Inlet: mass flow rate and enthalpy Outlet: pressure				
	and enthalpy				
Simulation time	10 s to reach stationary conditions				
No. of test points in the calculation	21				

## A.3 CTF Input Deck for the GE 3x3 Test Point 2B2

```
* GE 3x3 void experiments: Janssen, E., "Two-phase flow and heat transfer in \quad *
* multi-rod geometries - Final Report", GEAP-10347, March 1971
* CTF input deck for test case 2B2
* MAIN PROBLEM CONTROL DATA
*************************
* CARD INPUT 1
     ICOBRA
* CARD INPUT 2
    INITIAL
                 DUMPF
                 1
* CARD INPUT 3
      EPSO
                OITMAX IITMAX
      0.001
* CARD COBRA 1
*---->
 *** GE 3x3 mixing test 2B2 ***
***********************************
* GROUP 1 - Calculation Variables and Initial Conditions
* CARD GROUP 1
* NGR
* Card 1.1
* NGAS IRFC EDMD IMIX ISOL NDM6 NDM7 NDM8 NDM9 NM10 NM11 NM12 NM13 NM14
   1 1 1 2 0 0 0
                             0
                                 0 0 0 0 0
* Card 1.2
```

```
GTOT
         AFLUX DHFRAC
   1.359
          32.322 0.0
* Card 1.3
          HIN HGIN VFRAC1 VFRAC2
  PREF
   70.00 918.77 288.4 1.0 0.9999
* Card 1.4
* GTP(1) VFRAC(3) GTP(2) VFRAC(4) GTP(3)
                                      VFRAC(5) GTP(4)
                                                    VFRAC(6)
  air
        .0001
* GROUP 2.0 - Channel Description
***********************************
* CARD GROUP 2
* NGR
  2
* Card 2.1
* NCHA NDM2 NDM3 NDM4 NDM5 NDM6 NDM7 NDM8 NDM9 NM10 NM11 NM12 NM13 NM14
 16 0 0 0 0 0 0 0 0 0 0 0 0
* Card 2.2
 I
                PW ABOT ATOP NMGP
          AN
   1 0.00005050 0.02834629 0.0 0.0 0.0
   2 0.00011766 0.04148719 0.0 0.0 0.0
   3 0.00011766 0.04148719 0.0 0.0 0.0
   4 0.00005050 0.02834629 0.0 0.0 0.0
   5 0.00011766 0.04148719 0.0 0.0 0.0
   6 0.00018675 0.04548398 0.0 0.0 0.0
   7 0.00018675 0.04548398 0.0 0.0 0.0
   8 0.00011766 0.04148719 0.0 0.0 0.0
   9 0.00011766 0.04148719 0.0 0.0 0.0
  10 0.00018675 0.04548398 0.0 0.0 0.0
  11 0.00018675 0.04548398 0.0 0.0 0.0
  12 0.00011766 0.04148719 0.0 0.0 0.0
  13 0.00005050 0.02834629 0.0 0.0 0.0
  14 0.00011766 0.04148719 0.0 0.0 0.0
  15 0.00011766 0.04148719 0.0 0.0 0.0
  16 0.00005050 0.02834629 0.0 0.0 0.0
****************************
* GROUP 3.0 - Transverse Channel Connection (Gap) Data
* CARD GROUP 3
* NGR
* Card 3.1
* NK NDM2 NDM3 NDM4 NDM5 NDM6 NDM7 NDM8 NDM9 NM10 NM11 NM12 NM13 NM14
 24 0 0 0 0 0 0
                           0
                               0 0 0 0 0
* Card 3.2
* K IK JK GAPN LNGTH WKR FWL IGPB IGPA FACT IGAP JGAP IGAP JGAP IGAP JGAP
 1 1 2 0.00343 0.01471 0.50 0.5 0 0 1.0 -1 3 0 0 0
* Card 3.3
* GMLT ETNR
1.000 0.000
* Cards 3.2 and 3.3 continued
 2 1 5 0.00343 0.01471 0.50 0.5 0 0 1.0 -1 9
                                                        0
                                                             0
                                                                 0
                                                                      0
1.000 0.000
 3 2
          3 0.00343 0.01875 0.50 0.5
                                  0
                                     0 1.0
                                                    5
                                                             0
                                                                 0
                                                                      0
                                                        0
1.000 0.000
          6 0.00427 0.01471 0.50 0.0
                                0 0 1.0 -1 11
1.000 0.000
                                     0 1.0
 5 3
           4 0.00343 0.01471 0.50 0.5
                                 0
                                                        0
                                                            0
                                                                 0
                                              3
                                                   -1
                                                                      0
1.000 0.000
           7 0.00427 0.01471 0.50 0.0
                                0
                                     0 1.0 -1 13
                                                        Ω
                                                            0
                                                               0
                                                                      0
 6 3
1.000 0.000
 7 4
           8 0.00343 0.01471 0.50 0.5
                                0
                                     0 1.0 -1 14
                                                        0
                                                             0
                                                                 0
                                                                      0
1.000 0.000
                                     0 1.0
 8 5
           6 0.00427 0.01471 0.50 0.0
                                0
                                             -1
                                                   10
                                                        0
                                                             0
                                                                 0
                                                                      0
1.000 0.000
 9 5
          9 0.00343 0.01875 0.50 0.5 0 0 1.0
                                             2
                                                   16
                                                        0
                                                             0
                                                                 0
                                                                      0
1.000 0.000
```

```
10
        6
             7 0.00427 0.01875 0.50 0.0
                                        0
                                             0 1.0
                                                                  0
                                                                       0
                                                                            0
                                                       8
                                                           12
1.000 0.000
            10 0.00427 0.01875 0.50 0.0
                                        0
                                             0 1.0
                                                                       0
                                                                            0
                                                       4
                                                           18
                                                                  0
  11 6
1.000 0.000
  12
             8 0.00427 0.01471 0.50 0.0
                                        0
                                              0 1.0
                                                      10
                                                            -1
                                                                  0
                                                                       0
                                                                            0
1.000 0.000
  13
            11 0.00427 0.01875 0.50 0.0
                                        0
                                              0 1.0
                                                       6
                                                            20
                                                                  0
                                                                       0
                                                                            0
1.000 0.000
            12 0.00343 0.01875 0.50 0.5
                                                                            0
  14
                                              0 1.0
1.000 0.000
            10 0.00427 0.01471 0.50 0.0
  15
       9
                                        0
                                             0 1.0
                                                      -1
                                                            17
                                                                  0
                                                                       0
                                                                            0
1.000 0.000
            13 0.00343 0.01471 0.50 0.5
                                        0
                                             0 1.0
                                                                       0
                                                                            0
 16 9
                                                       9
                                                            -1
                                                                  0
1.000 0.000
            11 0.00427 0.01875 0.50 0.0
  17
       10
                                        0
                                             0 1.0
                                                      15
                                                           19
                                                                  0
                                                                       0
                                                                            0
1.000 0.000
            14 0.00427 0.01471 0.50 0.0
  18 10
                                        0
                                              0 1.0
                                                      11
                                                            -1
                                                                  0
                                                                       0
                                                                            0
1.000 0.000
  19
       11
            12 0.00427 0.01471 0.50 0.0
                                        0
                                              0 1.0
                                                      17
                                                            -1
                                                                  0
                                                                       0
                                                                            0
1,000 0,000
            15 0.00427 0.01471 0.50 0.0
  20 11
                                        0
                                             0 1.0
                                                      13
                                                            -1
                                                                  0
                                                                       0
                                                                            0
1.000 0.000
  21 12
            16 0.00343 0.01471 0.50 0.5
                                        0
                                             0 1.0
                                                      14
                                                            -1
                                                                  0
                                                                       0
                                                                            0
1.000 0.000
  22 13
            14 0.00343 0.01471 0.50 0.5
                                        0
                                             0 1.0
                                                      -1
                                                            23
                                                                  0
                                                                       0
                                                                            0
1.000 0.000
            15 0.00343 0.01875 0.50 0.5
                                        0
                                             0 1.0
                                                      22
                                                                       0
                                                                            0
 23 14
                                                           24
                                                                  0
1.000 0.000
  24 15
            16 0.00343 0.01471 0.50 0.5
                                      Ω
                                             0 1.0
                                                      23
                                                           -1
                                                                 Ω
                                                                       0
                                                                            Ω
1.000 0.000
* Card 3.4
* NLGP
    0
* GROUP 4.0 - Vertical Channel Connection Data
* CARD GROUP 4
* NGR
   4
* Card 4.1
* NSEC NSIM IREB NDM4 NDM5 NDM6 NDM7 NDM8 NDM9 NM10 NM11 NM12 NM13 NM14
           0 0 0 0 0 0 0 0 0 0
   1 1
* Card 4.2
* ISEC NCHN NONO
                   DXS IVAR
             36 0.05080
   1
       16
* Card 4.3
   I KCHA KCHA KCHA KCHA
                                ксна кснв
                                           KCHB
                           KCHA
                                                KCHB
                                                     KCHB
                                                           KCHB
                                                               KCHB
    1
              0
                   0
                         0
                              0
                                   0
                                        1
                                              0
                                                   0
                                                        0
                                                             0
                                                                  0
    2
         2
              0
                   0
                         0
                              0
                                   0
                                        2
                                              0
                                                   0
                                                        0
                                                             0
                                                                  0
    3
                   0
                         0
                                                   0
                                                             0
                                                                  0
         3
              0
                              0
    4
         4
              0
                   0
                         0
                              0
                                   0
                                        4
                                              0
                                                   0
                                                        0
                                                             0
                                                                  0
    5
         5
              0
                   0
                         0
                              0
                                   0
                                        5
                                             0
                                                   0
                                                        0
                                                             0
                                                                  0
    6
         6
              0
                   0
                         0
                              0
                                   0
                                                             0
                                                   0
                                                                  0
    7
         7
              0
                   0
                         0
                              0
                                   0
                                        7
                                             0
                                                   Ω
                                                        0
                                                             0
                                                                  0
    8
         8
              0
                         0
                 0
                      0
                                       9
                                                0
    9
        9
             0
                              0
                                   0
                                             0
                                                      0 0
                                                                  0
                                                0
   10
        10
             0
                   0
                        0
                              0
                                   0
                                                       0
                                                             0
                                       10
                                             0
                                                                  0
   11
        11
              0
                   0
                         0
                              0
                                   0
                                                   0
                                                        0
                                                             0
                                                                  0
                                       11
                                                0
             0 0 0
                                      12
                                                      0
   12
        12
                             0
                                   Ω
                                             Ω
                                                            Ο
                                                                  0
             0 0 0
                                                             0
   13
        13
                                      13
                            0
   14
        14
             0 0 0
                                  0
                                      14
                                           0
                                                0
                                                     0 0
                                                                  0
                 0 0
   15
        15
             0
                             0
                                  0
                                       15
                                             0
                                                  0
                                                       0
                                                             0
                                                                  0
   16
        16
             0
                             0
                                   0
                                       16
                                             0
                                                   0
                                                        0
                                                             0
                                                                  0
* Card 4.5
```

\* IWDE  

```
* Card 4.6
* MSIM
 576
************************************
* GROUP 7 - Local Loss Coefficient and Grid Spacer Data
************************************
* CARD GROUP 7
* NGR
  7
* Card 7.1
* NCD NGT IFGQ IFSD IFES IFTP IGTM NFBS NDM9 NM10 NM11 NM12 NM13 NM14
       1 1 1 1 0
 18
                        0
                           0
                               0
* Card 7.2
    J ICDM
* CDL
     2 1
2 2
            4 13
                  16
                            0
                                             0
0.111
                      0
                                0
0.120
          3 5
                  8
                      9 12
                            14
                               15
                                   0
                                      0
                                          0
                                             0
0.177 2 6 7 10 11
0.111 8 1 4 13 16
                      0
                            0
                               0
                                      0
                                             0
                        0
                                         0
0.120 8 2 3 5 8
0.177 8 6 7 10 11
                                  0 0
0 0
                  8
                        12
                            14
                               15
                                        0
                      0
                        Ω
                            Ω
                               0
                                            0
0.111 14 1 4 13 16
                              0
                     0
                        0
                           0
                                 0 0 0
                                             0
0
0.111 20 1 4 13 16
0.120 20 2 3 5 8 9 12 14 15 0 0
                                             0
0.177 20 6 7 10 11 0 0 0 0 0 0
0.111 26 1 4 13 16 0 0
                           0 0 0 0 0
                                            0
0.120 26 2 3 5 8 9 12 14 15 0 0 0
0.177 26 6 7 10 11 0 0 0 0 0 0
         1 4 13
0.111 32
                  16
                      0
                            0
                                Ω
                                      Ω
                                          0
                                             0
0.120 32 2 3 5
                 8
                      9 12 14 15 0
                                      0
0.177 32 6 7 10
                 11
* GROUP 8.0 - Rod and Unheated Conductor Data
***********************************
* CARD GROUP 8
* NGR
* Card 8.1
* NRRD NSRD NC NRTB NRAD NLTY NSTA NXF NCAN RADF W3 NM12 NM13 NM14
  9 12 1 1 0 0 0 1 0 0 -1 0 0
* Card 8.2
 N IFTY IAXP NRND DAXMIN RMULT HGAP ISECR HTAMB
       1 0 0.00000 1.000 0.00000 1 0.000 0.000
  1 1
* Card 8.3
* NSCH PIE    1 0.250
         2 0.250 6 0.250 5 0.250 0 0.000 0 0.000 0 0.000 0 0.000
* Cards 8.2 and 8.3 continued
   2 \quad 1 \quad 1 \quad 0 \ 0.00000 \quad 1.000 \ 0.00000 \quad 1 \quad 0.000 \quad 0.000
   2 0.250
         3 0.250 7 0.250 6 0.250 0 0.000 0 0.000 0 0.000
                                                        0 0.000
        1 0 0.00000 1.000 0.00000 1 0.000 0.000
         4 0.250 8 0.250 7 0.250 0 0.000 0 0.000 0 0.000
   3 0.250
                                                        0.000
     1
        1 0 0.00000 1.000 0.00000 1 0.000 0.000
   5 0.250
         6 0.250 10 0.250 9 0.250 0 0.000 0 0.000
                                                  0.000
                                                          0 0.000
   5 1 1 0 0.00000 1.000 0.00000 1 0.000 0.000
   6 0.250
         0 0.000
```

```
6 1 1 0 0.00000 1.000 0.00000 1 0.000 0.000
     7 0.250
               8 0.250 12 0.250 11 0.250 0 0.000
                                                                   0 0.000 0 0.000
                                                                                              0 0.000
     7 1 1 0 0.00000 1.000 0.00000 1 0.000 0.000
     9 0.250 10 0.250 14 0.250 13 0.250 0 0.000 0 0.000 0 0.000
                                                                                             0 0.000
              1 0 0.00000 1.000 0.00000 1 0.000 0.000
               10 0.250
                                                                                                0 0.000
     9 1 1 0 0.00000 1.000 0.00000 1 0.000 0.000
    11 0.250
               0 0.000
                                                                                              0.000
* Card 8.5
   N ISTY HPERIM PERIMI RMULT NOSLCHC NSLCHC HTAMBS TAMBS
    1 2 0.01698 0.00000 1.00 1 0 0.000 0.000

      1
      2 0.01698 0.00000 1.00
      1
      0 0.000
      0.000

      2
      3 0.01875 0.00000 1.00
      2
      0 0.000
      0.000

      3
      3 0.01875 0.00000 1.00
      3
      0 0.000
      0.000

      4
      2 0.01698 0.00000 1.00
      4
      0 0.000
      0.000

      5
      3 0.01875 0.00000 1.00
      8
      0 0.000
      0.000

      7
      3 0.01875 0.00000 1.00
      9
      0 0.000
      0.000

      8
      3 0.01875 0.00000 1.00
      12
      0 0.000
      0.000

      9
      2 0.01698 0.00000 1.00
      13
      0 0.000
      0.000

      10
      3 0.01875 0.00000 1.00
      14
      0 0.000
      0.000

      11
      3 0.01875 0.00000 1.00
      15
      0 0.000
      0.000

      12
      2 0.01698 0.00000 1.00
      15
      0 0.000
      0.000

* Card 8.6
   I NRT1 NST1 NRX1
       9 12
* IRTAB     1 2 3 4 5 6 7 8 9 0 0 0
* IRTAB     1 2 3 4 5 6 7 8 9 10 11
* Card 8.9
* AXIALT TRINIT
0.0000000 300.00000
1.8300000 300.00000
*************************
* GROUP 9.0 - Conductor Geometry Description
*****************************
* CARD GROUP 9
* NGR
* Card 9.1
* NFLT IRLF ICNF IMWR NDM5 NDM6 NDM7 NDM8 NDM9 NM10 NM11 NM12 NM13 NM14
     3 0 0 0 0 0 0 0
                                              0 0 0
* Card 9.6
 I FTYP DROD DIN NFUL ITOX ITIX NDM8 NDM9 NM10 NM11 NM12 NM13 NM14
   * Card 9.7
* NODR MATR
              TREG
                        QREG
    2 1 0.00100 1.00000
* Card 9.6
* I FTYP DROD DIN NFUL ITOX ITIX NDM8 NDM9 NM10 NM11 NM12 NM13 NM14
    2 wall 0.01698 0.00100 1 1 1 0 0 0 0 0
* Card 9.7
* NODR MATR TREG
                        QREG
    2 1 0.00100 0.00000
```

```
* Card 9.6
* I FTYP DROD DIN NFUL ITOX ITIX NDM8 NDM9 NM10 NM11 NM12 NM13 NM14
  3 wall 0.01875 0.00100 1 1 1 0 0 0 0 0
* Card 9.7
* NODR MATR TREG QREG
   2 1 0.00100 0.00000
*******************************
* GROUP 10 - Material Properties Tables
*************************************
* CARD GROUP 10
* NGR
 10
* Card 10.1
* NMAT NDM2 NDM3 NDM4 NDM5 NDM6 NDM7 NDM8 NDM9 NM10 NM11 NM12 NM13 NM14
  1 0 0 0 0 0 0 0 0 0 0 0
* Card 10.2
* N NTDP RCOLD
                                  IMATAN
                              Inconel 600
 1 6 8470.57
* Card 10.3
  TPROP CPF1 THCF
-73 0.377 13.40
     93 0.464 15.71
     204 0.485 17.44

    427
    0.527
    20.90

    649
    0.586
    24.79

    871
    0.623
    28.83

* GROUP 11.0 - Axial Power Tables and Forcing Functions
* CARD GROUP 11
* NGR
 11
* Card 11.1
* NQA NAXP MNXN NQ NGPF NQR NDM7 NDM8 NDM9 NM10 NM11 NM12 NM13 NM14
 1 1 2 0 0 1 0 0 0 0 0 0 0
* Axial Power Forcing Functions
* Card 11.2
    YQA
     0.0
* Card 11.3
* I NAXN
  1 2
* Card 11.4
* Y
           AXIAL
 0.000000 1.000000
 1.830000 1.000000
* Total Power Forcing Functions
* Card 11.5
              FQ
   YQ
* 0.0000 0.0000
* 1.0000 1.0000
* 100.0000 1.0000
* Radial Power Forcing Functions
* Card 11.7
     YQR
     0.0
* Card 11.8

        FQR
        FQR
        FQR
        FQR
        FQR
        FQR

        1.0000
        1.0000
        1.0000
        1.0000
        1.0000
        1.0000

  FQR
   1.0000
   1.0000
* GROUP 12 - Turbulent mixing data
```

```
*********************************
* CARD GROUP 12
* NGR
  12
* Card 12.2
* AAAK DFRD THEM
  1.4 0.01448
              5.0
* GROUP 13.0 - Boundary Condition Data
* CARD GROUP 13
* NGR.
  13
* Card 13.1
* NBND NKBD NFUN NGBD NDM5 NDM6 NDM7 NDM8 NDM9 NM10 NM11 NM12 NM13 NM14
 32 0 1 0 0 0 0 0 0 0 0 0 0
* Card 13.2
* NPT
* Card 13.3
* ABSC ORDINT ABSC ORDINT ABSC ORDINT
  0.0 0.000 0.1 0.000 0.2 1.000 1500.0 1.000
* Card 13.4
* Inlet b.c. ------
* IBD1 IBD2 ISPC N1FN N2FN N3FN BCVALUE1 BCVALUE2 BCVALUE3 INITGAS
       1 2 1 0 0
                           0.0
   1
                                    918.77
                                             0.0000
  1 2 1 0 0
                              0.0 918.77 0.0000
                                                          1
                                                          1
                                                         1
                                                          1
                                                         1
                                                         1
                                                          1
* Outlet b.c. ------
* IBD1 IBD2 ISPC N1FN N2FN N3FN BCVALUE1 BCVALUE2 BCVALUE3 INITGAS
                                   918.77
                                     918.77 --

    38
    1
    0
    0
    0
    0.0000

    38
    1
    0
    0
    0
    0.0000

   1
   2
      38 1 0 0 0 0.0000 918.77 70.00
   3
       38 1 0 0 0 0.0000 918.77 70.00
    4
   5 38 1 0 0 0 0.0000 918.77 70.00
6 38 1 0 0 0 0.0000 918.77 70.00
7 38 1 0 0 0 0.0000 918.77 70.00
8 38 1 0 0 0 0.0000 918.77 70.00
                                                          1
                                                          1
       38 1 0 0 0 0.0000 918.77
   9
                                               70.00
                                                         1
                                    918.77

    10
    38
    1
    0
    0
    0
    0.0000

    11
    38
    1
    0
    0
    0.0000

    12
    38
    1
    0
    0
    0.0000

                                               70.00
                                                         1
                                      918.77
                                                70.00
                                    918.77
                                               70.00
                                                         1
   13 38 1 0 0 0 0.0000 918.77
                                               70.00
  14 38 1 0 0 0 0.0000 918.77
                                               70.00
                                                         1
                                    918.77
          1 0 0 0 0.0000
1 0 0 0 0.0000
   15
      38
                                               70.00
                                                          1
   16
       38
                                      918.77
                                                70.00
* Group 14 - Output Options
*NGRP
  14
 N1 NOU1 NOU2 NOU3 NOU4 IPRP IOPT IRWR NDM9 NM10 NM11 NM12 NM13 NM14
         0 0 0 0 4 1 0 0 0 0 0
```

*PRC	Ή								
*	5	13	17	20					
*PRT	'G								
*	9	25	36						
*PRT	'R								
*	4	10	12						
*PRT	'S								
*	5	8							
*									
****	***	****	****	******	******	******	******	******	***
* Gr	oup	15 -	TIME	DOMAIN DATA					*
****	*************************								
* CA	* CARD GROUP 15								
* NG									
	15								
* Ca	rd								
*		DTM		DTMAX	TEND	EDINT	DMPINT	RTWFP	
		00000	01	.01	10.	10.	20.0	1.0	
* Ca	rd								
*				f negative stop)					
		0	01	0.0	0.0	0.0	0.0	0.0	
*									
*************************									
* EN	D G	ROUP	TIME	DOMAIN DATA					*
****	****************************								

BIBLIOGRAPHY

- [1] S.G. Beus. A Two-Phase Turbulent Mixing Model for Flow in Rod Bundles. Technical Report WAPD-TM-2438, Bettis Atomic Power Laboratory, 1971.
- [2] C.Y. Paik, L.E. Hochreiter, J.M. Kelly, and R.J. Kohrt. Analysis of FLECHT-SEASET 163-Rod Blocked Bundle Data using COBRA-TF. Technical Report NUREG/CR-3046, United States Nuclear Regulatory Commission, 1985.
- [3] D. W. Radcliffe R. T. Lahey Jr., B. S. Shirlakar. Two-Phase Flow and Heat Transfer In Multirod Geometries: Subchannel and Pressure Drop Measurements in a Nine-Rod Bundle for Diabatic and Adiabatic Conditions. Technical report, General Electric, 1970.
- [4] R.K. Salko and M.N. Avramova. CTF Theory Manual. The Pennsylvania State University.
- [5] M. J. Thurgood et al. COBRA/TRAC—A Thermal-Hydraulics Code for Transient Analysis of Nuclear Reactor Vessels and Primary Coolant Systems. Technical Report NUREG/CR-3046, Pacific Northwest Laboratory, 1983.
- [6] M. Yilmaz. Development of Burnup Dependent Fuel Rod Model in CTF. PhD thesis, 2014.